



# LOAD-Z

Low-NO<sub>x</sub>, Operable Ammonia Combustor Development  
for Zero-Carbon Power

## RTX Technology Research Center

University Turbine Systems Research (UTSR) & Advanced Turbines Program  
Review 2023 Meeting, State College, PA  
30 Oct. – 1 Nov. 2023

*Prime Contractor:* RTX Technology Research Center (RTRC)

*Subcontractor:* University of Connecticut (UConn)

# LOAD-Z Project Team

Organization	Role in project
RTX Technology Research Center (RTRC)	<ul style="list-style-type: none"> <li>- Project lead (prime).</li> <li>- Experiments in high-pressure staged combustion &amp; swirl-stabilized combustion of NH<sub>3</sub>.</li> <li>- Modeling, design, &amp; testing of low-NOx gas turbine combustor for NH<sub>3</sub>.</li> </ul>
University of Connecticut (UConn)	<ul style="list-style-type: none"> <li>- Fundamental NH<sub>3</sub> combustion experiments:               <ul style="list-style-type: none"> <li>• counterflow (strained) laminar flames</li> <li>• turbulent flame speed &amp; structure</li> </ul> </li> <li>- Development of chemical kinetic models/mechanisms.</li> </ul>



Lance Smith (PI)  
Gas turbine combustor design & test



Paul Papas (co-PI)  
Combustion experiments & modeling



Jordan Snyder  
Combustion diagnostics & measurements



Steve Zeppieri  
Combustion kinetics & modeling



University of  
Connecticut



Chih-Jen (Jackie) Sung (UConn lead)  
Flame structure, emissions, and chemical kinetic models



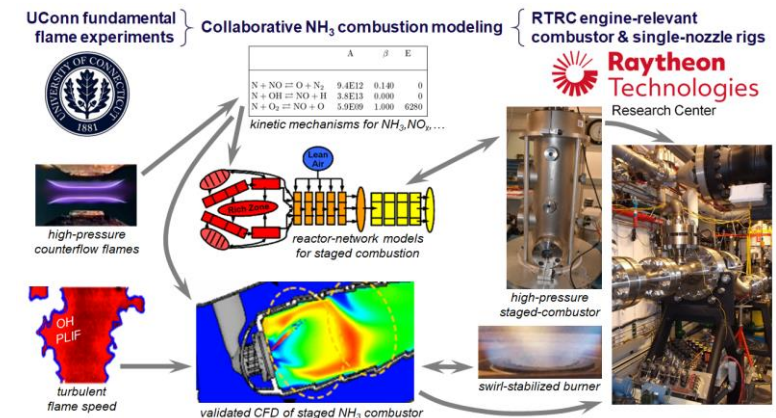
Baki Cetegen  
Turbulent combustion and diagnostics



James Stevens  
Combustion experiments & diagnostics

# Outline

- Project Objectives & Partnership
- Counterflow Flame Studies – Reaction Rates for  $\text{NH}_3$  Combustion
  - Experiments at 1 – 6 atm pressure (UConn)
  - Kinetic modeling in 1-D flames w/ radiation (RTRC)
- Burner Studies –  $\text{NO}_x$  Emissions
  - Flat-flame burner experiments (30 atm pressure staged reactor rig)
  - Chemical Reactor Network (CRN) modeling of GT combustor
- Next Steps – Year-2 plans
  - UConn experiments incl. new rig
  - RTRC high-pressure experiments & modeling





# LOAD-Z Project – Objectives & Partnership

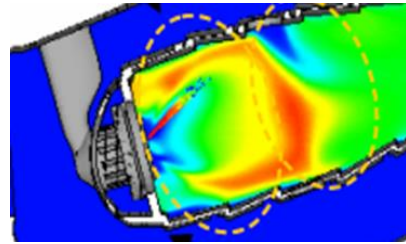
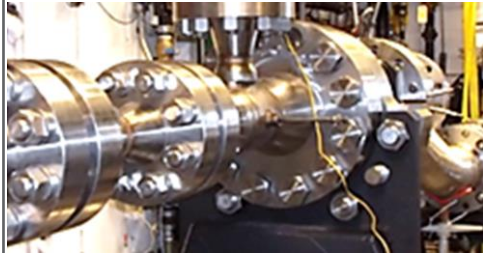
## Low-NOx, Operable Ammonia Combustor Development for Zero-Carbon Power

- **End Goal:** Demonstrate performance of gas-turbine-capable NH<sub>3</sub>-fuel combustor, at single-nozzle scale.
- **Go Get:** Obtain fundamental & engineering NH<sub>3</sub>-flame data at gas-turbine-relevant conditions →  $P_{inlet}, T_{inlet}$ .

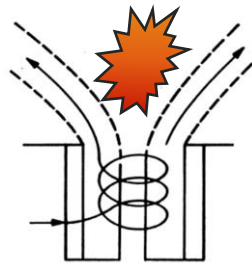
**RTRC**

**UConn**

Year-4  
Year-3  
Year-2  
Year-1  
Time



- Single-nozzle high-pressure combustor, fired w/NH<sub>3</sub> fuel
- Measure emiss. & performance: NOx, efficiency, stability

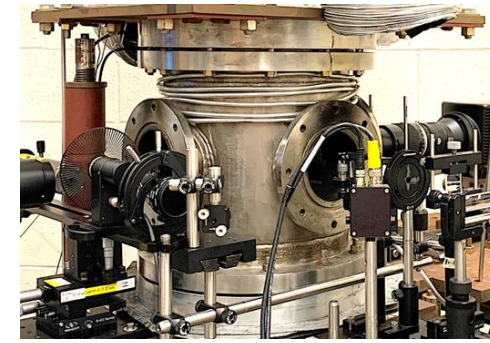


- 1-atm swirl-stab. burner
- Piloting studies w/ H<sub>2</sub>

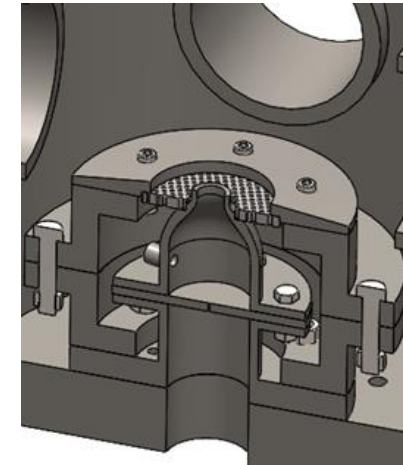
- Flat-flame high-P burner (>>1-atm) for NOx evaluation

### Modeling:

- CFD for design
- Kinetic improve. w/ exp. data
- CFD & validation
- Turb. models for NH<sub>3</sub> comb. & NOx (no post-process.)
- CRN modeling
- Counterflow
- Kinetic mechanisms



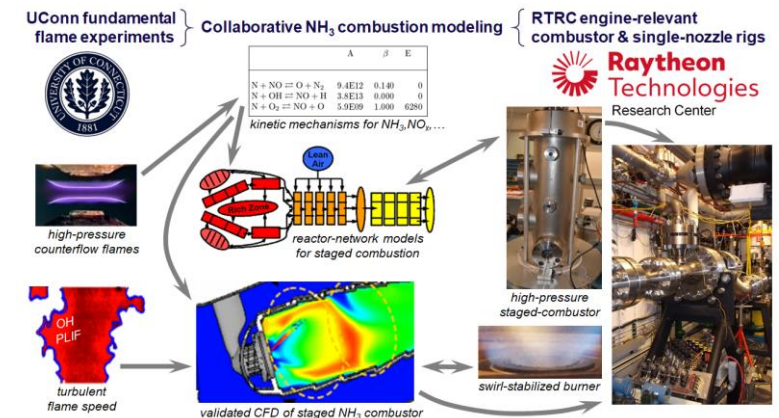
- Counterflow flame rig, compatible w/NH<sub>3</sub> fuel
- Measure strained flames w/ inlet P, T > ambient



- Turbulent S<sub>L</sub> rig, for NH<sub>3</sub> @ P, T > ambient (~20% turb. intensity)

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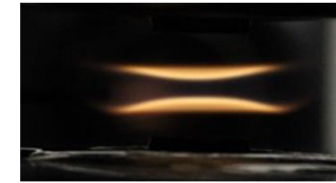


# Current Gaps for Understanding Ammonia Combustion

## Combustion Challenges

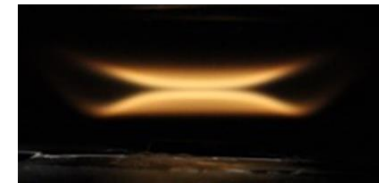
- Low overall reaction rate
  - Relatively low flame speed & **extinction strain rates**
- Flame stabilization & efficiency in **highly turbulent flow (GT comb.)**
- Quantifying NO<sub>x</sub> formation & emissions

## Counterflow NH<sub>3</sub>/Air Premixed Flames



Decreasing Strain Rate  
(Laminar Flame Speed Measurement)

Increasing Strain Rate  
(Extinction Limit Measurement)



## What is needed for improving ammonia models?

- Data at elevated pressure and temperatures relevant to gas turbines
- Extinction strain rate measurements for **both premixed** and **non-premixed** counterflow flames
  - ✓ Local velocity field measurements needed to determine **local** strain rate
  - ✓ Quantitative speciation (temperature) profiles (e.g., NO) at elevated pressures and temperatures
- Other data sets (shock tube, etc.) to better constrain reaction rate constant definitions

