



University Turbine Systems Research (UTSR) Workshop
Atlanta, Georgia, November 3-5, 2015



Turbomachinery Technology Development at FTT

3 November 2015



Jim Downs
Director of Technology
Florida Turbine Technologies, Inc.
561-427-6250
jdowns@fttinc.com
www.fttinc.com

Outline



- FTT company overview
- DOE-sponsored turbomachinery technology programs at FTT
- Opportunities for collaboration with UTSR universities

OEM Capability – Innovation Emphasis



Independent Know-How of Industry Systems (Industrial and Aero)

- Design practice, design criteria, and lessons learned
- Fast iteration conceptual design
- Detailed design know-how
- Expert knowledge of material systems and manufacturing processes

Calibrated Analysis Tools and Design System

- Internal and external software
- Continuously improved with new information/data

Technology Portfolio

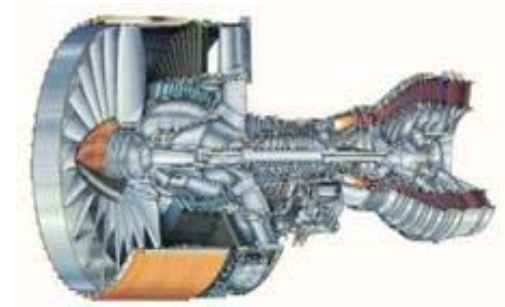
- Intellectual property
- Government-sponsored technology programs

Proven Supply Chain for Prototype Hardware

- Manufacturing engineering to support delivery

Validation and Field Experience

- Production as-manufactured hardware verification
- Field issue root cause
- Power producer engineering support



DOE-Sponsored Technology Programs



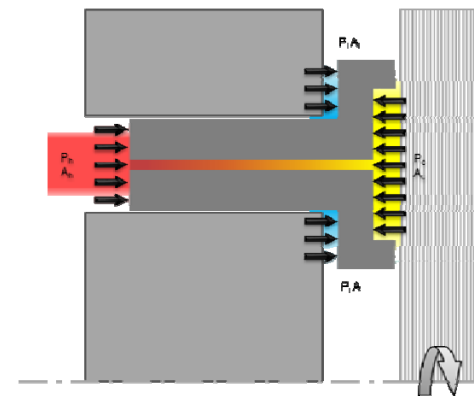
- TurboGT™ Gas Turbine with ARTICReturn™ Cooling (FE-0023975)

- 18 month Phase I program
Wraps-up early next year



- Air Riding Seal Technologies for Gas Turbines (DE-SC0008218)

- 2 year program
Concludes next year



DOE Advanced Turbines Program



R&D Activities to Accelerate Turbine Performance and Efficiency Beyond Current SOA and Reduce the Risk to Market for Novel and Advanced Turbine-Based Power Cycles

Goals:

- Efficiency – High, 65% (LHV) or Greater in Combined Cycle
- Operability – Support Load Following Capabilities to Meet Demands of Modern Power Grid
- Higher Firing Temperatures (1700°C/3100°F)
- Emissions – Low NO_x and other Criteria Pollutants
- Cost – ? ... FTT Primary Goal = Lower Cost

Ref: DOE NETL Web Site

November 3, 2015

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Path to High Efficiency Goal



Combined Cycle Power Plant

A lot of pieces, but the efficiency is predominantly driven by the gas turbine(s)

Ref:<http://tse1.mm.bing.net/th?id=OIP.Meed81da2ead7c689b2dbf71a16d1f9b6o0&w=300&h=300&c=0&pid=1.9&rs=0&p=0>

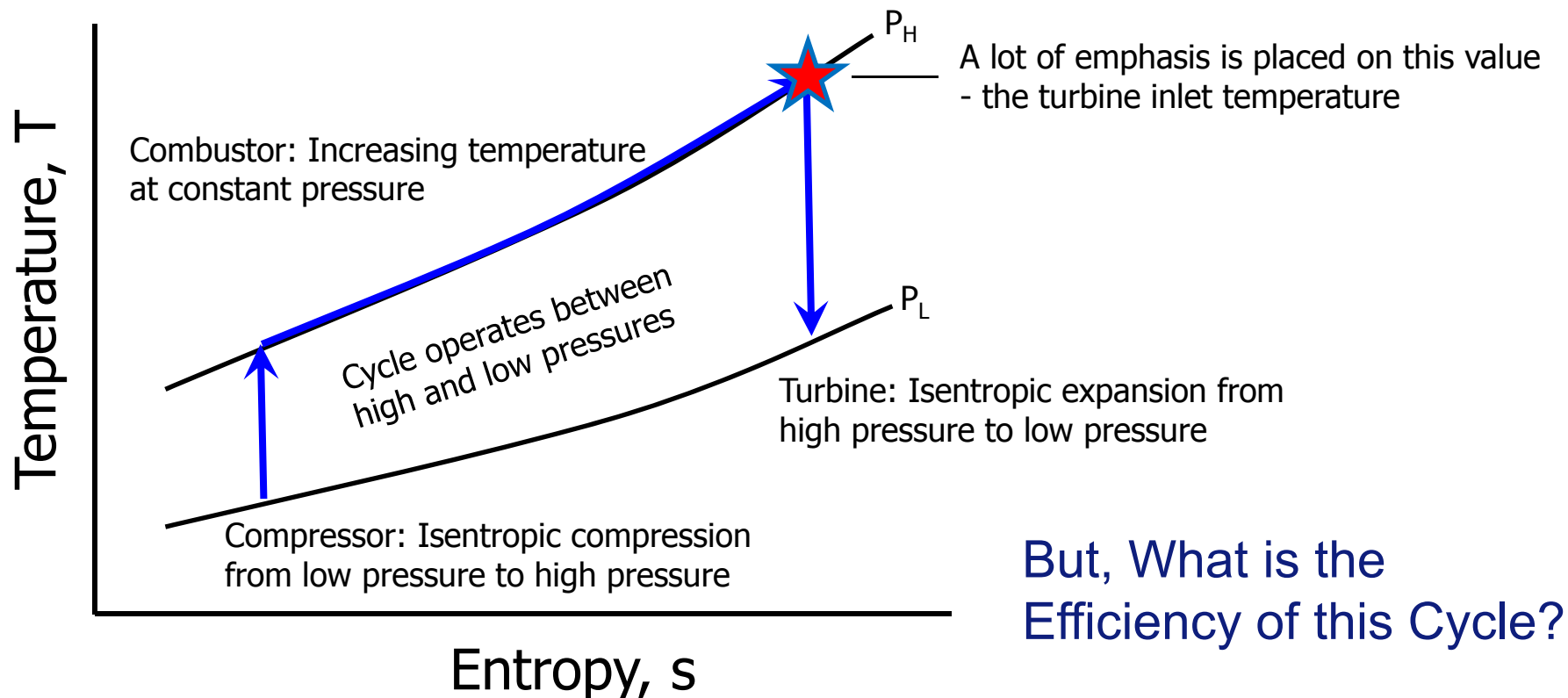
→ Focus here is on gas turbine performance

Gas Turbine Performance



What's the Limit? How Good Can it Get?

