Advancing Turbine Technologies for Relevant Inlet Temperature Profiles in the Steady Thermal Aero Research Turbine (START) Lab





PennState Steady Thermal Aero Research Tur

Research Staff

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Michael Barringer Reid Berdanier Justin Brumberg Jeremiah Bunch Tom Houck Matt Meier Karen A. Thole Joel Wagner

Assoc. Res. Prof. **Associate Professor** Research Engineer Eng. Technician Eng. Proj. Manager Asst. Res. Prof. Affiliate Faculty **Research Scientist**

NEXT

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Graduate Students

Nicholas Gailey	Maria Rozman
Connor Wiese	Chad Schaeffer
Margaret Nunn	Ethan Bonn
Ryan Van Domelen	Win Horning
Abigail Altland	Matt Stuber
Christopher Mickel	

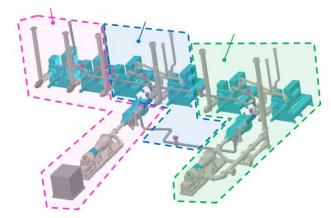
DOE-NETL PM: Matthew Adams



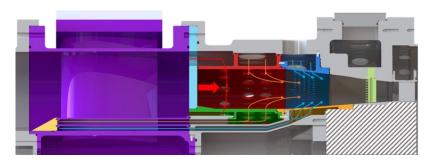




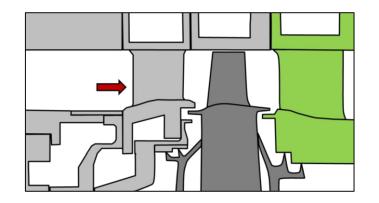
This presentation will review key project details outlining a portion of new capabilities comprising an expanded turbine research space at Penn State