**What are highly sensitive reactions with large specific rate constant uncertainty?**



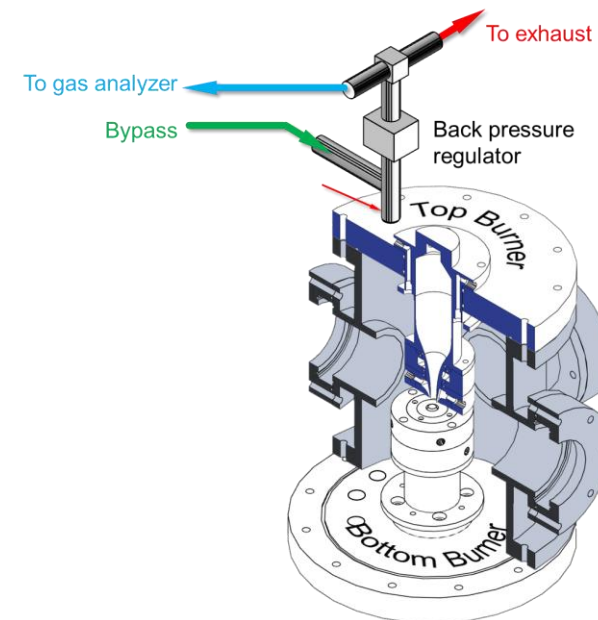
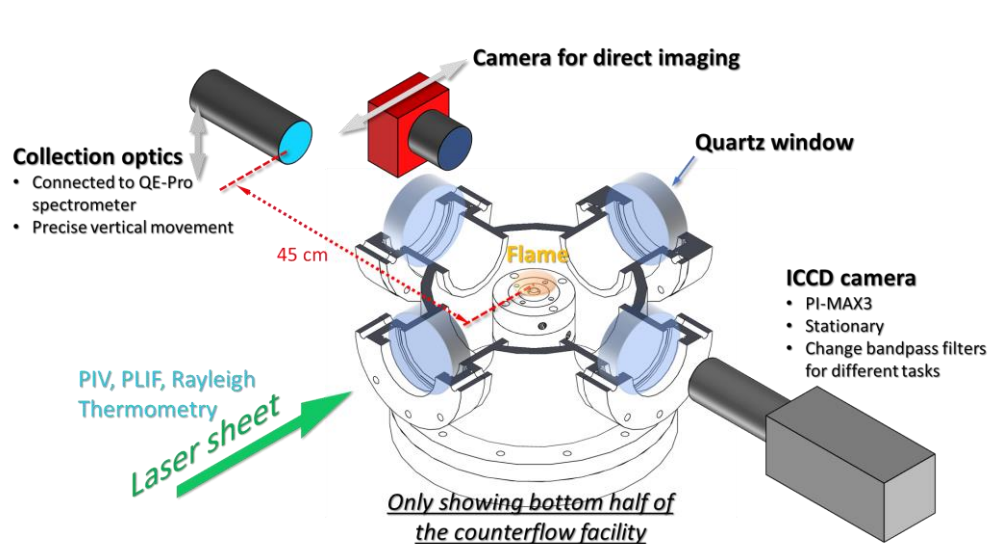


# UCONN Counterflow Strained Flame Rig with Diagnostics



## Capability of Counterflow Flame Rig

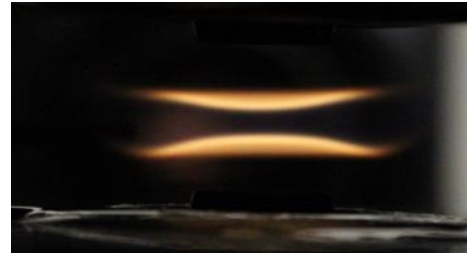
- *Ammonia compatible*
- *8+ atm*
- *200 °C (now) → 500 °C (later)*
- *Premixed, non-premixed, and partially-premixed combustion*



2



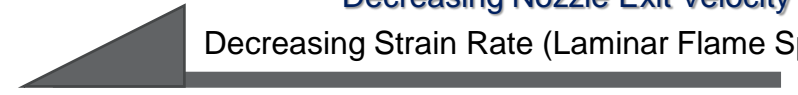
# Premixed NH<sub>3</sub> Counterflow Flames



Premixed NH<sub>3</sub> –Air Flames,  $\phi=1$

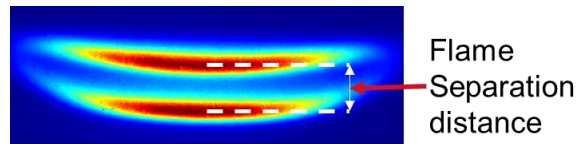
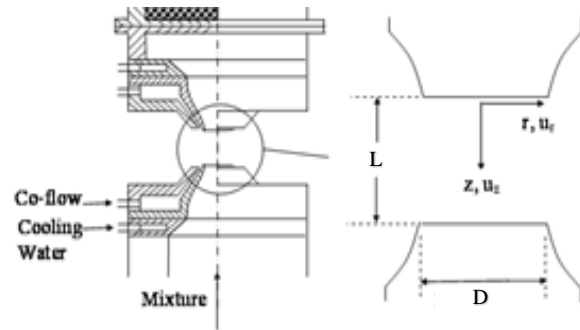
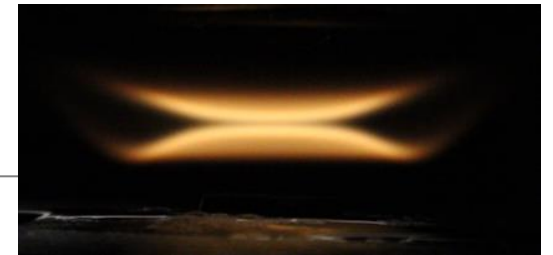
Decreasing Nozzle Exit Velocity

Decreasing Strain Rate (Laminar Flame Speed Measurement)

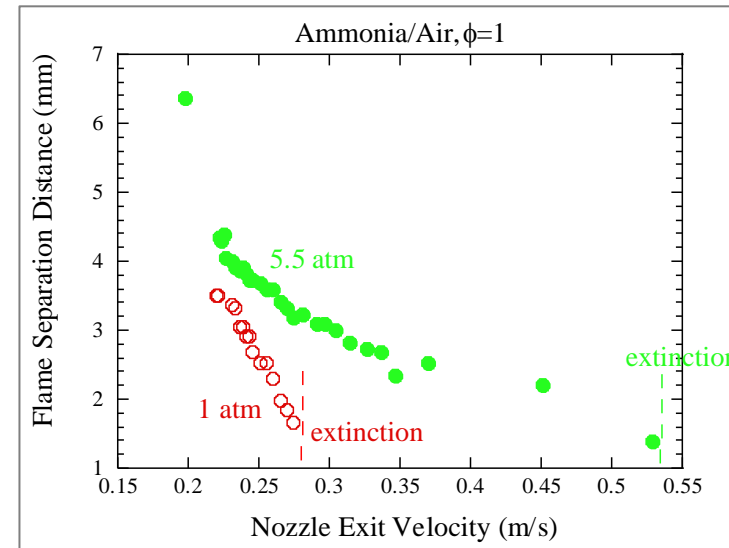


Increasing Strain Rate (Extinction Limit Measurement)

Increasing Nozzle Exit Velocity



Flame Separation distance



Global Strain Rate:

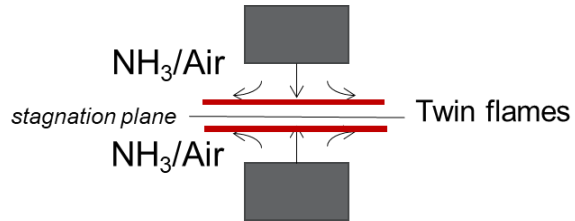
$$\kappa_G = \frac{4U_{ext}}{L}$$

Flame strain (stretch) rate is represented by the velocity gradient ahead of the flame and is proportional to nozzle exit velocity.





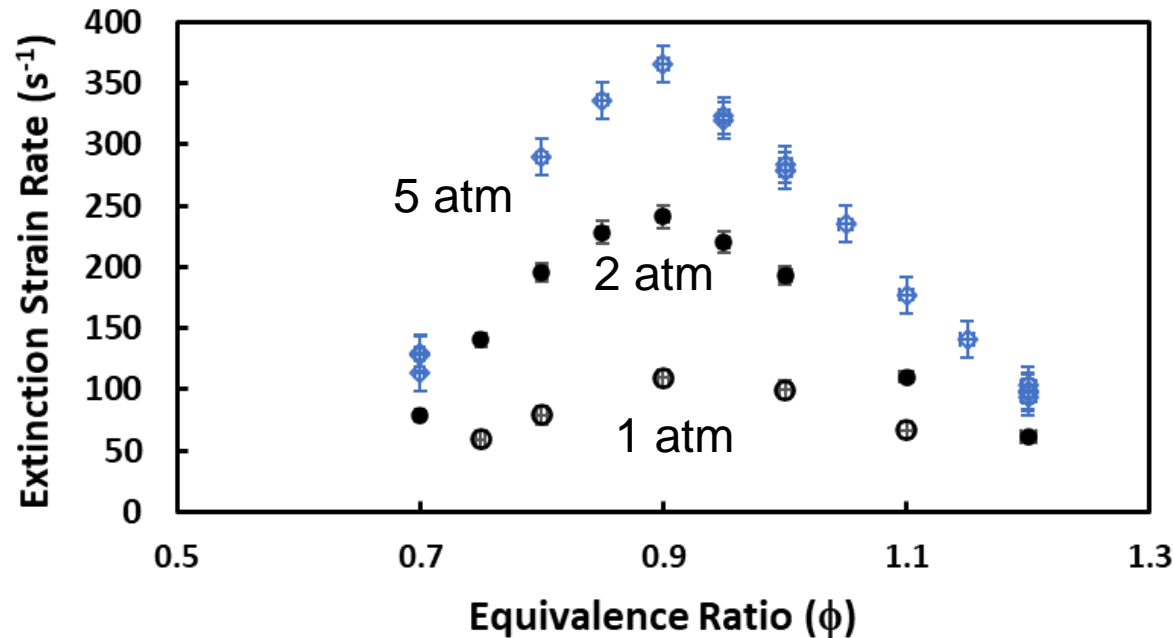
# Premixed Counterflow $\text{NH}_3/\text{Air}$ Flames: Effect of Pressure and Equivalence Ratio



