

EWI Project No. 55232GTH Annual Review March 22, 2017

Additive Manufacturing of Fuel Injectors

NETL – 2017 Crosscutting Research and Rare Earth Elements Portfolios Review

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The Project Team



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A Caterpillar Company

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EVVI We Manufacture Innovation

A Caterpillar Company

Process Equipment Overview

Laser PBF EOS M280



Laser PBF – Open Architecture EWI-Designed and Built



Electron Beam PBF Arcam A2X



Binder Jetting ExOne Innovent



Material Extrusion Stratasys Fortus 450mc



Sheet Lamination UAM Fabrisonic



We Manufacture Innovation

Process Equipment Overview

Laser and Arc-DED Commercial Robot



Laser DED RPM 557



EB-DED Sciaky EBAM



Targeting \$10-\$15M AM equipment investment in Buffalo





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Solar Turbines Overview

- World's Largest Manufacturer of Industrial Gas Turbines (1 to 22 MW)
- Over 15,000 Gas Turbines Sold
- Over 6,000 Gas Compressors Sold
- Installations in over 100 Countries
- Direct End-to-End Sales & Service
- More than 2 Billion Fleet Operating Hours
- Global Workforce (7,000) Employees
- Based in San Diego, California, U.S.A.
- Subsidiary of Caterpillar Inc. Since 1981





Solar personnel living around the globe provide services and support that enhance the communities in which they live and work.





Energy... is fundamental to sustaining life, powering productivity, and conducting countless daily activities.





Solar Turbines

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Motivation

Gas turbine components

- Very specific design (difficult to cast)
- Long lead time
- Fuel injector tip
 - Alloy X
 - Ni-Cr-Fe-Mo alloy
 - Solid Solution Strengthened





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Objective

Objective:

- To develop a novel process to qualify the AM technique of laser powder bed fusion (L-PBF) for complex gas turbine components made of high temperature nickel-based alloys
- To investigate the effect of input powder stock and AM process variables on resultant microstructure and mechanical properties for the alloy material
- Post-processing, including heat treatment and the use of finishing technologies will also be employed in order to achieve required dimensional and surface finish requirements for the component.



Relevance to Fossil Energy

- Alloy-X is used in many industrial gas turbine applications.
- AM will enable design and energy efficiencies:
 - Faster and less costly design optimization.
 - Future applications could enable more energy efficient designs by reducing design constraints
 - Increasing fuel efficiency
 - Providing higher operating temperature



Milestone 1: Powder Characterization

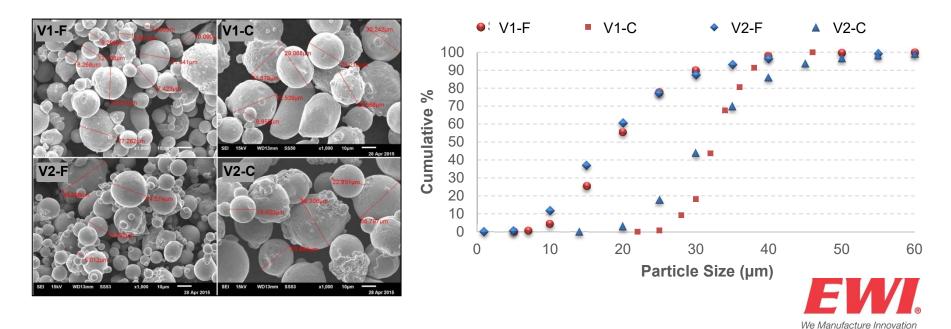
- Complete
- Milestone 2: NIST Test Artifacts Complete
 - Complete
- Milestone 3: Process Parameter Report Delivered
 - Complete
- Milestone 4: Property Data Curves Delivered
 - In-progress
- Milestone 5: Specification Document Delivered



Milestone 1- Powder Evaluation

Powder evaluation:

Vendor	Туре	Min. Desired Size (µm)	Max. Desired Size (μm)	Fine (%)	Coarse (%)	Cost Comparison per lb. (350 lb order)
V1	Fine	5	38	0.1% < 5 um	0.8% > 38 um	100%
V1	Coarse	20	45	4.2% < 20 um	0.5% > 45 um	132%
V2	Fine	5	38	2% < 5.5 um	1 > 38 um	190%
V2	Coarse	16	45	1% < 16 um	1% > 45 um	195%