New START⁺ facility design specifications

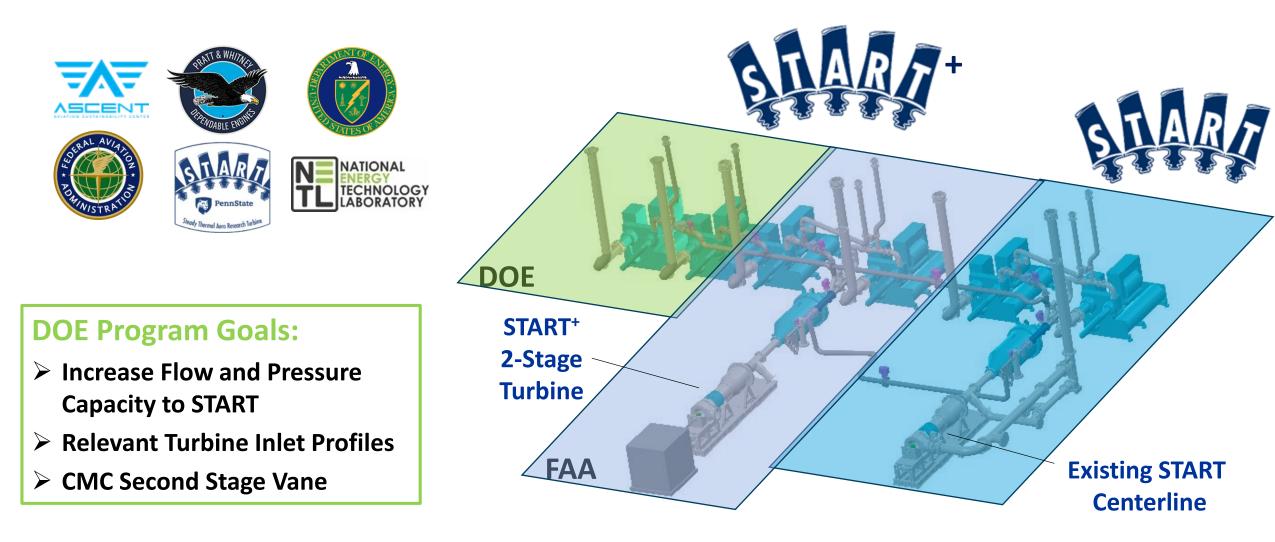


Turbine inlet profile generator capabilities



CMC hardware overview and integration plan

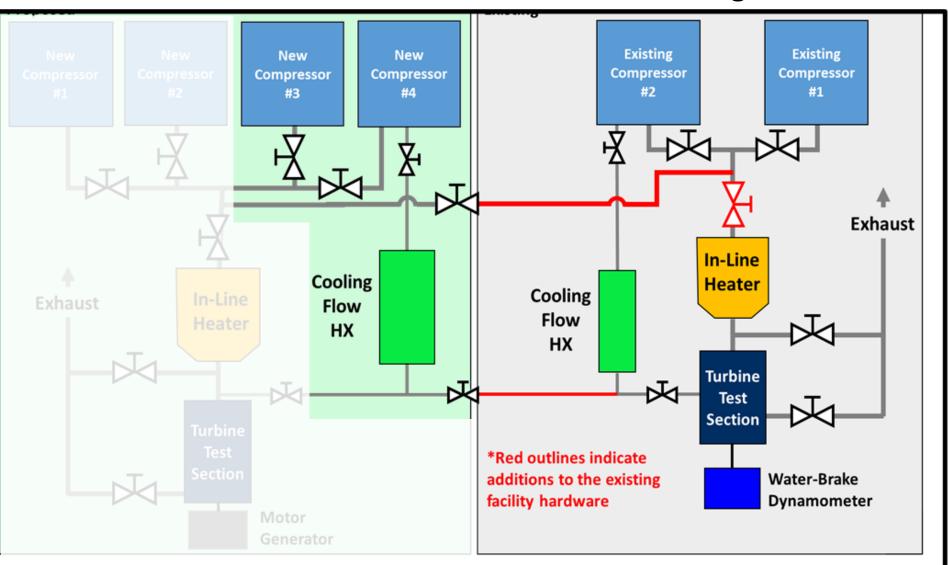
START will expand the lab to increase flow and pressure capacity for studying relevant turbine inlet profiles, CMC second-stage vanes, and a two-stage turbine



New compressors will enhance the air supply capability of the existing START facility to enable a broader range of turbine operating conditions

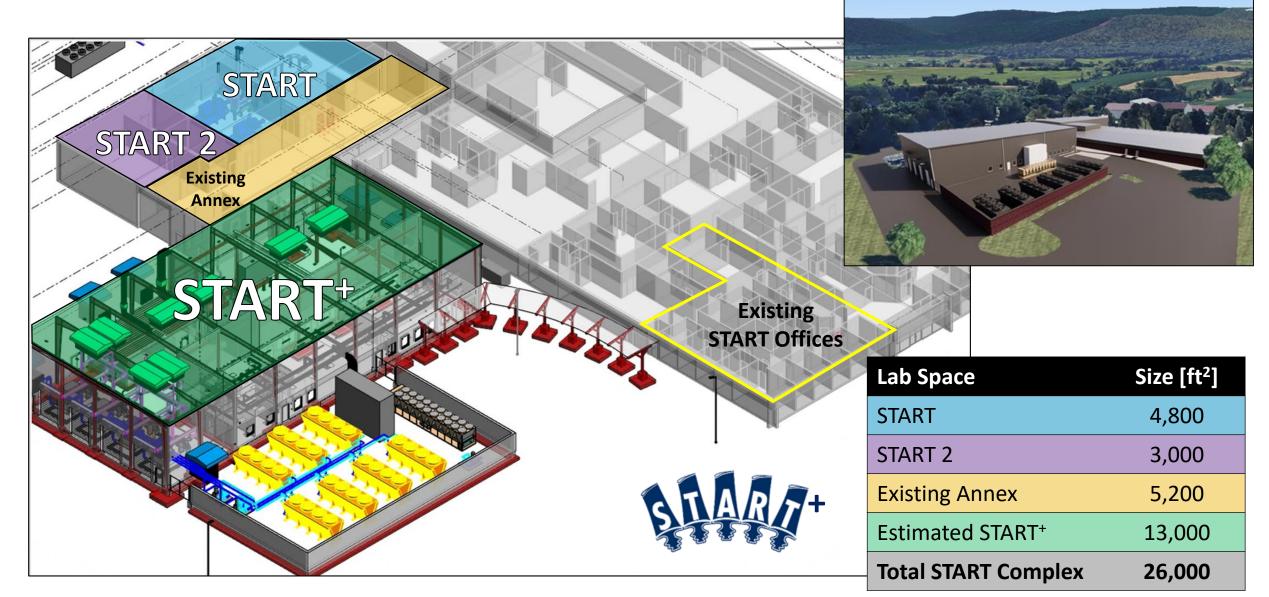
New Addition

Existing



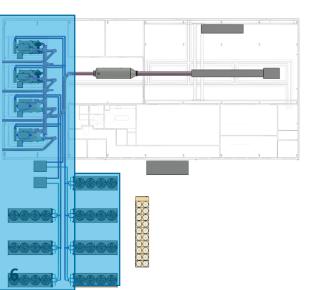


The design of the new lab addition is complete including the arrangement of the air compressors and equipment