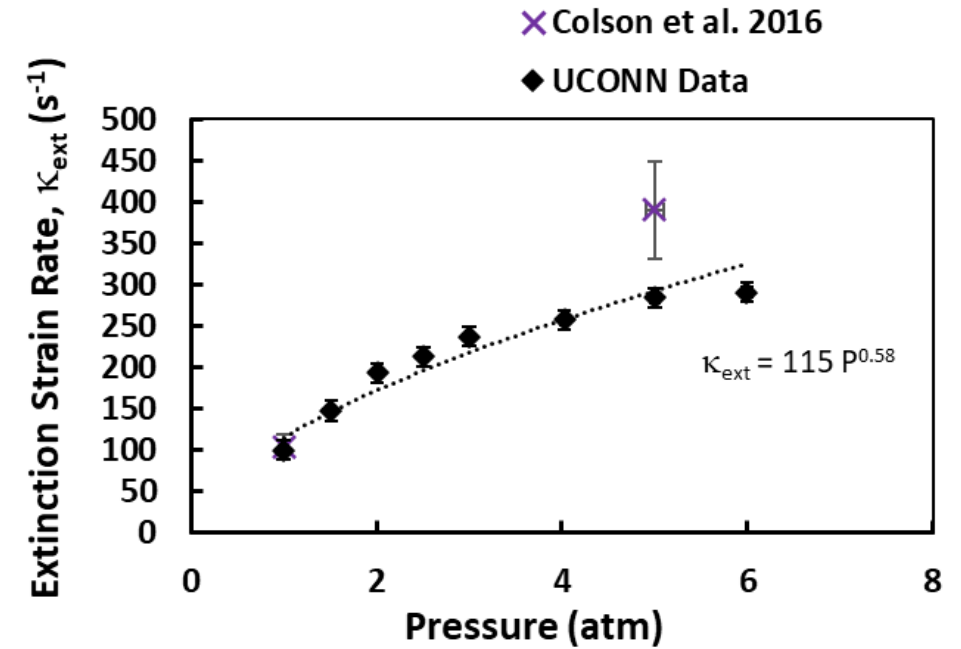
## UCONN Counterflow Extinction Experiments

- Initial reactant temperature  $T_i = 294 - 298 \text{ K}$

Effect of Equivalence Ratio on Extinction Strain Rate

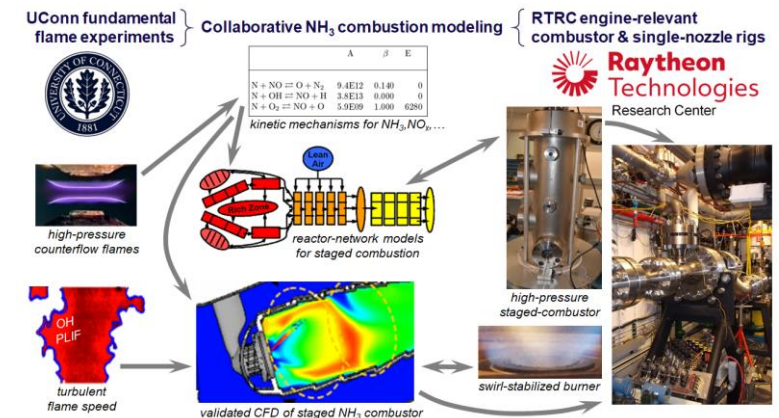


Effect of Pressure on  $\text{NH}_3\text{-Air}$  Flame Extinction ( $\phi = 1$ )



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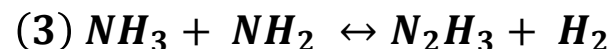
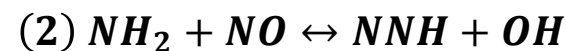
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# Chemical Kinetic Mechanisms

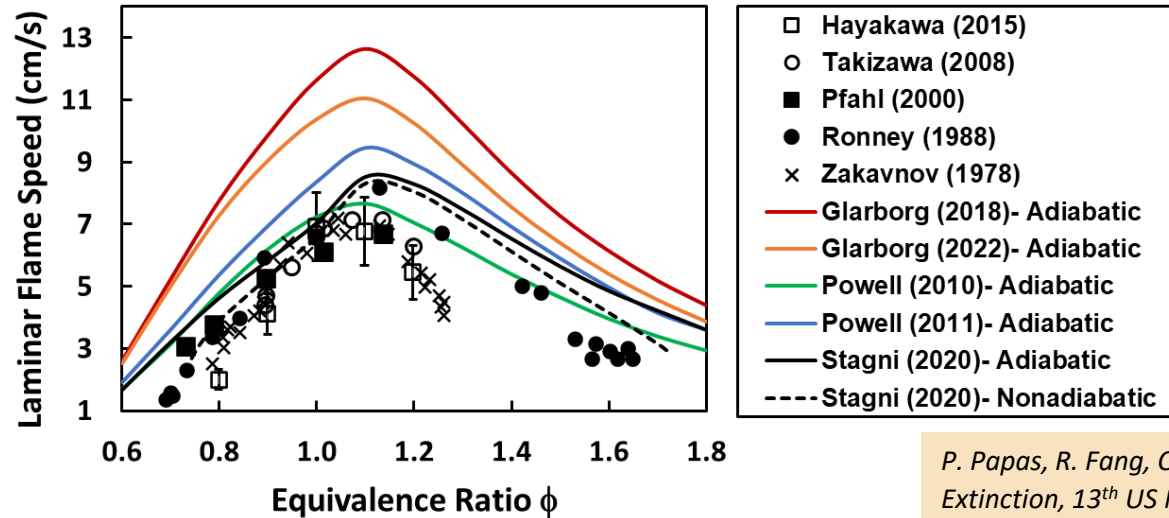
*Selected published, comprehensive N/H chemical mechanisms*

- Open-source computational framework developed by Dave Goodwin at Caltech  Cantera
- **Glarborg et al. Mechanism (2018 and 2022 versions)**  
[Glarborg, Miller, Ruscic, Klippenstein: Modeling nitrogen chemistry in combustion, Prog. Energy Combust. Sci. (2018) 31-68]  
[Glarborg: The NH<sub>3</sub>/NO<sub>2</sub>/O<sub>2</sub> system: Constraining key steps in ammonia ignition and N<sub>2</sub>O formation, Combust. Flame, Vol. 257 (2023)]
- **Stagni et al. Mechanism**  
[Stagni, Cavallotti, Arunthanayothin, Song, Herbinet, Battin-Leclerc, Faravelli: React. Chem. Eng. 5 (2020) 696–711]
- **Powell et al. Mechanism- (Referred to as “RTRC”)**  
[Powell, Papas, Dreyer: Hydrogen- and C<sub>1</sub>-C<sub>3</sub> Hydrocarbon-Nitrous Oxide Kinetics in Freely, Propagating and Burner Stabilized Flames, Shock Tubes, and Flow Reactors, Combust. Sci. Tech. 182 (2010) 252-283]  
[Powell, Papas, Dreyer: Flame Structure measurements of NO in Premixed Hydrogen-Nitrous Oxide Flames, Proc. Combust. Inst. 33 (2011) 1053-1062]
  - Two “RTRC” mechanisms differ by only 3 different rate expressions for amine radical reactions:



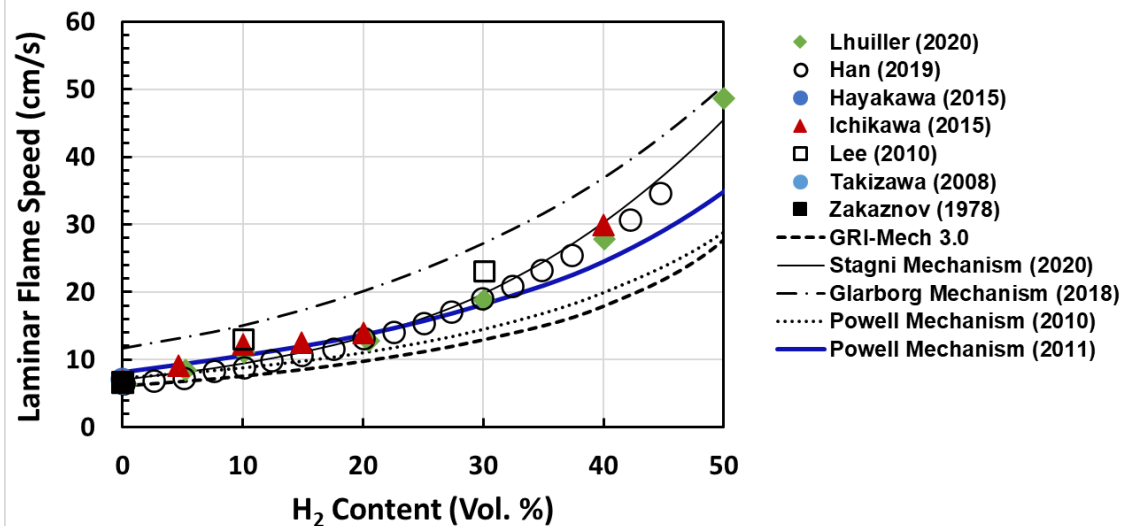
# N/H Chemical Kinetic Mechanisms

Comparison of selected published N/H chemical mechanisms with laminar flame speed data