Understanding the fundamentals of the ideal thermodynamic cycle:

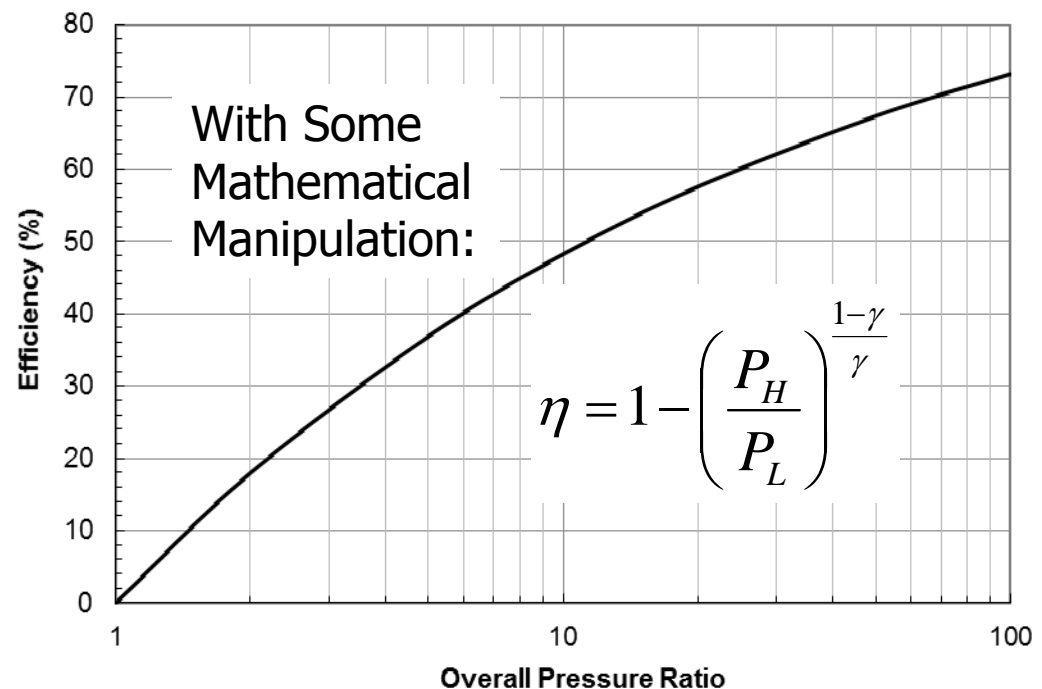
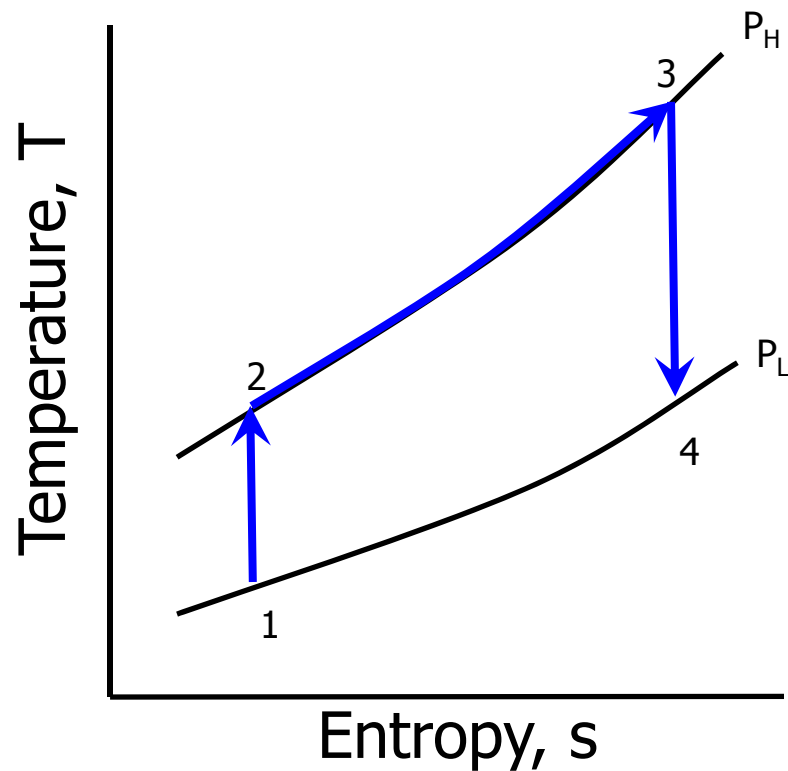


Gas Turbine Performance



Ideal Simple Cycle Efficiency – As Good as it Gets

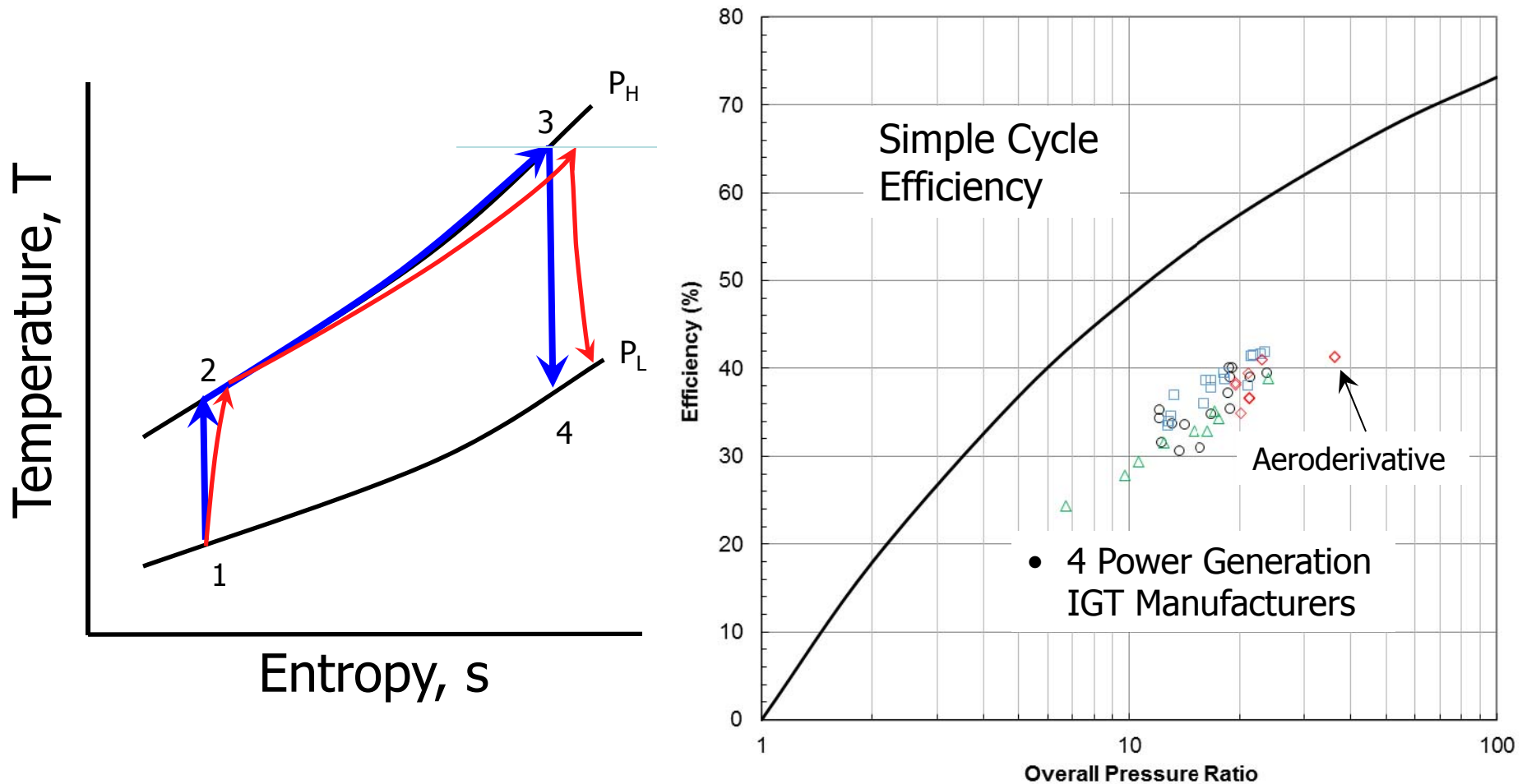
$$\text{Efficiency } (\eta) = \frac{\text{Useful Output}}{\text{Total Energy Input}}$$
$$\approx \frac{(T_3 - T_4) - (T_2 - T_1)}{(T_3 - T_2)}$$