Our START team worked to complete the compressor specification with FS Elliott

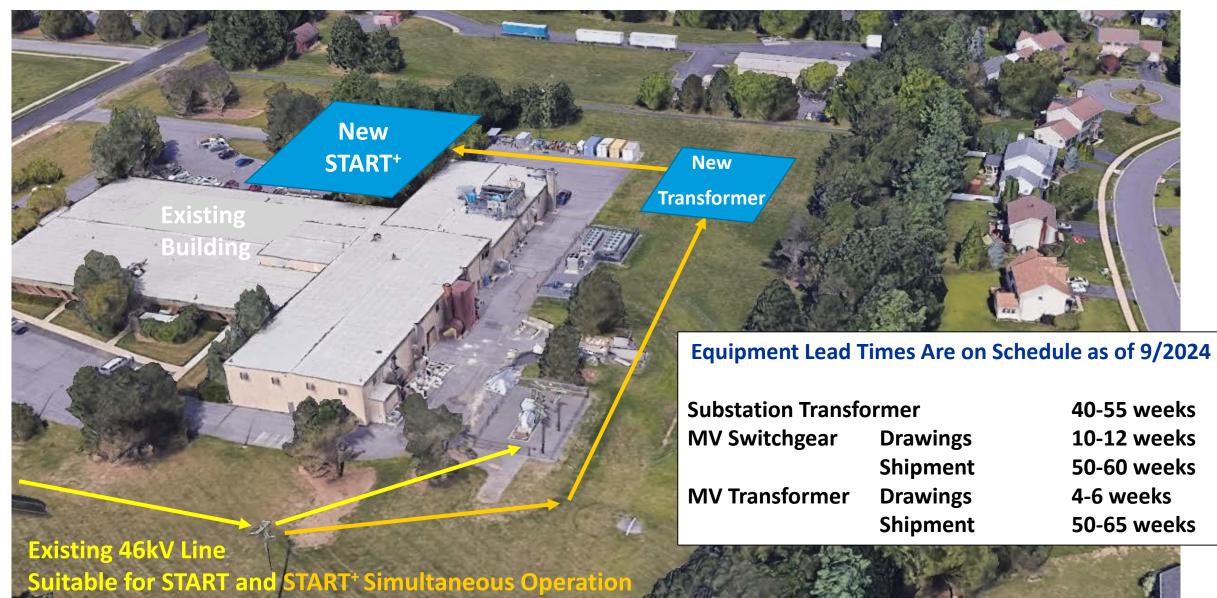
Compressor Specification	Existing START	START ⁺
Compressor Stages	2	3
Total Horsepower	3000	9000
Flow Rate per Unit	12 lbm/s	12 lbm/s
Number of Compressors	2	4
Total Number of Coolers	2	7