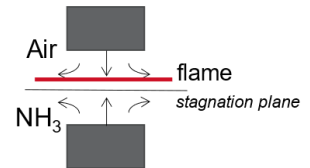
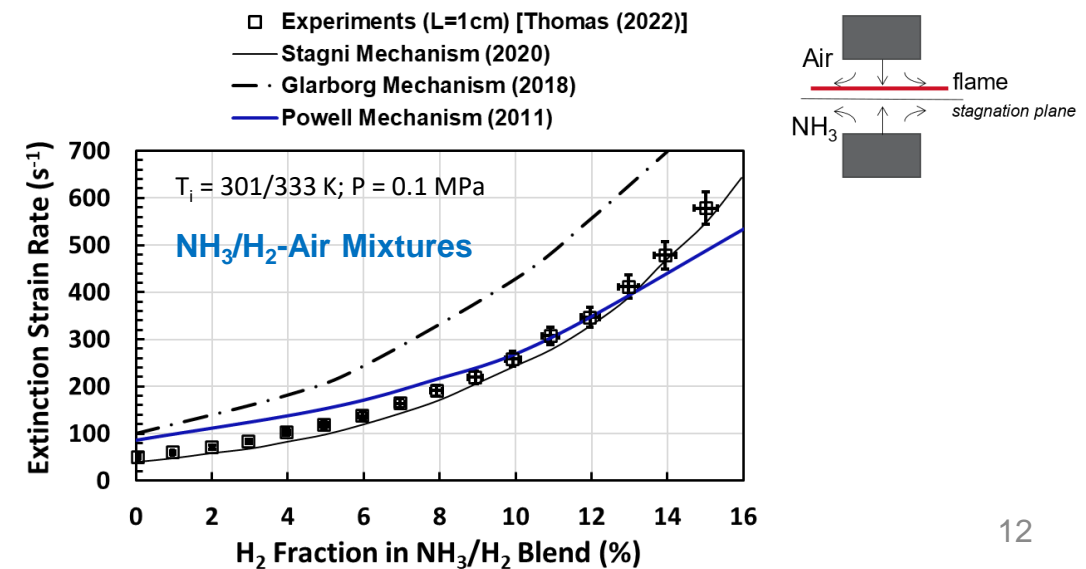


P. Papas, R. Fang, C.-J. Sung, L. L. Smith, J. F. Stevens An Assessment of Kinetic Models for Ammonia Flame Extinction, 13<sup>th</sup> US National Combustion Meeting, College Station, TX, March 19-22, 2023.

## Premixed NH<sub>3</sub> Flame Speed: Effect of added H<sub>2</sub>



## Non-Premixed NH<sub>3</sub> Extinction Strain Rate



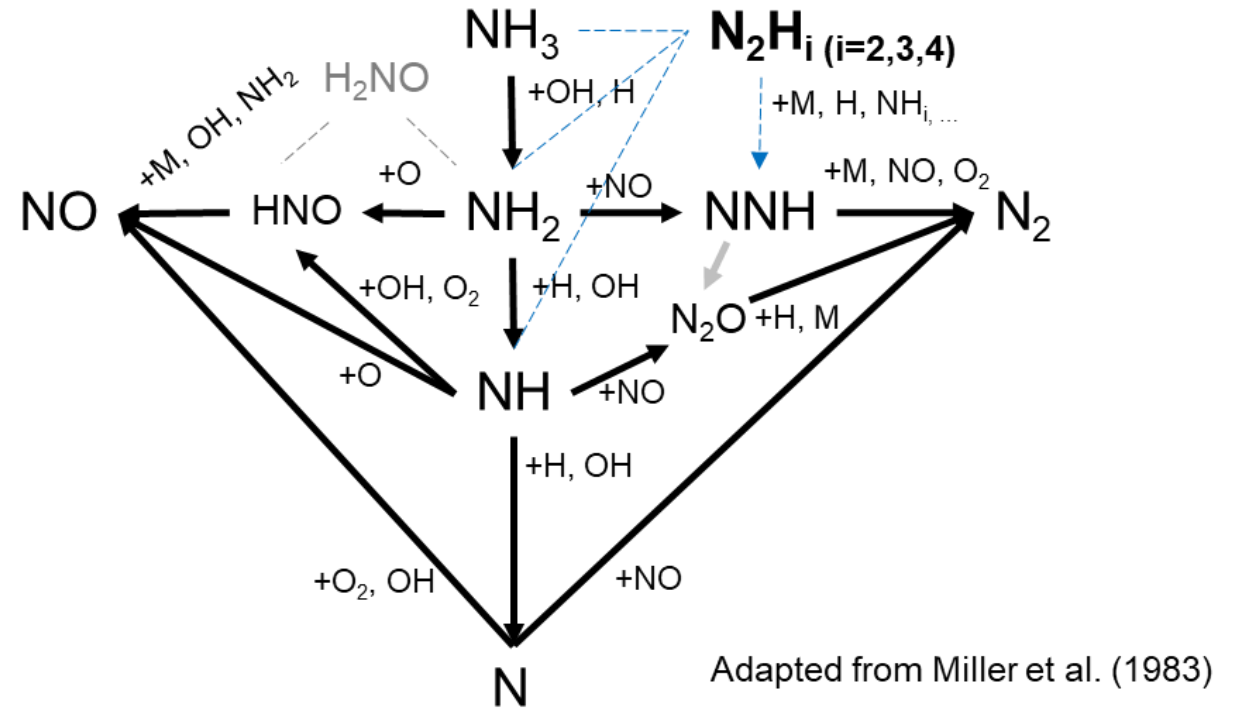


# Chemical Kinetic Mechanisms

## Example Data Sets used for “RTRC” N/H Mechanism Validation

Experiment	References
<b>Flow reactor studies</b>	
N <sub>2</sub> O decomposition	Allen et al. (1995)
H <sub>2</sub> / N <sub>2</sub> O/ H <sub>2</sub> O/ N <sub>2</sub>	Allen et al. (1998)
<b>Flame speed studies</b>	
H <sub>2</sub> / N <sub>2</sub> O/ N <sub>2</sub>	Powell et al. (2009)
CH <sub>4</sub> , C <sub>2</sub> H <sub>2</sub> , C <sub>3</sub> H <sub>8</sub> / N <sub>2</sub> O/ N <sub>2</sub>	Powell et al. (2009)
NH <sub>3</sub> / NO/ N <sub>2</sub>	Mei et al. (2020)
NH <sub>3</sub> / H <sub>2</sub>	Lhuiller (2020); Han (2019); Ichikawa (2015)
NH <sub>3</sub> / Air	Hayakawa (2015); Takizawa (2008)
<b>Flame structure studies</b>	
H <sub>2</sub> / N <sub>2</sub> O/ CO/ Ar	Vandooren et al. (1997)
NH <sub>3</sub> / N <sub>2</sub> O/ Ar	Venizelos and Sausa (1998; 2002)
<b>Shock tube and induction time studies</b>	
H <sub>2</sub> / N <sub>2</sub> O/ Ar	Hidaka et al. (1985); Mével et al. (2009)
CH <sub>4</sub> / N <sub>2</sub> O/ Ar	Drummond (1969); Soloukhin (1971)
N <sub>2</sub> O/ CH <sub>4</sub> / CO/ Ar	Dean and Johnson (1980)
<b>Counterflow studies</b>	
NH <sub>3</sub> /H <sub>2</sub> (Air)- Extinction	Thomas et al. (2022)