Gas Turbine Performance



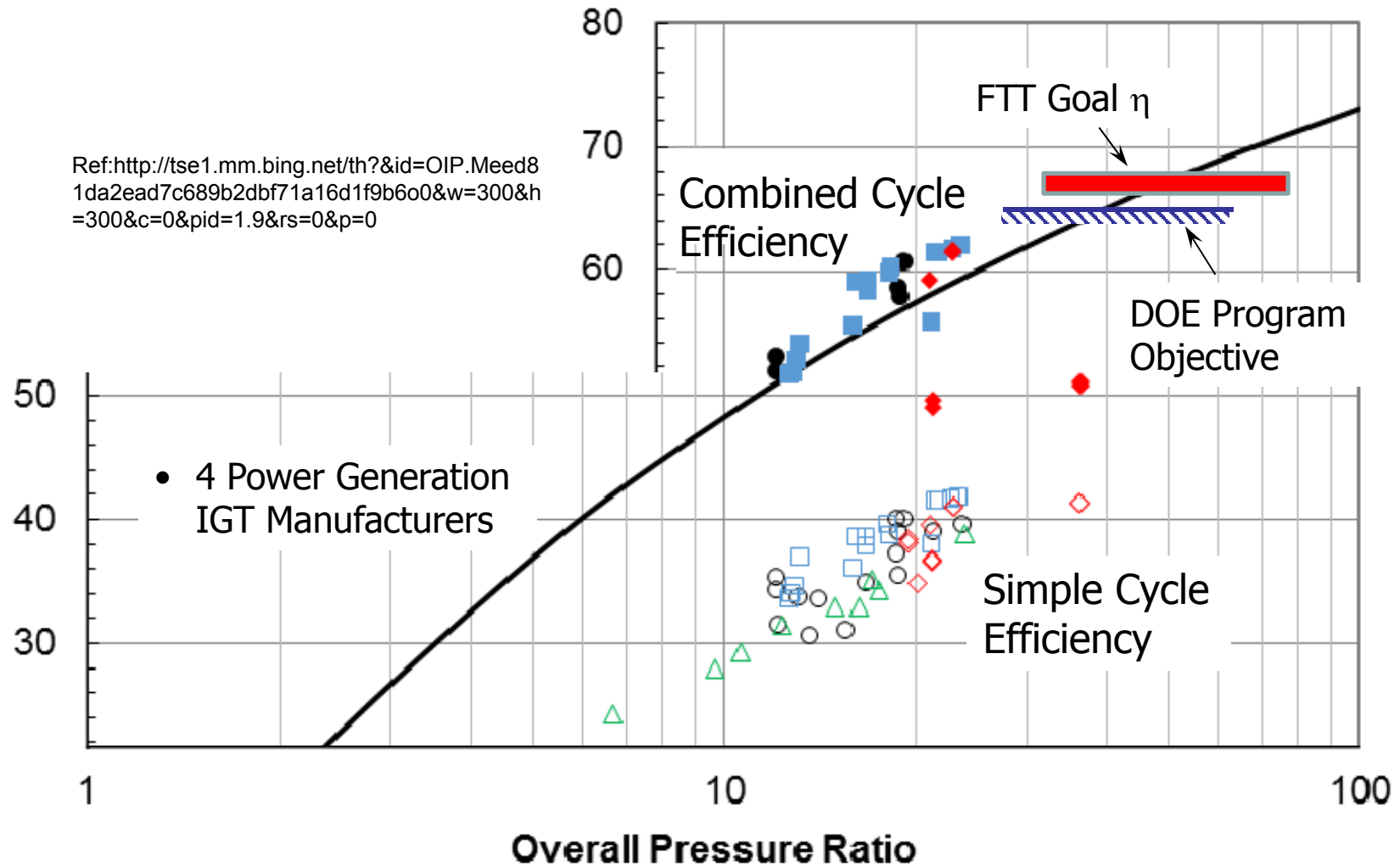
Reality \checkmark – Real Component Performance Reduces Efficiency



Gas Turbine Performance



Increased Pressure Ratio Needed to Reach Goals



FTT Design Solutions



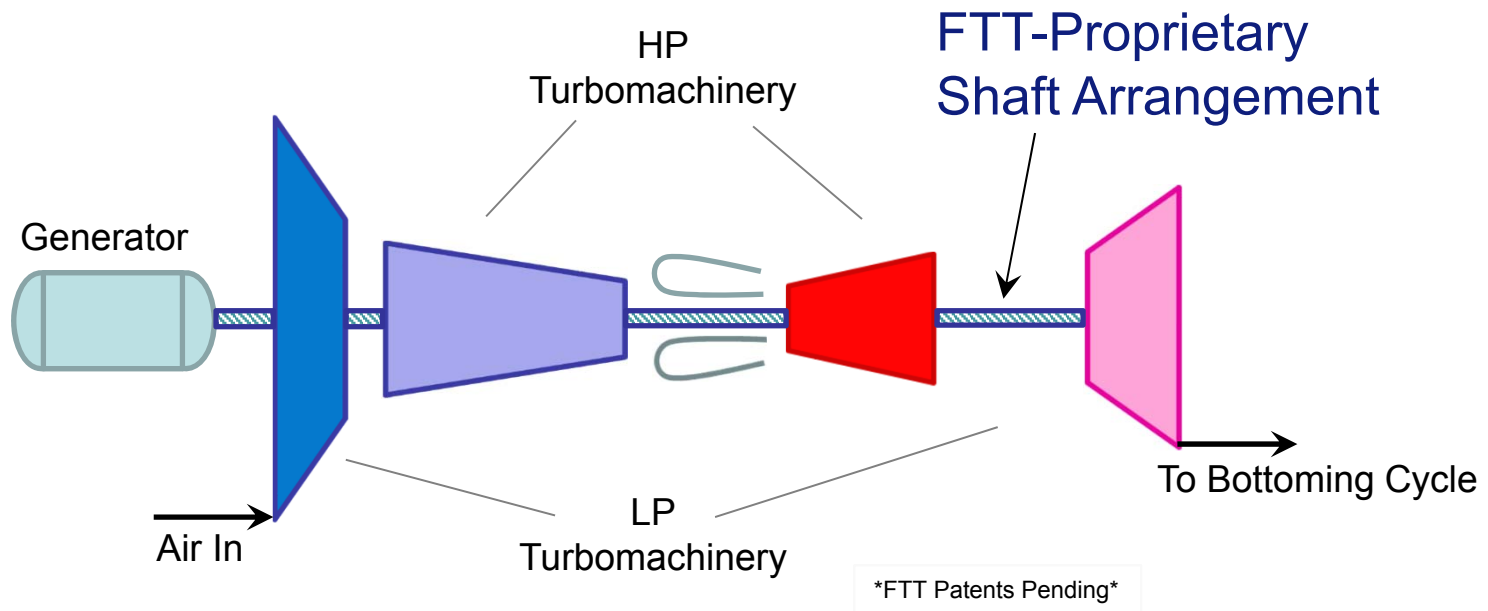
Problem – For Single Shaft Architecture, High Pressure Ratios Decrease the Size of the Core Hardware ... Reduces Component Efficiencies