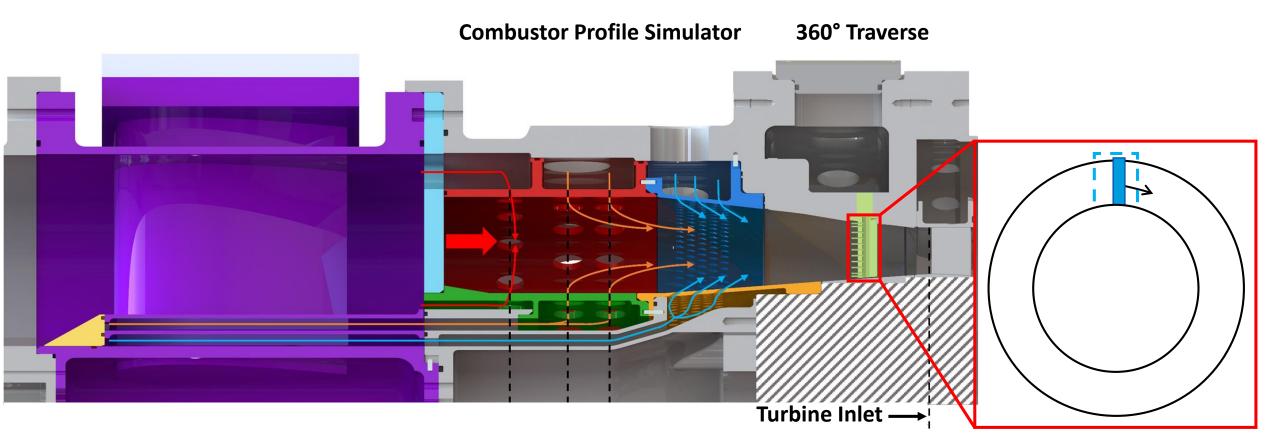


The new air compressors require additional high voltage power equipment, and lead times currently meet project schedule



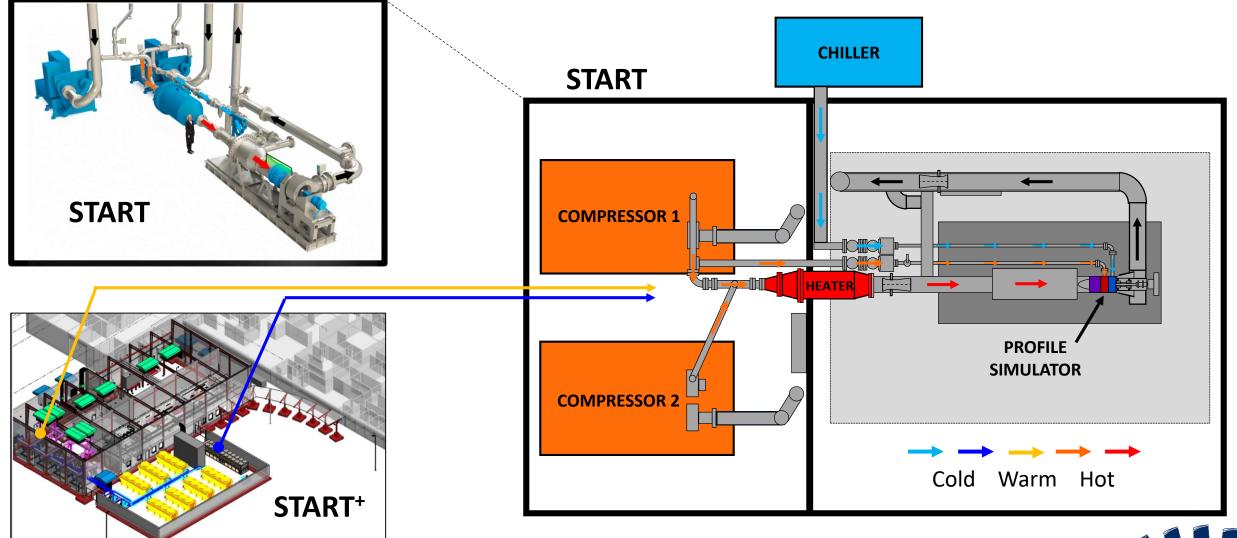
A detailed design has been developed to simulate different profile shapes at the inlet of the START test turbine