## Ammonia Oxidation Pathway Schematic



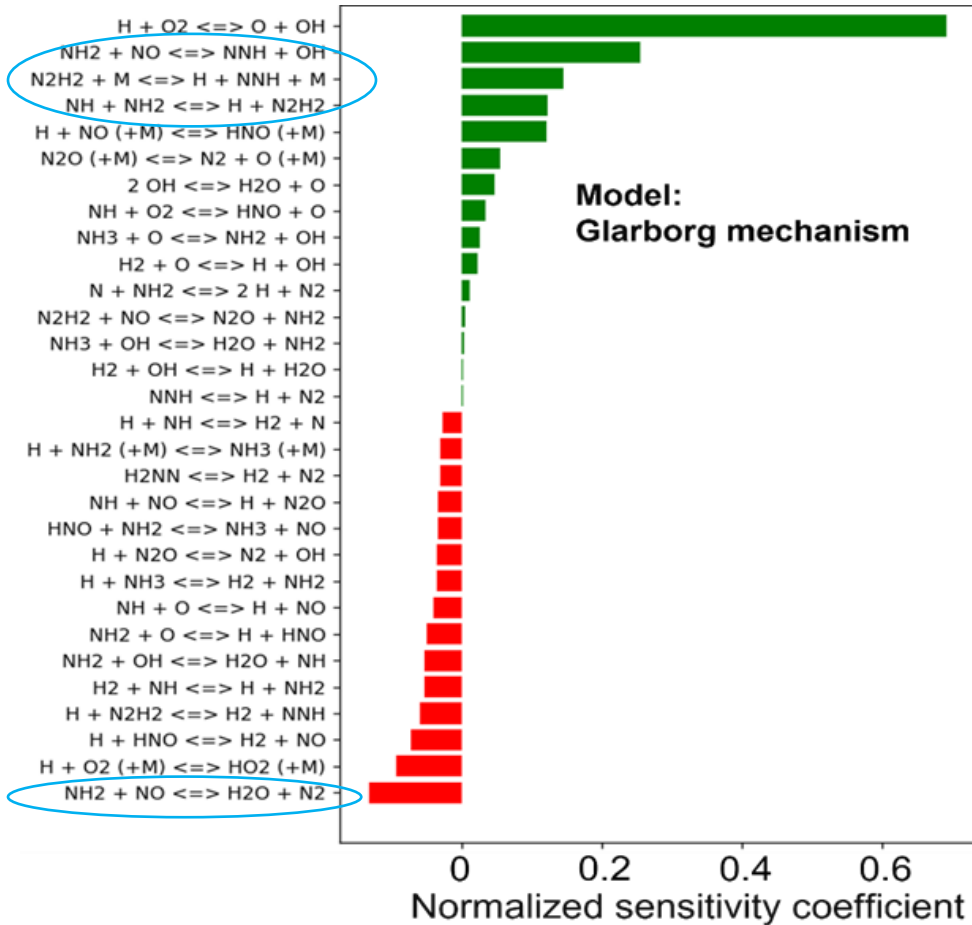
- Radical combinations involving NH<sub>i</sub> to form N<sub>2</sub>H<sub>i</sub> important under fuel-rich conditions & not well characterized



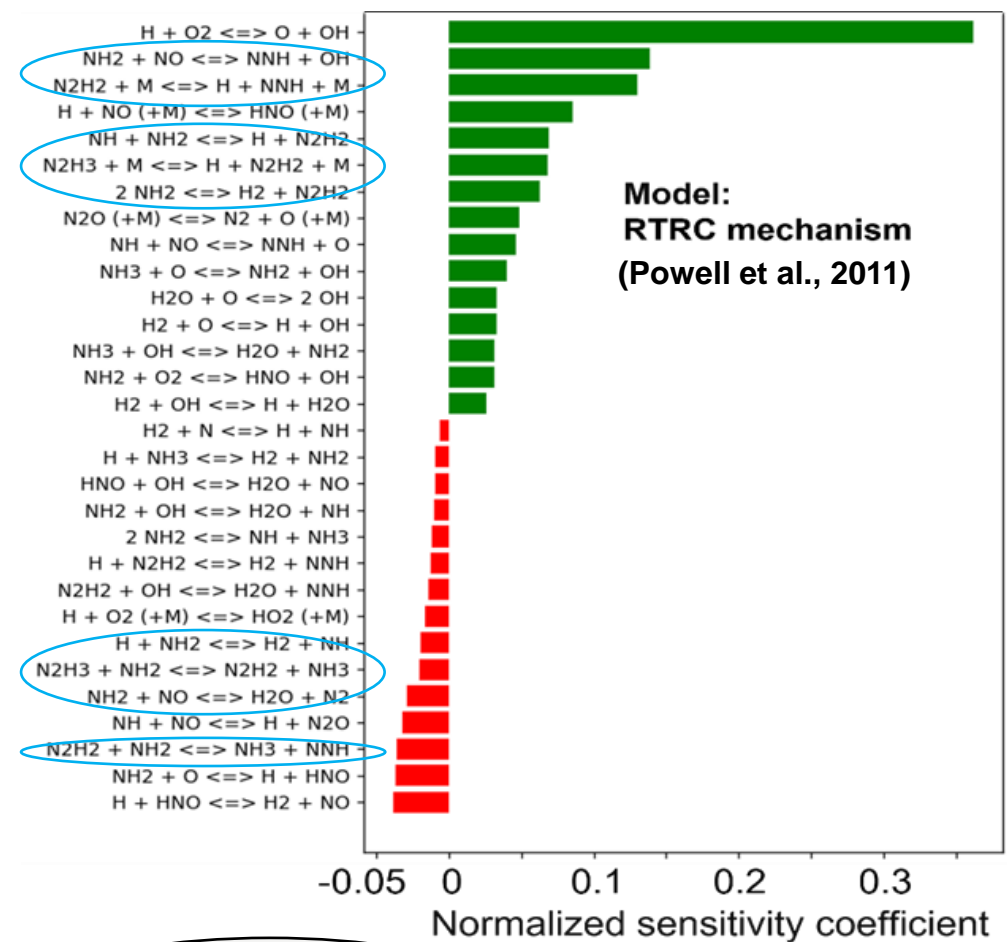
# Feature Sensitivity to Extinction Strain Rate

## *NH<sub>3</sub>/Air Counterflow Flame*

$T_i = 301/333$  K;  $P = 0.1$  MPa

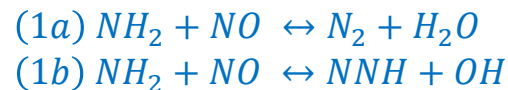


$T_i = 301/333$  K;  $P = 0.1$  MPa

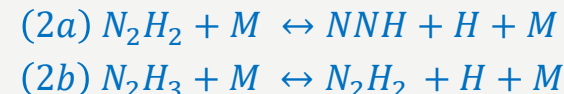


**Sensitive rate constants:**

### *NH<sub>2</sub>/NO Interactions*



### *NH<sub>i</sub>/N<sub>2</sub>H<sub>i</sub> Reactions*



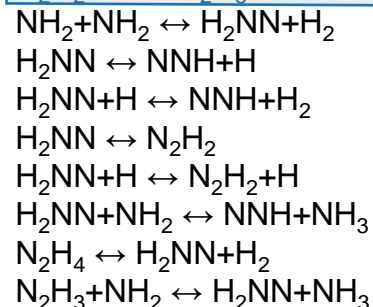
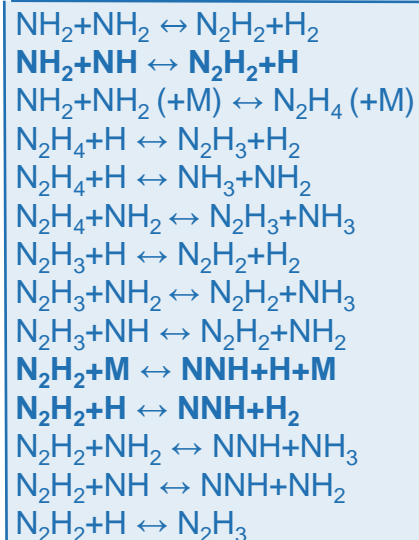
**Sensitive and Uncertain rate constants**



# Chemical Kinetic Mechanisms

Comparison of important  $N_2H_i/NH_i$  interactions between Glarborg, Stagni and RTRC mechanisms

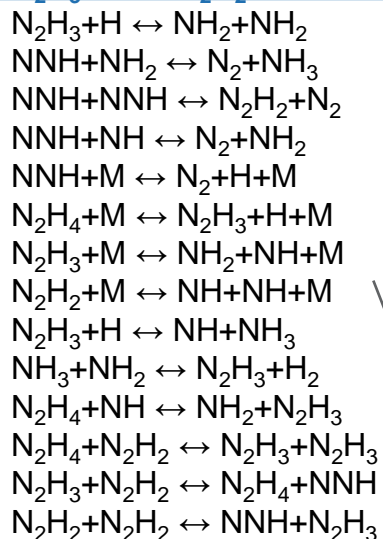
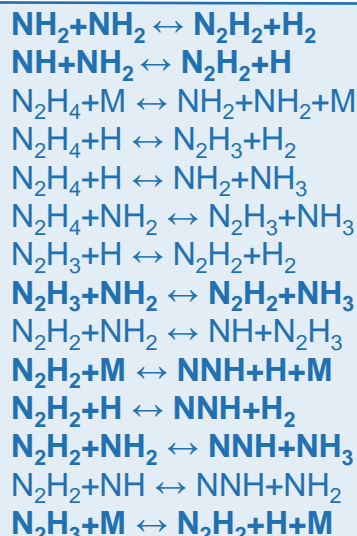
## Glarborg (2018)



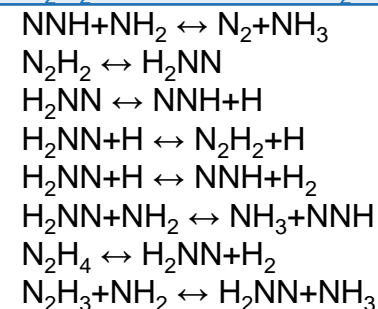
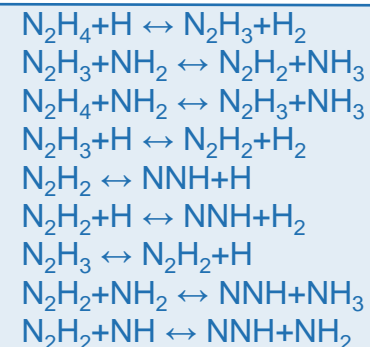
Pros: - Comprehensive N kinetics for NOx  
- Extended to low-temp. ignition chemistry

Cons: - Flame-speed predictions vs. data

## RTRC



## Stagni (2020)



Pros: - Flame-speed predictions match existing data

Cons: - Limited NOx data validation

Common  
reaction  
steps in all  
mechanisms

Pros: - Validated N/ O/ H chemistry for NOx, N<sub>2</sub>O  
- Flame-speed predictions match existing data

TBD: - Counterflow predictions → **Newly added UConn data** (for all mechs.)