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Milestone 2- NIST Test Artifacts Complete

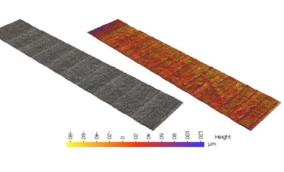
Minimum Feature Size:

 The L-PBF process was capable of producing fine features, and met capabilities of investment casting

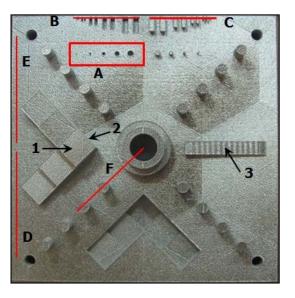
Surface roughness measurement:

- Fine powders were slightly better (S_a).
- Typical allowable limit for the surface finish of investment casting

 \circ (Ra) less than 125 µin. (3.17 µm)







Milestone 3- Process Parameter Report Delivered

Stress relief heat treatment:

- All of the specimens underwent the stress relief heat treatment, while still attached to the build plate
 - 2150°F
 - 1 hour
 - Rapid argon cooling.



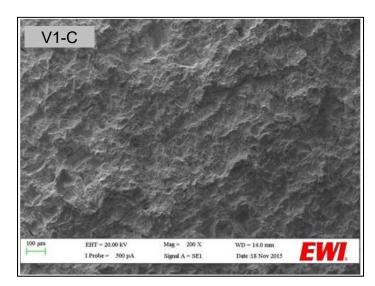
Mechanical test:

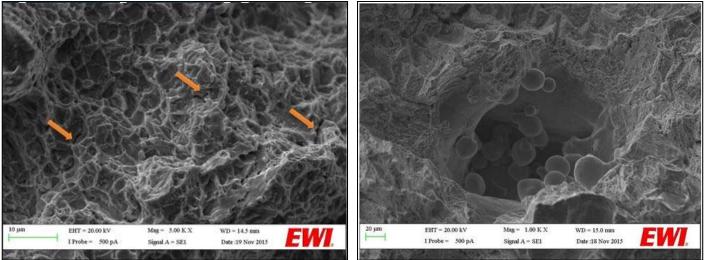
- Tensile
 - Room temperature
 - Elevated temperature (1500°F/815.5°C)
- Creep
 - Elevated temperature (1500°F/815.5°C)
 - Stress: 15 ksi
- Low cycle fatigue
 - Elevated temperature (1000°F/538°C).
 - Total strain range: 0.6%
 - Stress ratio: -1



Fractography:

- Room temperature tensile test
 - Intergranular fracture morphology
 - Dimples on the fracture surfaces
 - Secondary cracking
 - LOF surrounded by un-melted powder particles.







Powder down select:

Powder / Cos	Cost	Powder Compatibility with	Microstructure	RT-	Tensile	Test		ensile est	Creep		Fatigue
Properties	Comparison	AM Machine (ProX300)	(Micro cracks)	UTS	YS	EI%	UTS	EI%	Hrs (rpt.)	El% (rpt.)	Cycles
V1-C	100%										
V1-F	132%							No HT			
V2-C	190%										
V2-F	195%										

- Vendor 1
 - V1-C:
 - Lowest creep and fatigue properties
 - Powder leakage
 - V1-F
 - Low ductility at high temperature as well as the short creep rupture time could be improved using a proper heat treatment
 - Originally developed with the OEM.
 - Favorable powder cost



Powder down select:

Powder / Cost	Powder Compatibility with	Microstructure	RT-	RT- Tensile Test			ET- Tensile Test		Creep		
Properties	Comparison	AM Machine (ProX300)	(Micro cracks)	UTS	YS	El%	UTS	EI%	Hrs (rpt.)	EI% (rpt.)	Cycles
V1-C	100%										
V1-F	132%							No HT			
V2-C	190%										
V2-F	195%										