<http://www.netl.doe.gov/scng/projects/end-use/at/images/at31176HseriesTurbine.jpg>

- Today's IGT operates on a single shaft
- Size is limited due to last stage turbine blade pull stress as quantified by AN^2
- Requires Innovation to Mitigate Effects of Shrinking Core Size
- FTT Offers New Solutions to Historical IGT Problems



Aircraft Engine-Derived Multi-Shaft Architecture – TurboGT™



➔ 2X pressure permits 2X flow through expensive core parts

➔ Extended operability

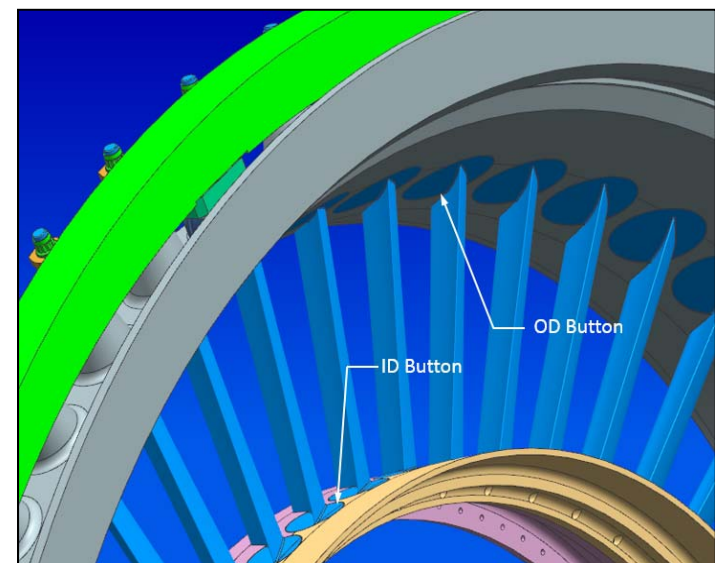
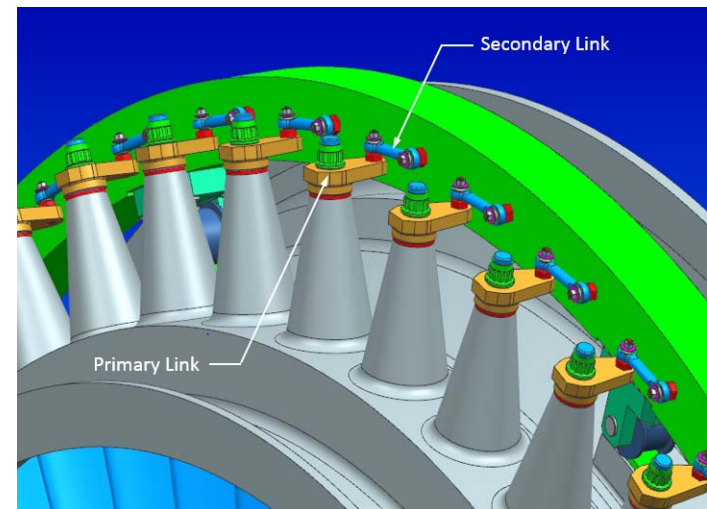
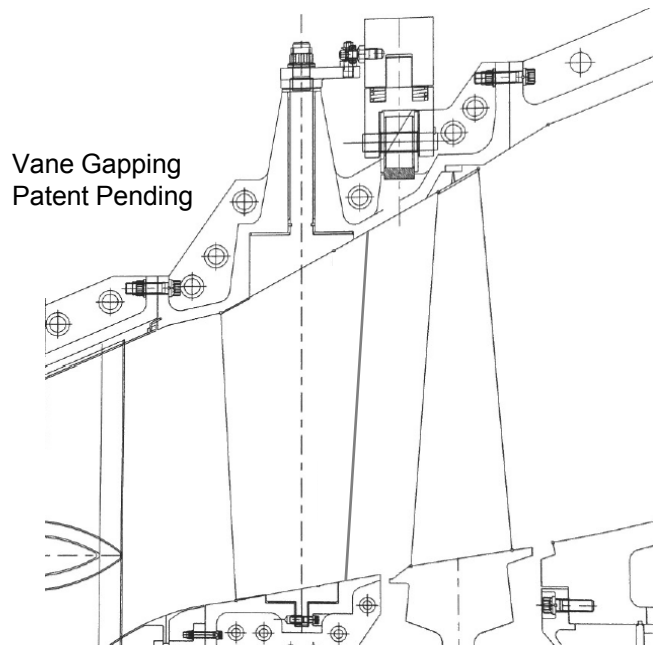
- Downturn to 10% power

Operability



Variable Turbine Geometry

- Improves responsiveness to intermittent power demands caused by increased use of renewables
- Enables substantial expansion of turn-down envelope

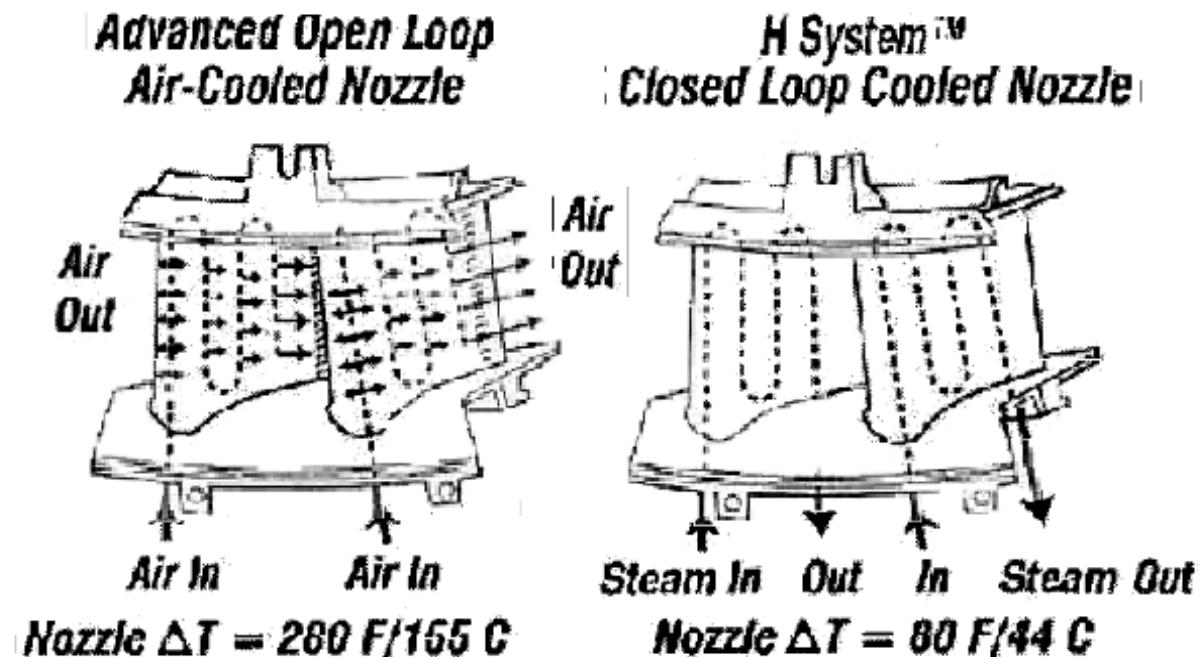


Turbine Cooling



Problem: C/A Dilution Increases Turbine Inlet Temperature

Independent Assessment/Conclusion:



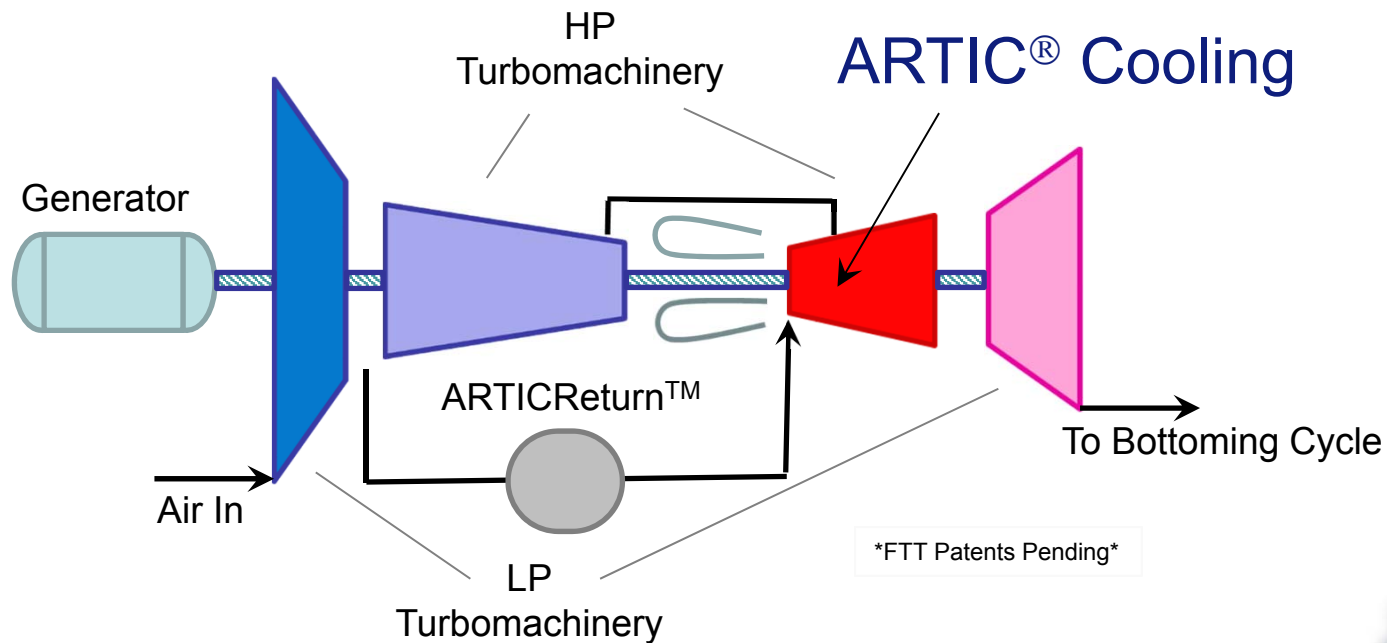
- Enables rotor inlet temperature to be increased 200°F/111°C
- Mitigates emissions impact

➔ Benefits of Internal Convection Cooling are Well-Established

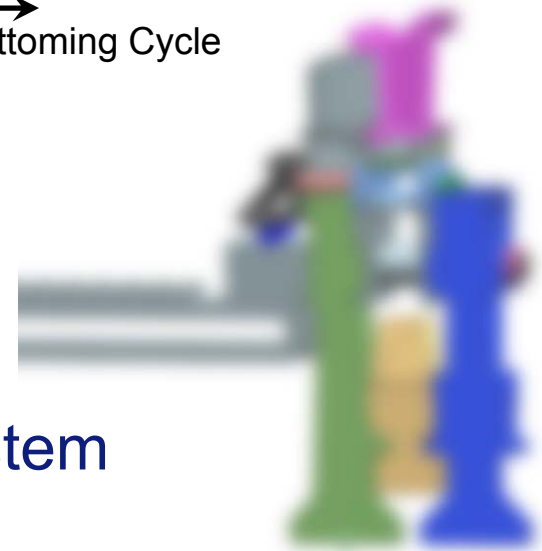
Turbine Cooling



Enhanced Cooling to Address Increased Turbine Inlet Temperature



Iteration #1
Geometry



→ ARTICReturn™ Cooling System

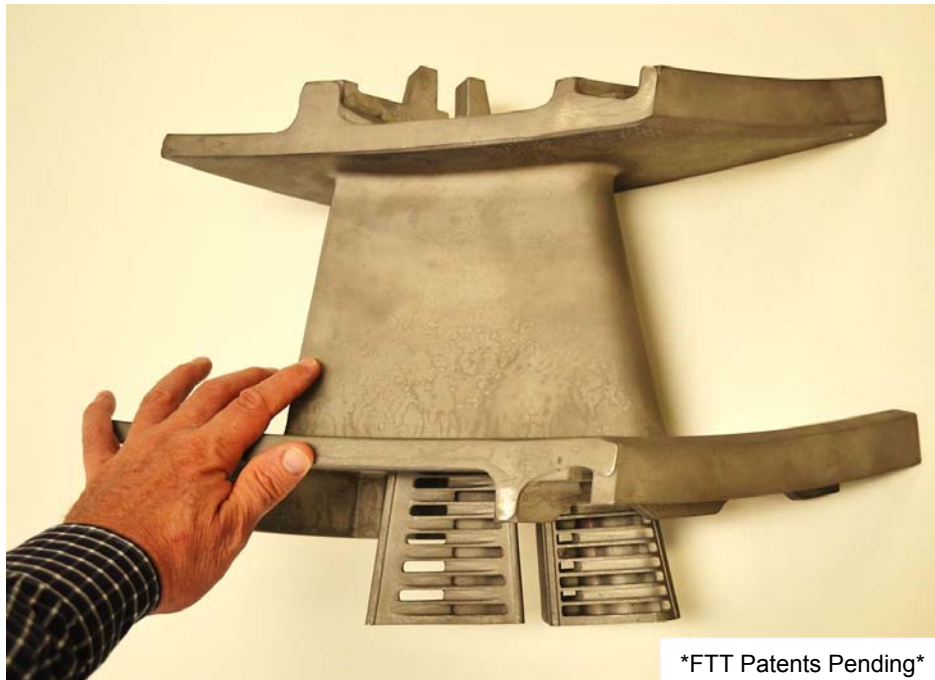
Turbine Cooling



Fundamentals of ARTIC[®] Vane 1 Cooling Technology Demo'd Under DOE Contract DE-FE0006696