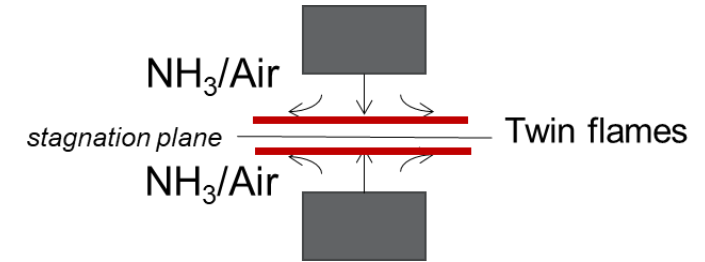
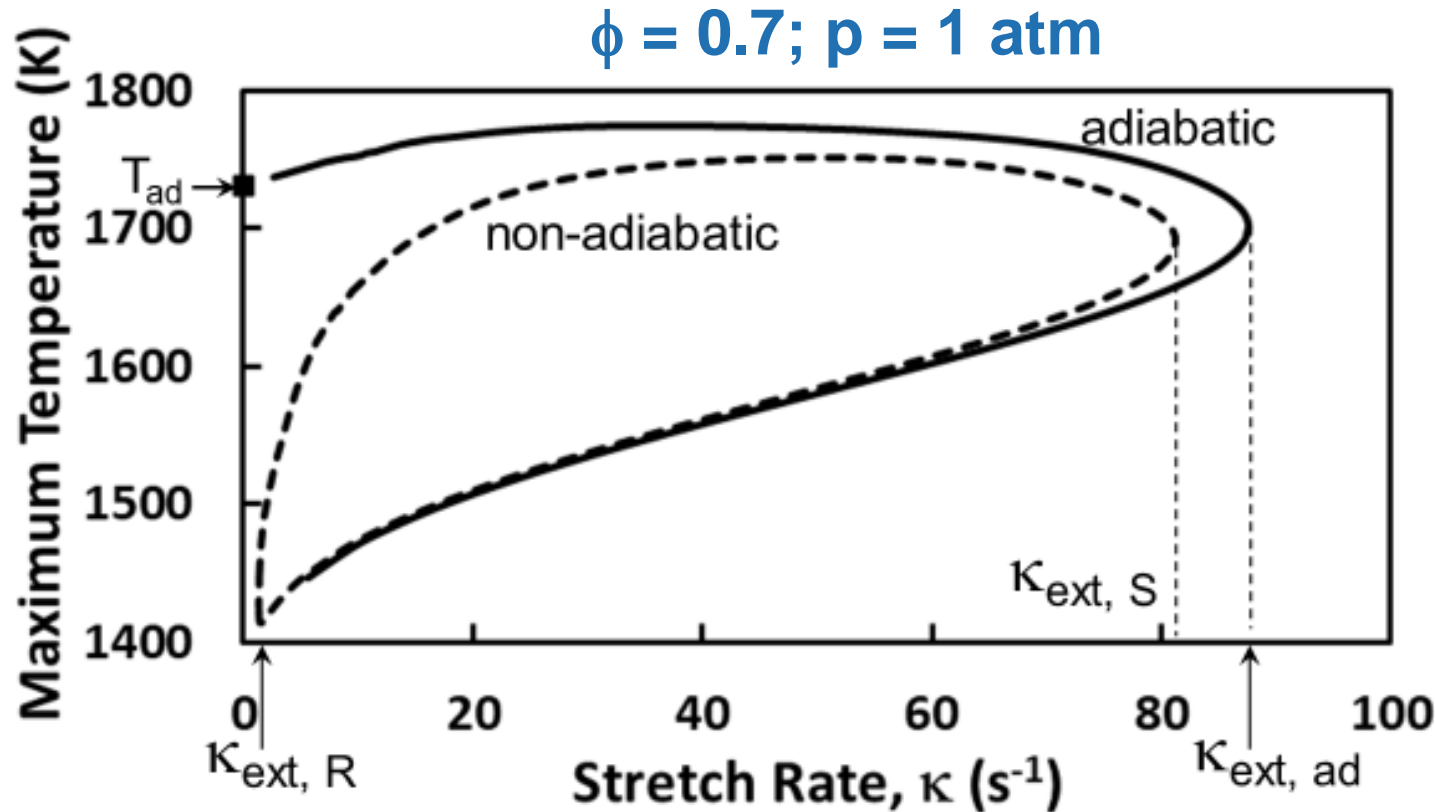
# Non-Adiabatic Premixed Ammonia Counterflow Flames

Modified Counterflow Model scripts to account for radiative heat loss

**Radiative heat loss per unit volume**

( $\sigma$  = Stefan-Boltzmann constant,  $\kappa_p$  is the Planck mean absorption coefficient;  $T_o$  = ambient unburnt reactant temperature,  $\kappa_p$  = total Planck's mean absorption coefficient)

$$q_r = -4\sigma\kappa_p(T^4 - T_o^4)$$



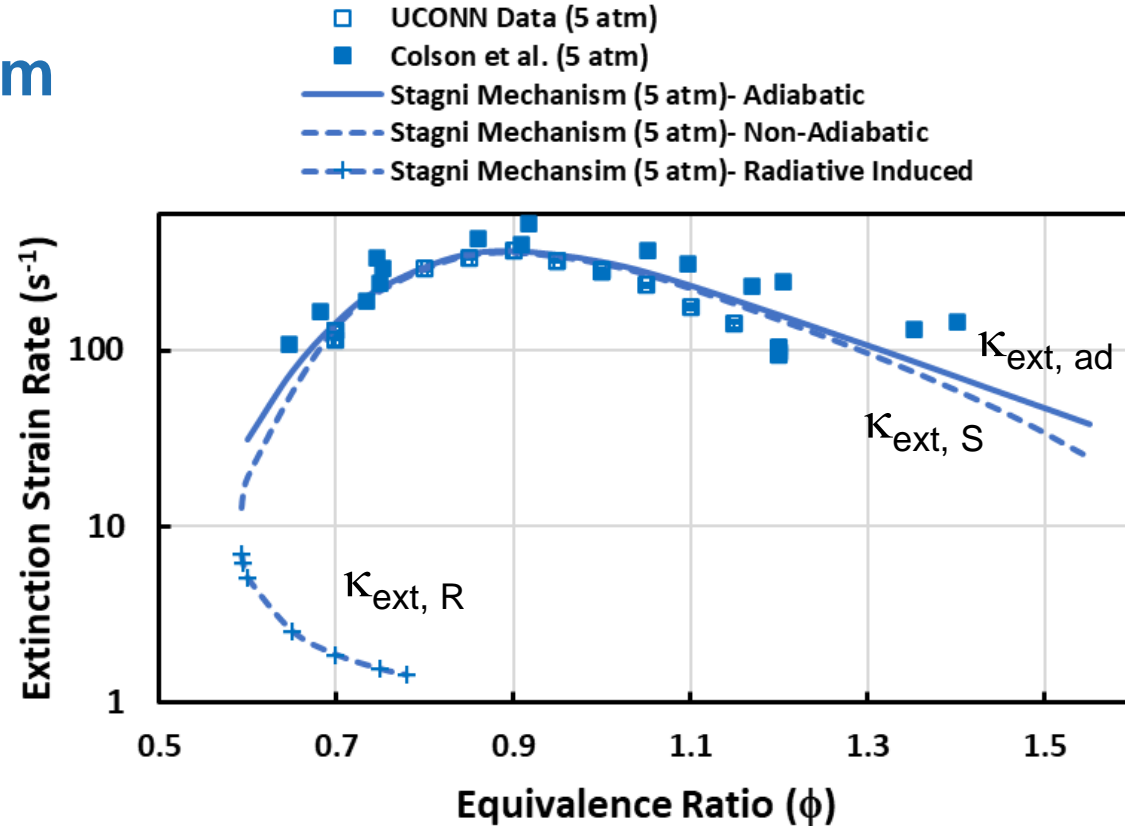
- Overall burning rate  $S \sim (Le^{-1} - 1)\kappa$
- $\text{NH}_3$  counterflow flames exhibit radiative-induced and stretch-induced extinction limits near lean flammability limit



# Non-Adiabatic Premixed Ammonia Counterflow Flames

Ammonia flames exhibit both stretch- and radiative-induced extinction states near lean flammability limit

$p = 5 \text{ atm}$

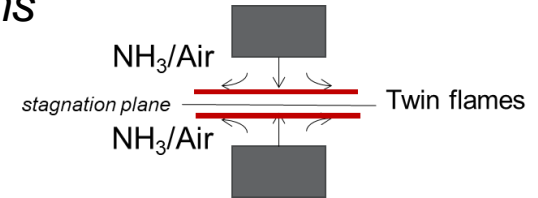


C-shaped curve for counterflow premixed fuel-lean, ammonia-air flames showing computed adiabatic stretch-induced stretch rate  $\kappa_{ext,ad}$  (solid line) as well as non-adiabatic stretch-induced stretch rates  $\kappa_{ext,S}$  (dashed line) and radiative-induced stretch rates  $\kappa_{ext,R}$  (+ symbols).