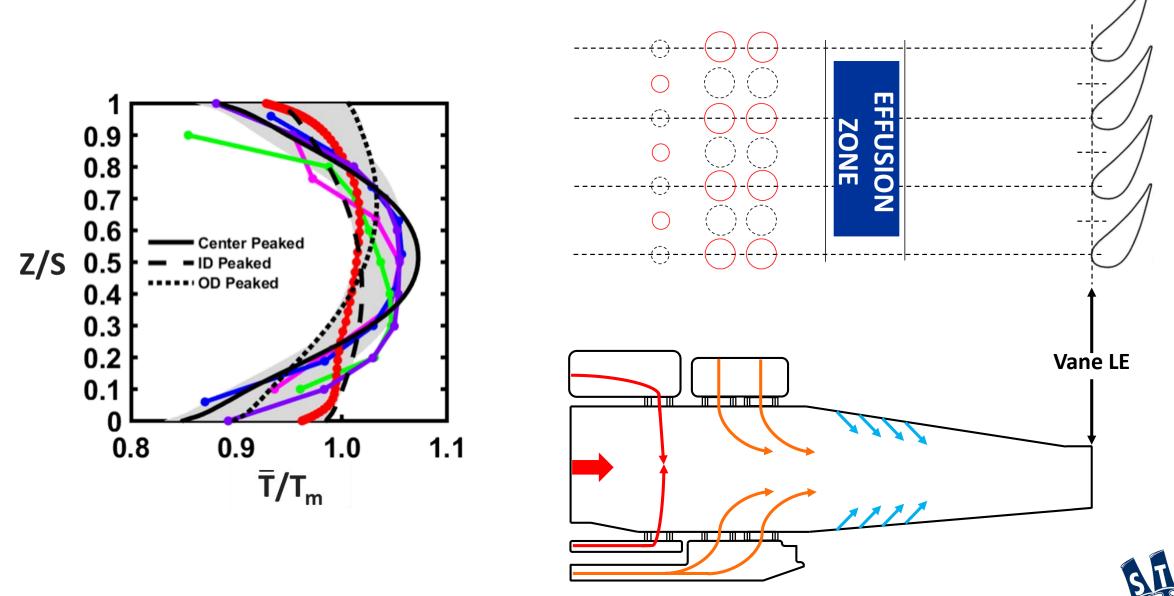


The simulator will use existing equipment and START⁺ capabilities to provide air flow at multiple temperatures for generating a range of inlet thermal profiles

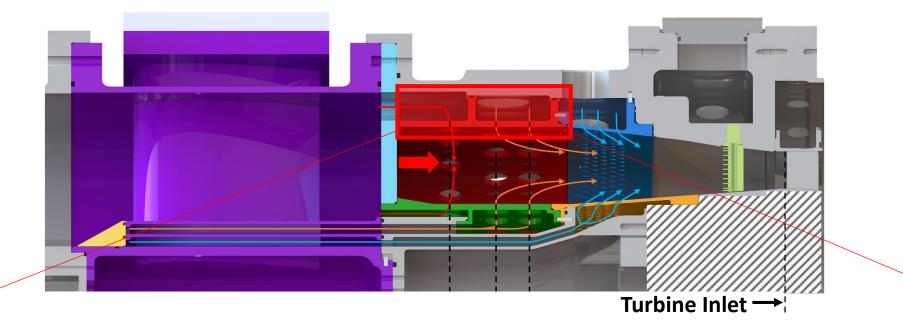


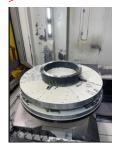


Numerous CFD studies were conducted to determine operating envelopes



The combustor simulator hardware components are continuing to progress well through manufacturing stages and should be ready for installation in 2025











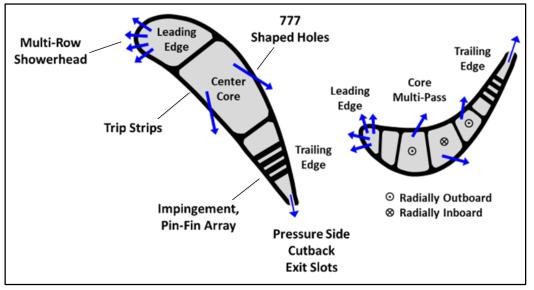




Inlet profile effects with the combustor simulator will be evaluated using the National Experimental Turbine (NExT) geometry as a primary research vehicle

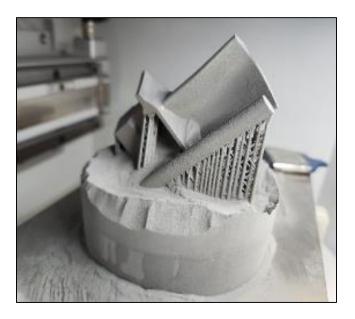


A fully-cooled one-stage turbine geometry designed collaboratively for research utility



[Thole et al., 2020]

A shared learning vehicle for manufacturing, computational capabilities, and turbine performance



The National Experimental Turbine (NExT) program continues to pay dividends to the turbine community and the OEM partners







A Caterpillar Company

Honeywell

AEROSPACE



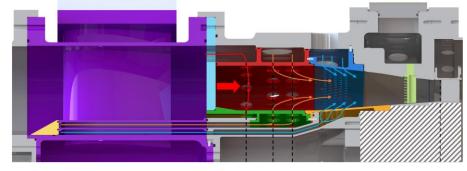
NExT research continues with OEM partners, including two funded proprietary projects to-date



A broad partnership of government agencies is well-engaged behind the NExT leadership of NETL