Demonstrator Hardware Produced

- Rapid prototype technologies



FTT Patents Pending

Demonstrated in Full-Scale IGT

- Engine ran from March to April, 2014 with ARTIC[™] parts installed
- Total estimated run time ~ 115 hours
 - ~25 hours at maximum power temperature rating
- Total estimated number of cycles ~ 50
- All objectives of planned test campaign were completed



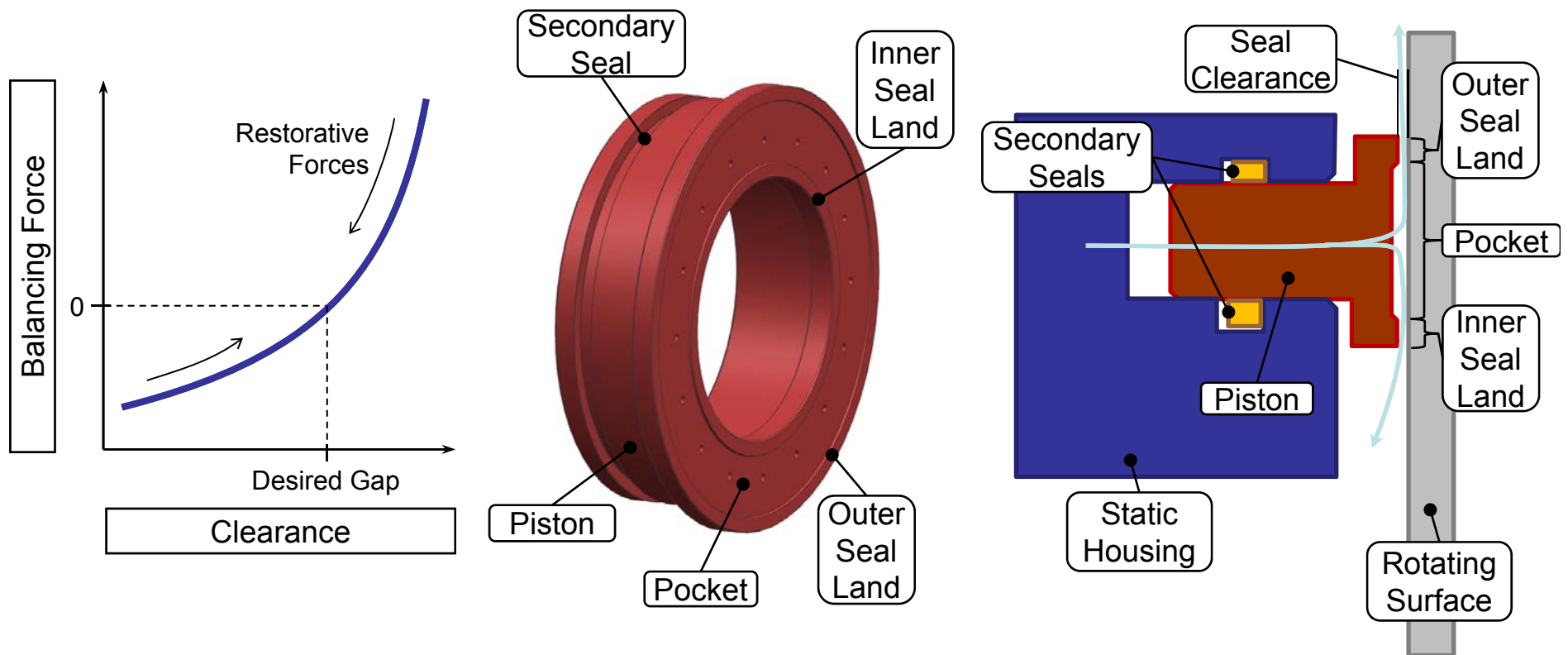
New Application in TurboGT[™] as ARTICReturn[™]

Advanced Leakage Control Technology



Air Riding Seal Concept (Jake Mills, Principal Investigator)

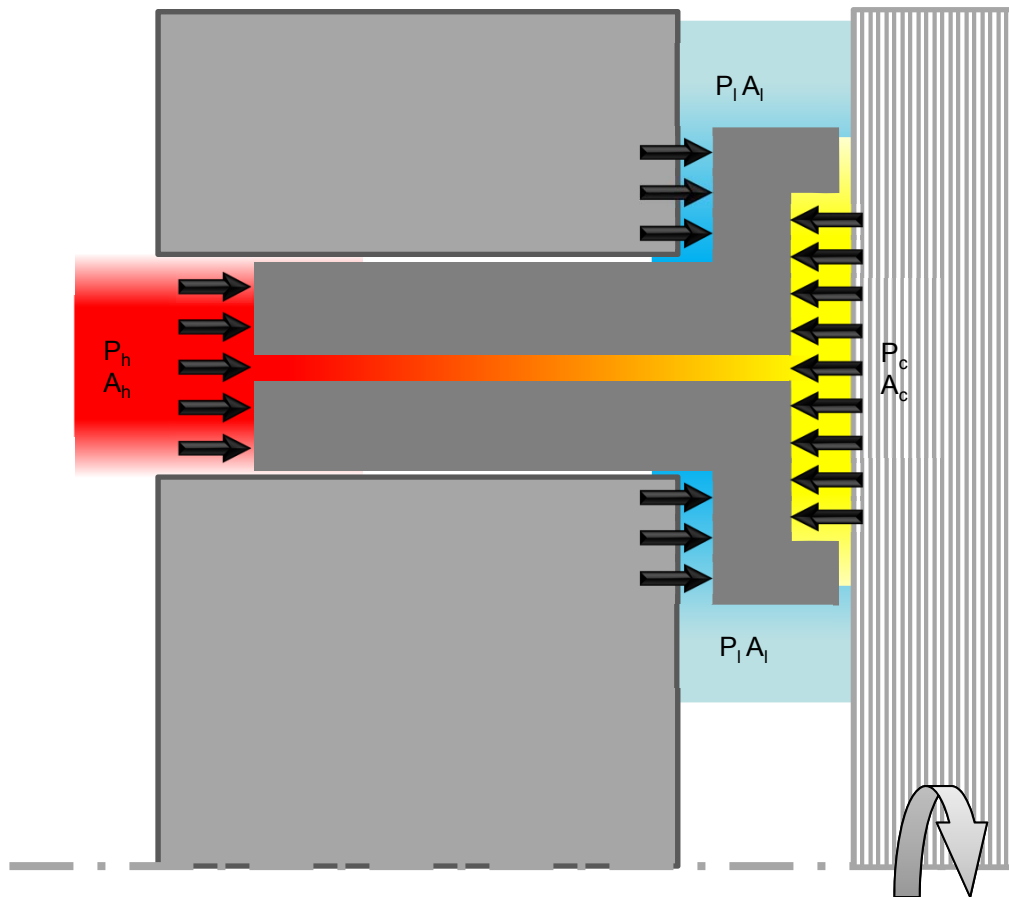
- Non-contacting static-to-rotating seal
- Hydrostatic balance of forces
- Ability to follow rotor to maintain close clearances