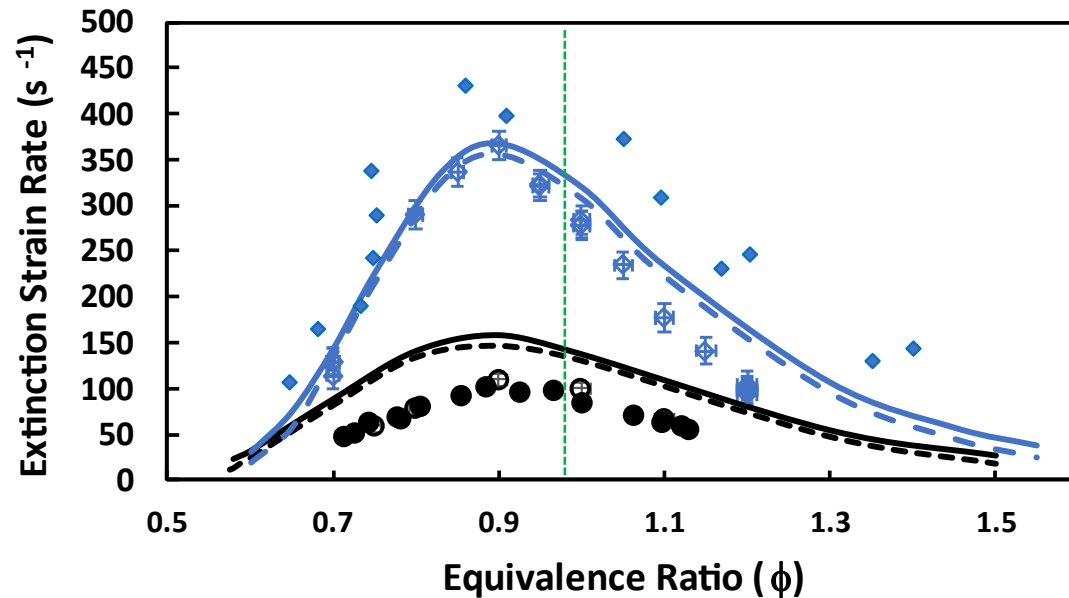


# Premixed Counterflow $\text{NH}_3/\text{Air}$ Flames: Effect of Pressure on Extinction Strain Rate

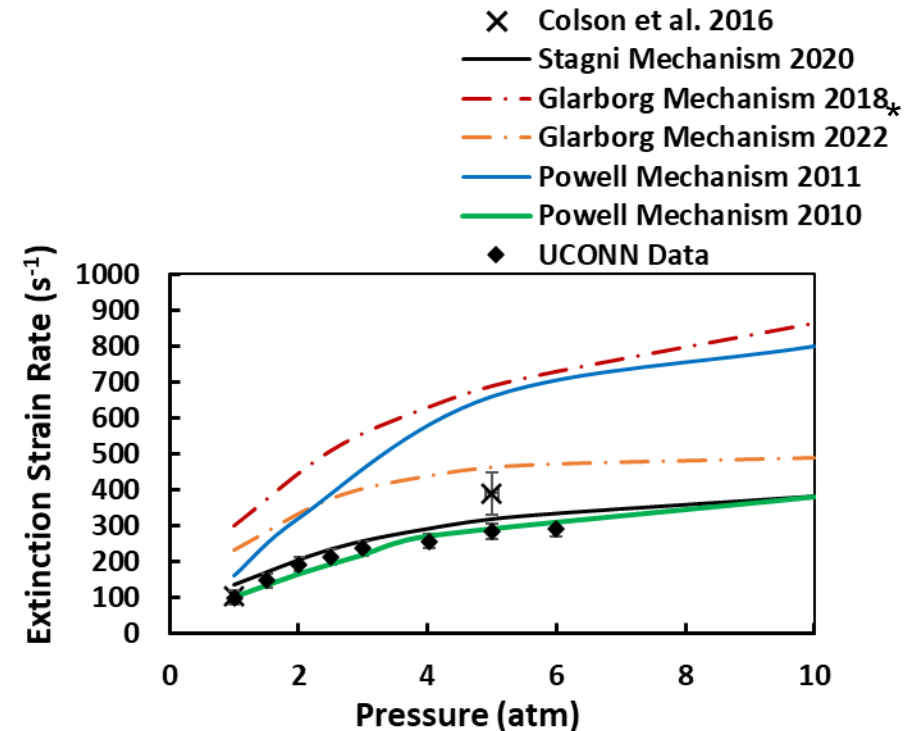
Comparison with available data from Colson et al. (2016) and N/H chemical mechanisms



- Colson et al. (1 atm)
- UCONN Data (1 atm)
- ◆ Colson et al. (5 atm)
- ◇ UCONN Data (5 atm)
- Stagni Mechanism (1 atm; Adiabatic)
- - - Stagni Mechanism (1atm; Nonadiabatic)
- Stagni Mechanism (5 atm; Adiabatic)
- - - Stagni Mechanism (5 atm; Nonadiabatic)



## Stoichiometric Premixed $\text{NH}_3/\text{Air}$ Flames

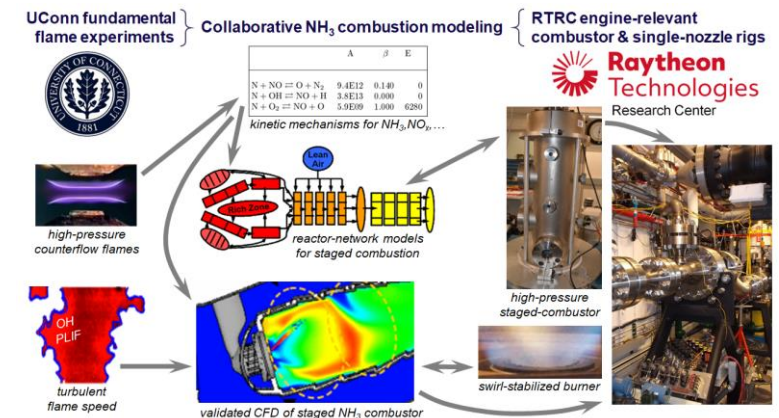


\* Glarborg, Combust. Flame, Vol. 257 (2023)



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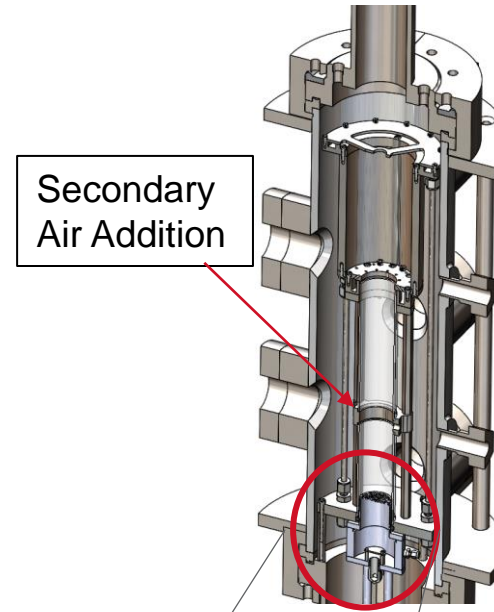
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# High-Pressure Staged Reactor Rig (RTRC) – NO<sub>x</sub> Evaluations

## Capability of High-Pressure Reactor

- 30+ atm
  - Vessel shell rated >500 K
  - Coflow / purge enables higher preheat & flame temperatures
  - Staged reaction-zones for fuel-rich & fuel-lean measurements
- 
- Planned diagnostic techniques
    - Extractive probe with in-line analyzers for species measurements (NO, NH<sub>3</sub>, ...)



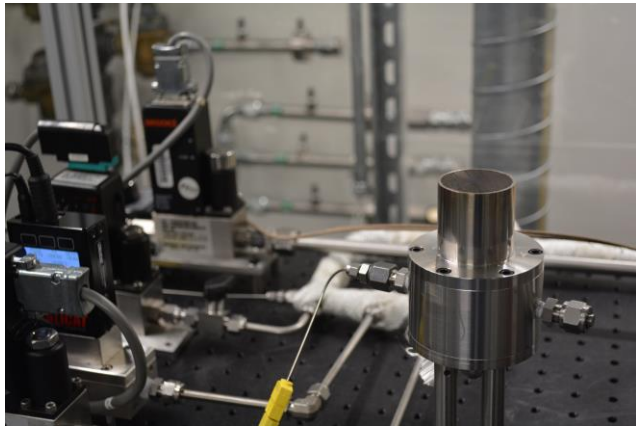
pump liquid NH<sub>3</sub> to >500psi



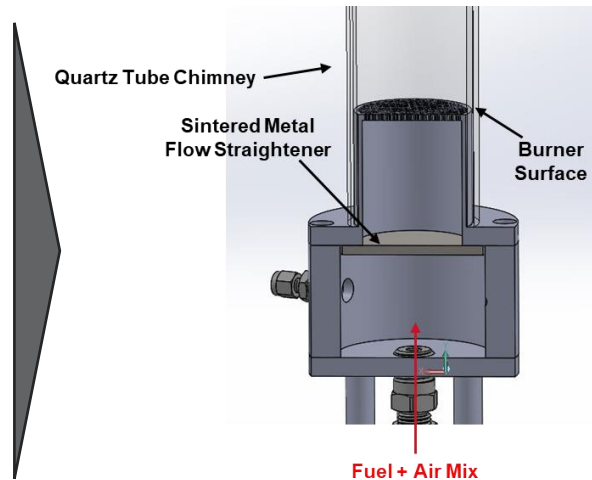


# RTRC High-Pressure Staged Flame Reactor Rig

- High-pressure burner rig modifications for  $\text{NH}_3$  compatibility
  - Removal of copper components from inside pressure vessel & burner
  - Modified burner insert for  $\text{NH}_3$  operation
  - For staged-burner testing (RQL), chimney modification for staged air addition