- Vendor 2
 - V2-C
 - Powder leakage
 - V2-F
 - Similar or better tensile properties than those of V2-C
 - Longer fatigue life
 - Lower creep life.
 - Unfavorable powder cost
 - Unfavorable microstructure (micro-cracks)



Heat Treatment Optimization (Microstructure Screening)

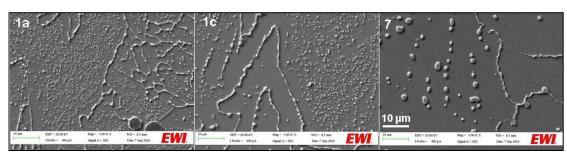
- The team investigated eight HTs.
 - As printed (no HT)
 - Solution Annealing
 - Temperature (#2)
 - Time (#5)
 - -HIP
 - Post annealing
- Microstructural screening to downselect to four heat treatments for mechanical testing

e	Solut	ion Anneali	HIP				
HT Procedure	Temp. (°F)	Time	Post Cooling	Temp. (°F)	Pressure (MPa)	Time (hrs)	
1-a	2150	15 min.	AC				
1-b	2150	40 min.	AC				
1-c	2150	1 hr	AC				
2				2150	100	4	
6*	2150	4 hr	AC				
7	2150	8 hr	AC				
8	2200	1 hr	AC				
AC: Arg	gon cooling	9					

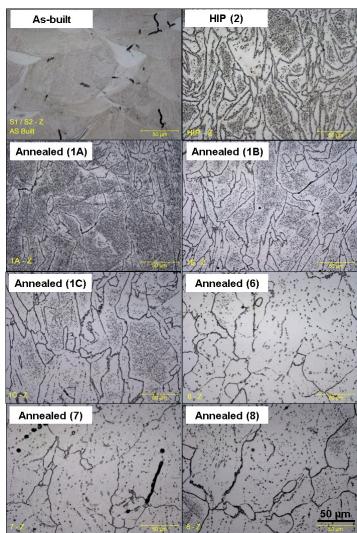


Heat Treatment Optimization

- Annealing time
- Annealing temperature
- External pressure (HIP)
 - Grain growth
 - Dissolution of precipitates



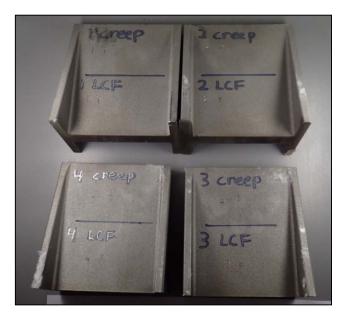
1a, 1c, and 7 had an annealing temperature of 2150°F, but with the annealing times of 15, 60, and 240 minutes



Heat Treatment Optimization (Mechanical testing)

- Additional test walls were produced.
- Four heat treatment procedure were selected for the further analysis of mechanical properties.
 - Room temperature tensile test
 - Elevated temperature tensile test
 - Low cycle fatigue (LCF)
 - Creep

edure	Locedure Procedure (hr) b b				HIP		Solution Annealing			
HT Proc	Temp. (°F)	Time (hr)	Post Cooling	Temp. (°F)	Pressure (MPa)	Time (hrs)	Temp. (°F)	Time	Post Cooling	
R1	2150	1	AC							
R2				2150	100	4				
R3				2150	100	4	2150	1	AC	
R4	2150	4	AC							