Advanced Leakage Control Technology



Air Riding Seal (ARS) Concept



Equilibrium $\Sigma F_x = 0$

$$P_h A_h + P_l A_l = P_c A_c$$

Increased Clearance $\Sigma F_x = \rightarrow$

$$P_h A_h + P_l A_l > P_c A_c$$

Reduced Clearance $\Sigma F_x = \leftarrow$

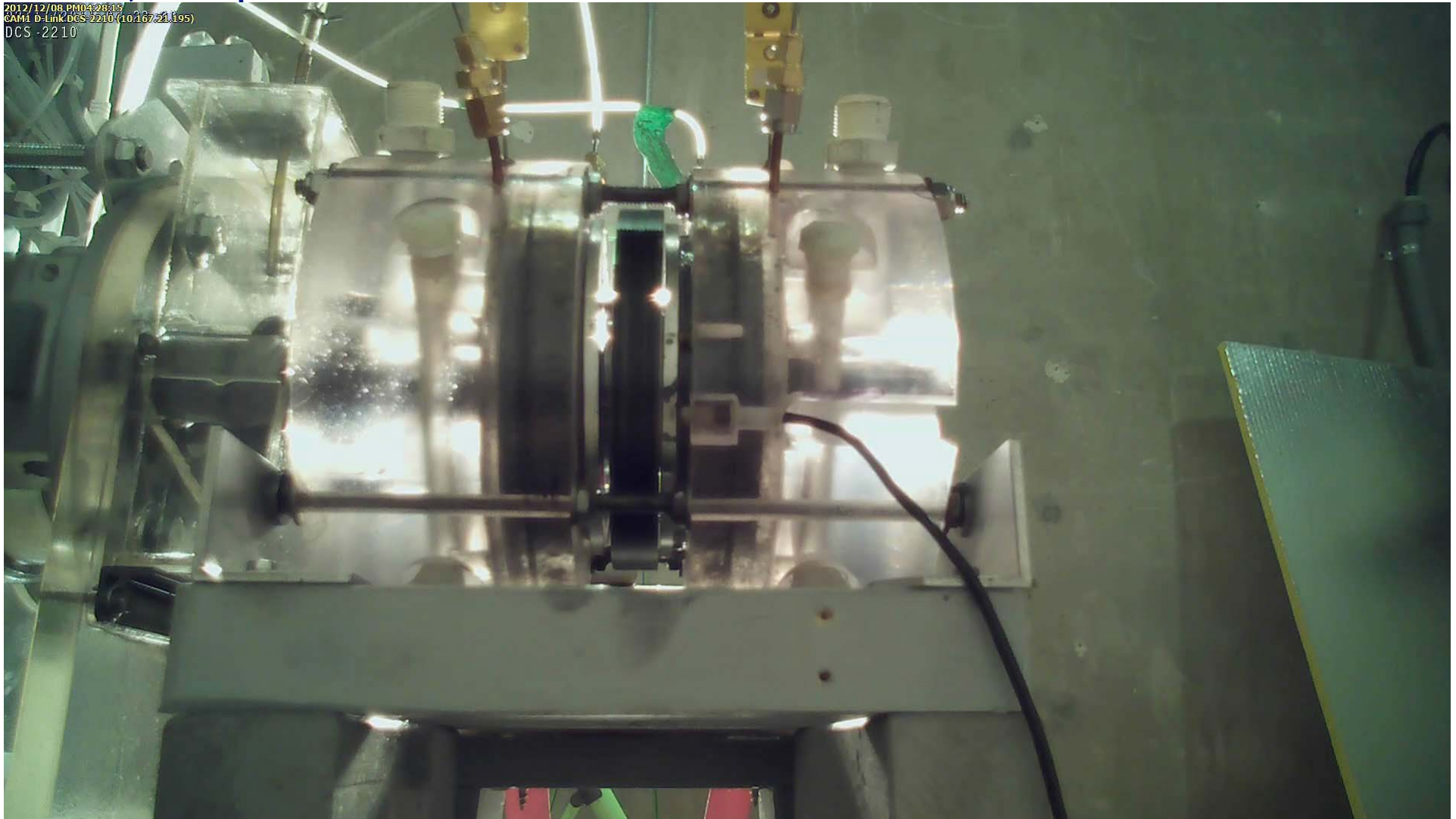
$$P_h A_h + P_l A_l < P_c A_c$$

ARS Technology Demo'd in Phase I



20,000rpm Seal Pressurization/Activation

2012/12/08 PM04:28:15
CAM1 D-Link DCS-2210 (10.167.21.195)
DCS-2210



November 3, 2015

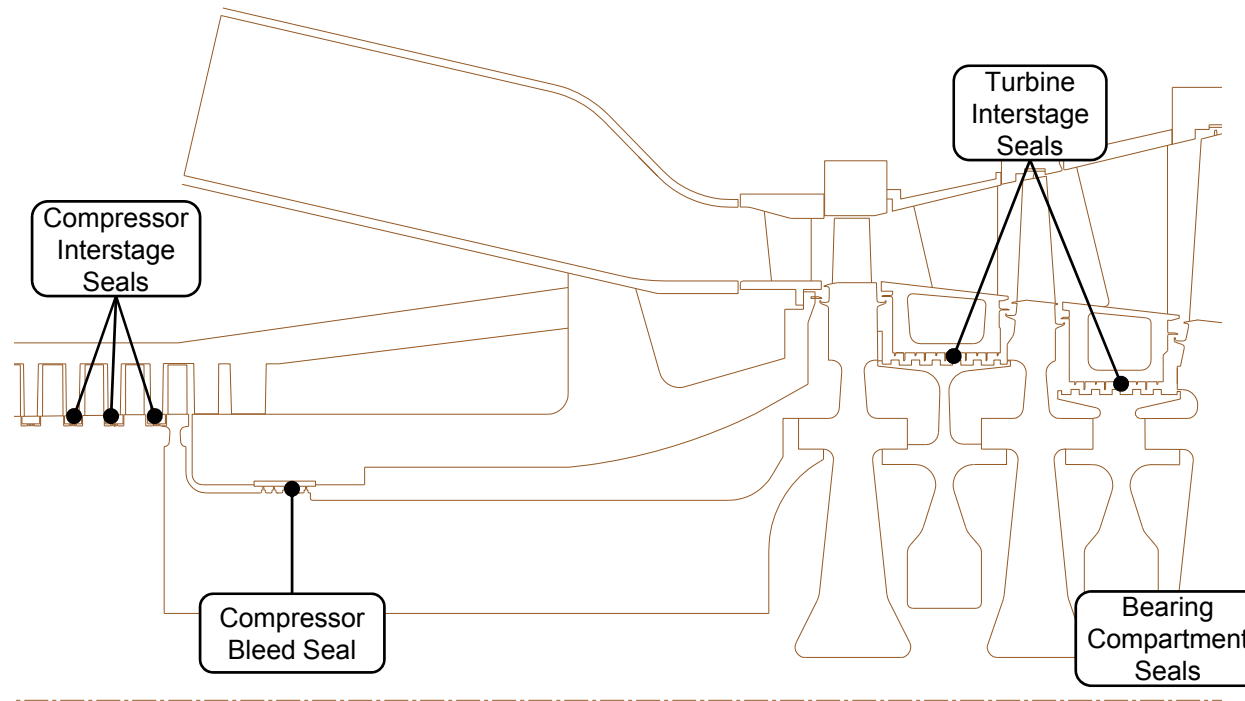
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ARS Application