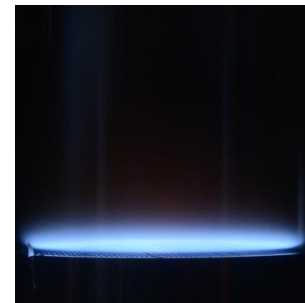


Shakedown Burner Testing Set-up

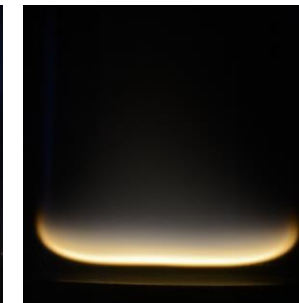


*Flat Flame Burner*

$\text{CH}_4$ -Air

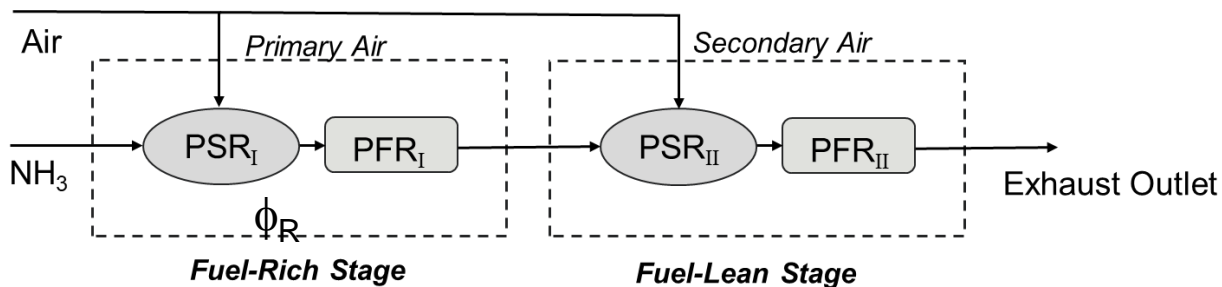


100%  $\text{NH}_3$ -Air

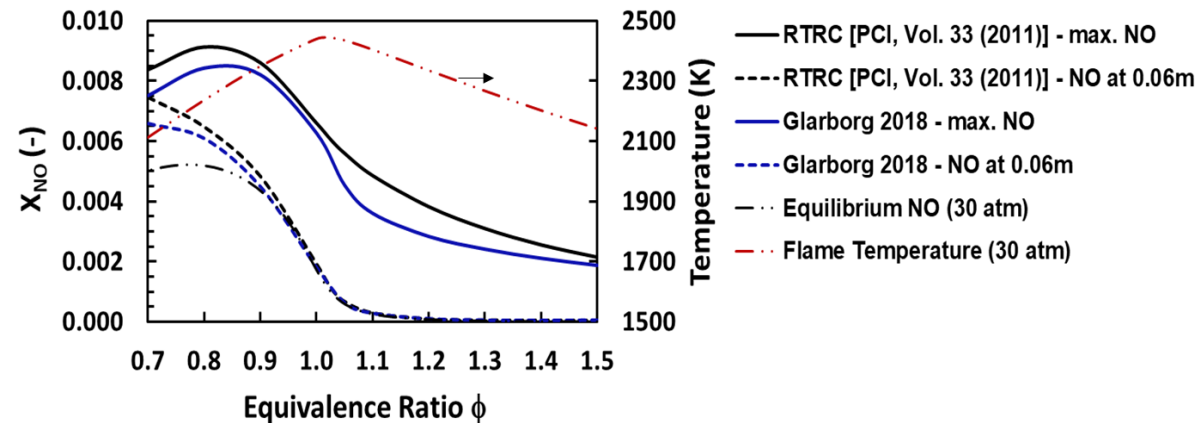


# Chemical Reactor Network (CRN) Modelling

## CRN Model Schematic for RQL Combustor



## Premixed Flame (30 atm)



- Validation against available simulation data from Li et al., Fuel 355 (2024) 129509.
- Overall “theoretical” NOx levels <30 ppm for a RQL architecture appear feasible**
- Established N/H mechanisms show wide variability for NOx

$$\phi_R = 1.25$$

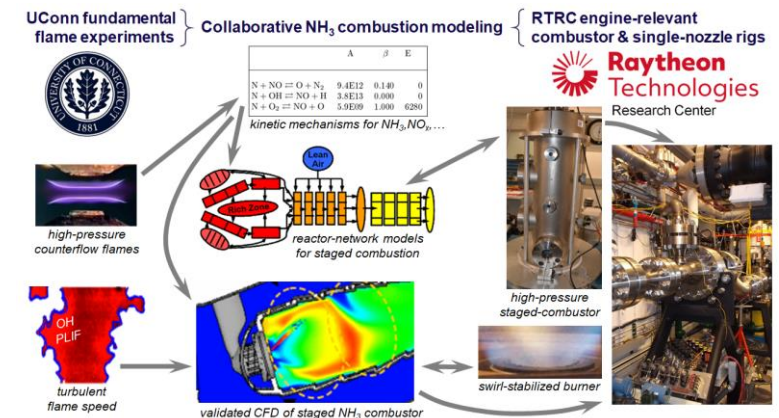
Total Residence time $\tau$ (ms)	$\tau_{\text{PSR}_I}$ (ms)	$\tau_{\text{PFR}_I}$ (ms)	$\tau_{\text{PSR}_{II}}$ (ms)	$\tau_{\text{PFR}_{II}}$ (ms)	Pressure (atm)	Inlet Temp. (K)	Outlet Temp. (K)	NOx* (ppm)
20	3	14	2	1	12	600	1850	38.5
20	3	14	2	1	30	700	1850	30.1
30	3	24	2	1	30	700	1850	22.9

\* 15% O<sub>2</sub> dry



# Outline

- Project Objectives & Partnership
- Counterflow Flame Studies – Reaction Rates for NH<sub>3</sub> Combustion
  - Experiments at 1 – 5 atm pressure (UConn)
  - Kinetic modeling in 1-D flames w/ radiation (RTRC)
- Burner Studies – NO<sub>x</sub> Emissions
  - Flat-flame burner experiments (30 atm pressure staged reactor rig)
  - Chemical Reactor Network (CRN) modeling of GT combustor
- Next Steps – Year-2 plans
  - UConn experiments incl. new rig
  - RTRC high-P experiments & modeling



# Next Steps: 2024 Experiments

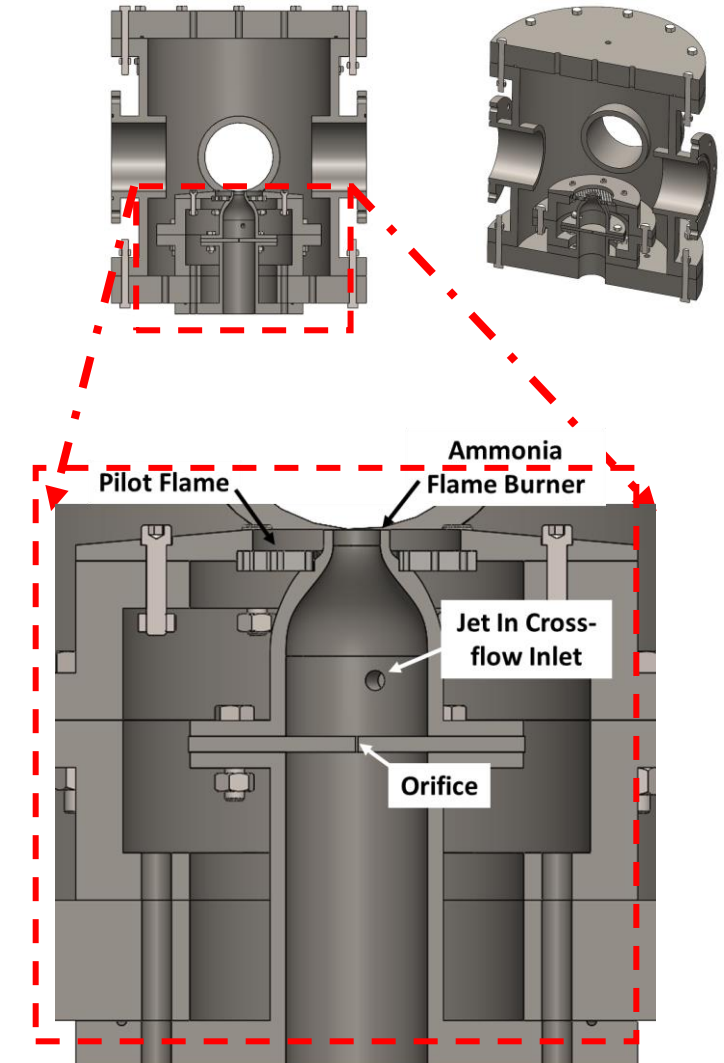
- UConn Counterflow testing continuing with preheat
- UConn Turbulent Flame Rig
  - Turbulent flame speed measurements (NH PLIF for flame surface area)
  - Effects of pressure, equivalence ratio, and turbulent intensity (Turbulent intensity range: 15–25%)
  - Pressure range: 1–7 atm; up to 500 K preheat



- **RTRC High-Pressure Staged Reactor Rig**

- System pressure up to 30 atm
- Air preheat up to 800K
- NO<sub>x</sub> measurements using probe extraction technique
- FTIR Analysis and other diagnostics

## UConn Turbulent Flame Rig



# Questions?

Please e-mail any questions to:

Lance Smith (lance.smith@rtx.com) or  
Paul Papas (paul.papas@rtx.com)



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