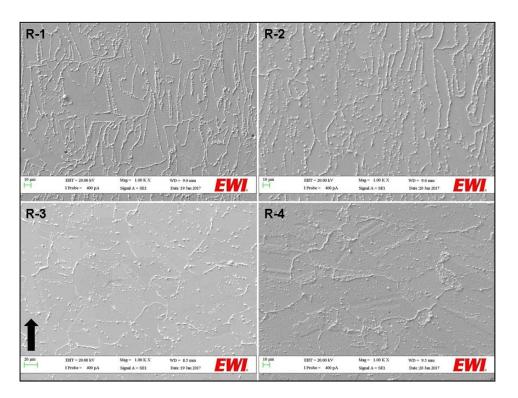


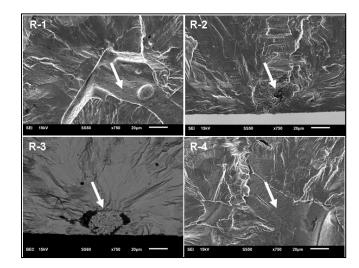


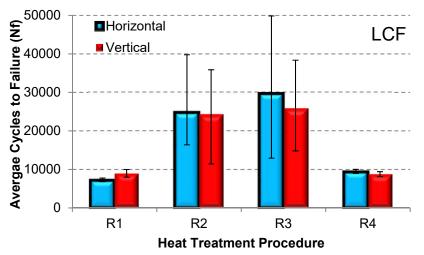
AC: Argon cooling

LCF

- Longer fatigue life in the HIP'd samples
 - Reduction of internal defects
 - Higher volume fraction of intergranular precipitates

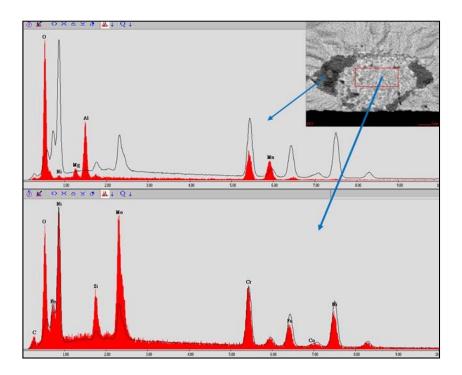


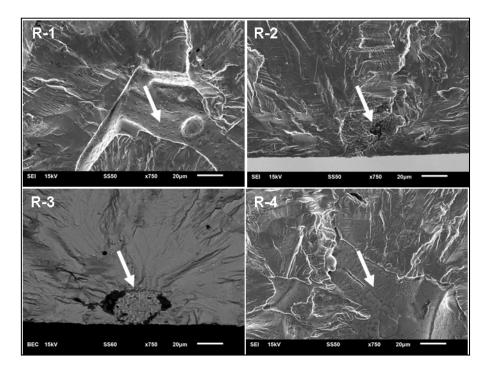




LCF

- Small LOF and solidification crack discontinuities
- Non-metallic SiMo and AlMnMg inclusions
 - New crack initiation mechanism
 - Scattering in the results





Milestone 3: Optimized Heat Treatment Down-Select

- HIP improved both LCF and Creep properties
- Post HIP reduced creep performance
- R2 selected for Milestone 4 activities

dure			HIP		Solution Annealing			Mechanical Properties			
HT Proced	Temp. (°F)	Time (hr)	Post Cooling	Temp. (°F)	Pressur e (MPa)	Time (hrs)	Temp. (°F)	Time	Post Cooling	LCF	Creep
R1	2150	1	AC								
R2				2150	100	4					
R3				2150	100	4	2150	1	AC		
R4	2150	4	AC								

AC: Argon cooling



Milestone 4- Property Data Curves Delivered

Builds

- Complete

Heat treatment

- Complete

Mechanical Testing

In-progress



Future Work

Milestone 4

- Generation of Property Data Curves
 - Room Temperature tensile
 - Elevated temperature tensile
 - -LCF
 - Creep

Milestone 5

- Development of the specification document
 - Powder
 - Process Parameters
 - Design

Final report



Summary

- The influences of powder feed on the dimensional accuracy and mechanical properties of additively manufactured Hastelloy X were analyzed.
 - One powder was down selected.
 - Process parameters were optimized.
- Heat treatment procedure was optimized for the additively manufactured Hastelloy X.
 - Eight heat treatment conditions were analyzed.
 - Microstructural and mechanical analysis were performed.
 - Improvements in LCF & Creep properties demonstrated

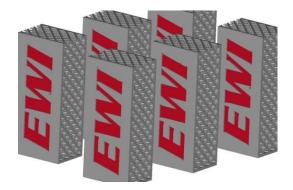


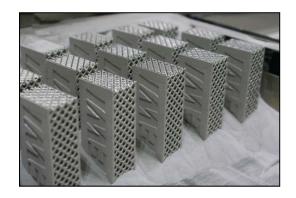


Questions

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http://ewi.org/technologies/additive-manufacturing/







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