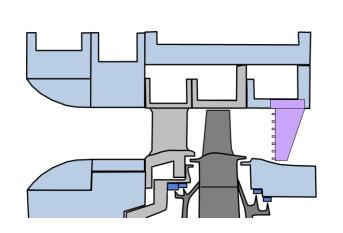
Turbine profiles effects will be measured using established instrumentation methodologies that underpin aerothermal research expertise in the START Lab

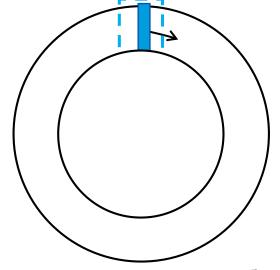


Blade heat transfer

Stage thermal efficiency

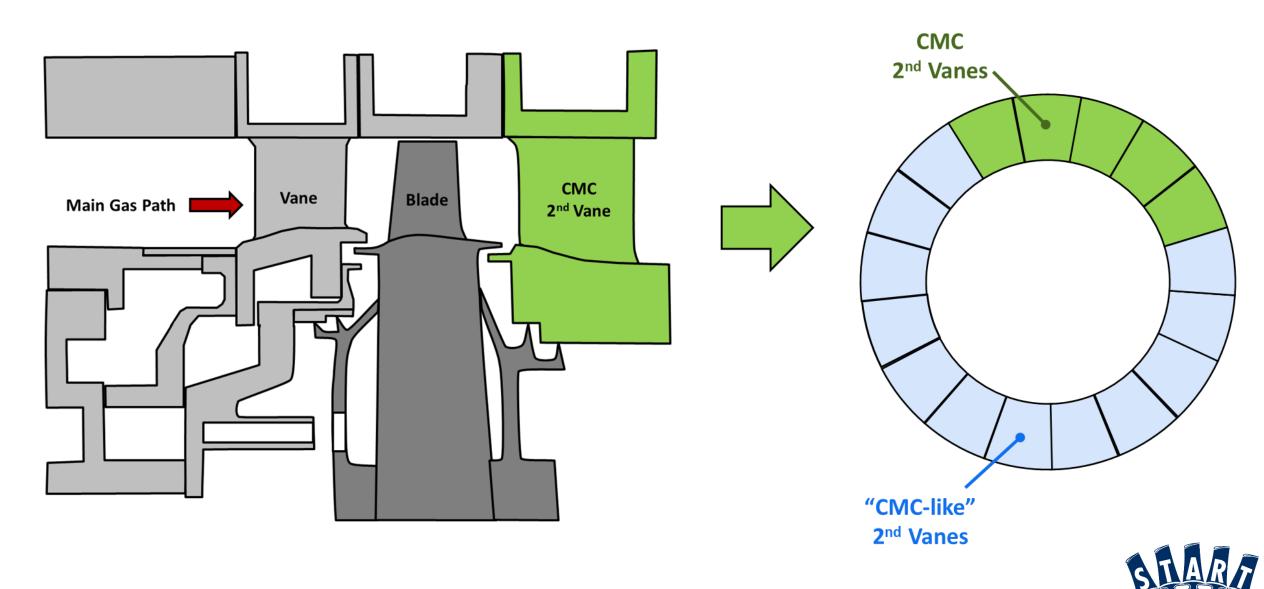




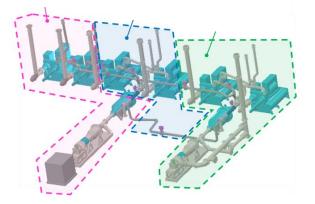




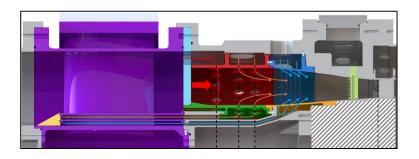
A CMC 2nd vane will be integrated into the existing START facility and evaluated for 1.5-stage turbine efficiency effects due to CMC-unique design characteristics



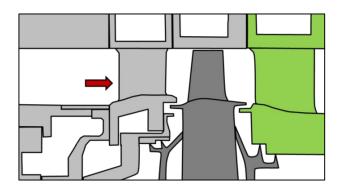
This project with DOE-NETL is enabling new opportunities that are important for the future advancement of turbine power generation needs



START⁺ will be a new facility that enhances existing START Lab research capabilities while also supporting a new 2-stage turbine



A new inlet profile generator will be complete in FY25 to experimentally demonstrate relevant turbulence BCs and temperature profiles for H₂



The installation of CMC-2V hardware will create new learning enabling integration of high-temp materials for future turbine designs