ARS technology applicable to a variety of rotating to static seals



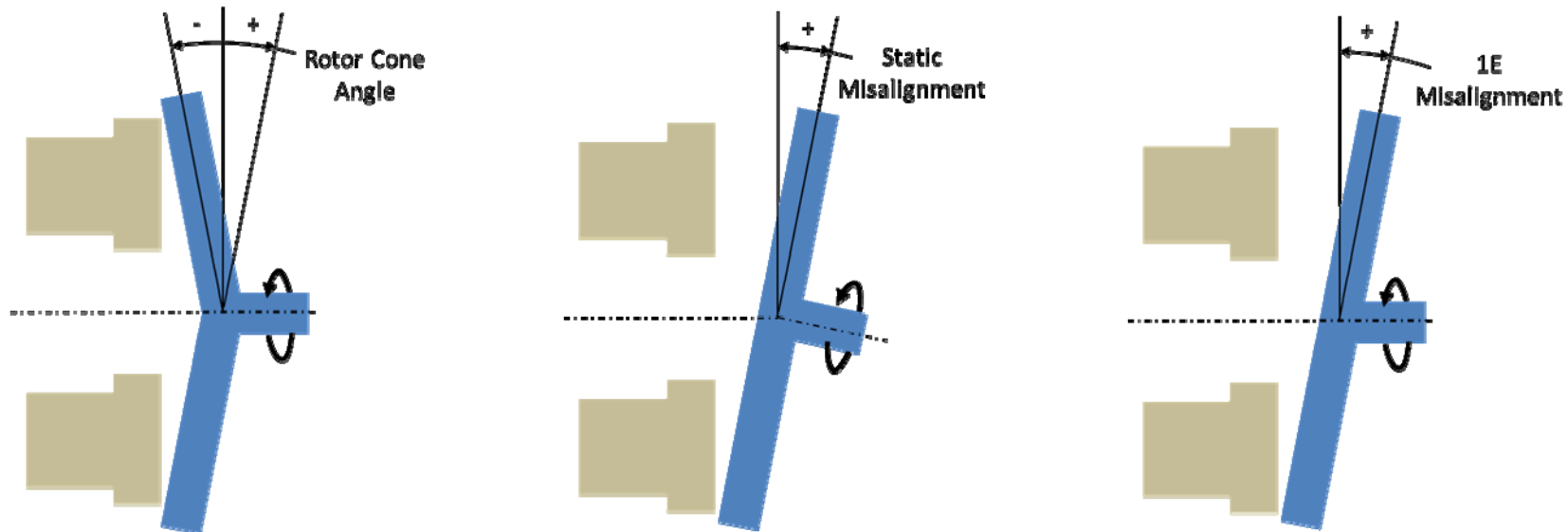
- Large utility scale engine performance model created to assess benefits of retrofitting an engine with the ARS technology

Rig Testing



Design-to-Test Operating Considerations

- Testing centered around engine operating conditions
- Rig allows for variation of a variety of design variables to understand the ARS design space and to calibrate/validate the analytical tools.



Future ARS Engine Testing



- To reduce cost and risk for future engine testing, the initial application of the ARS has been designed for the 501K
 - The ARS will be tested in a rig at engine conditions under the Phase II contract
 - Rig hardware has been designed to integrate into the engine
- ARS replaces the ‘thrust balance seal’ upstream of the first stage turbine

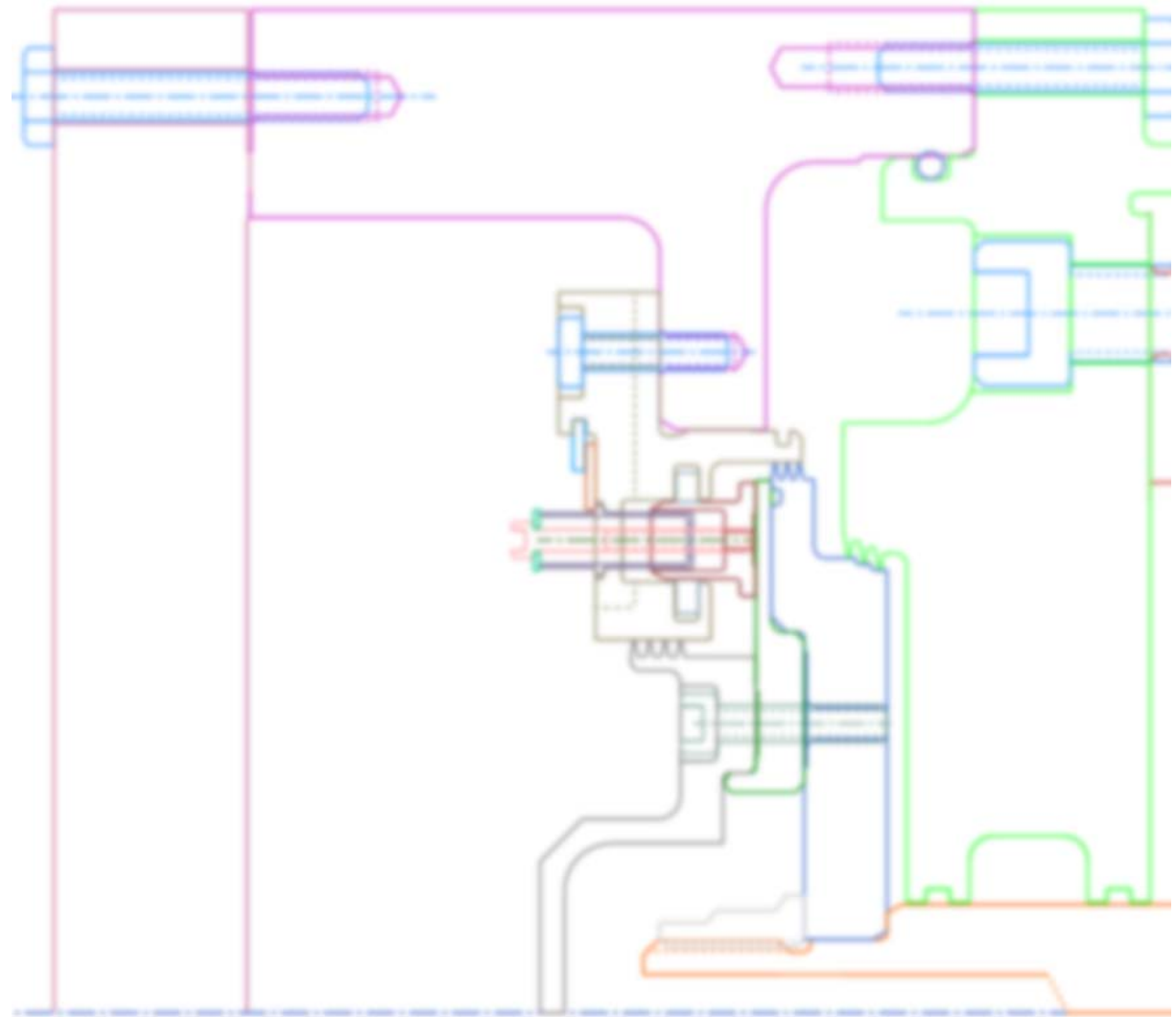
Project Overview



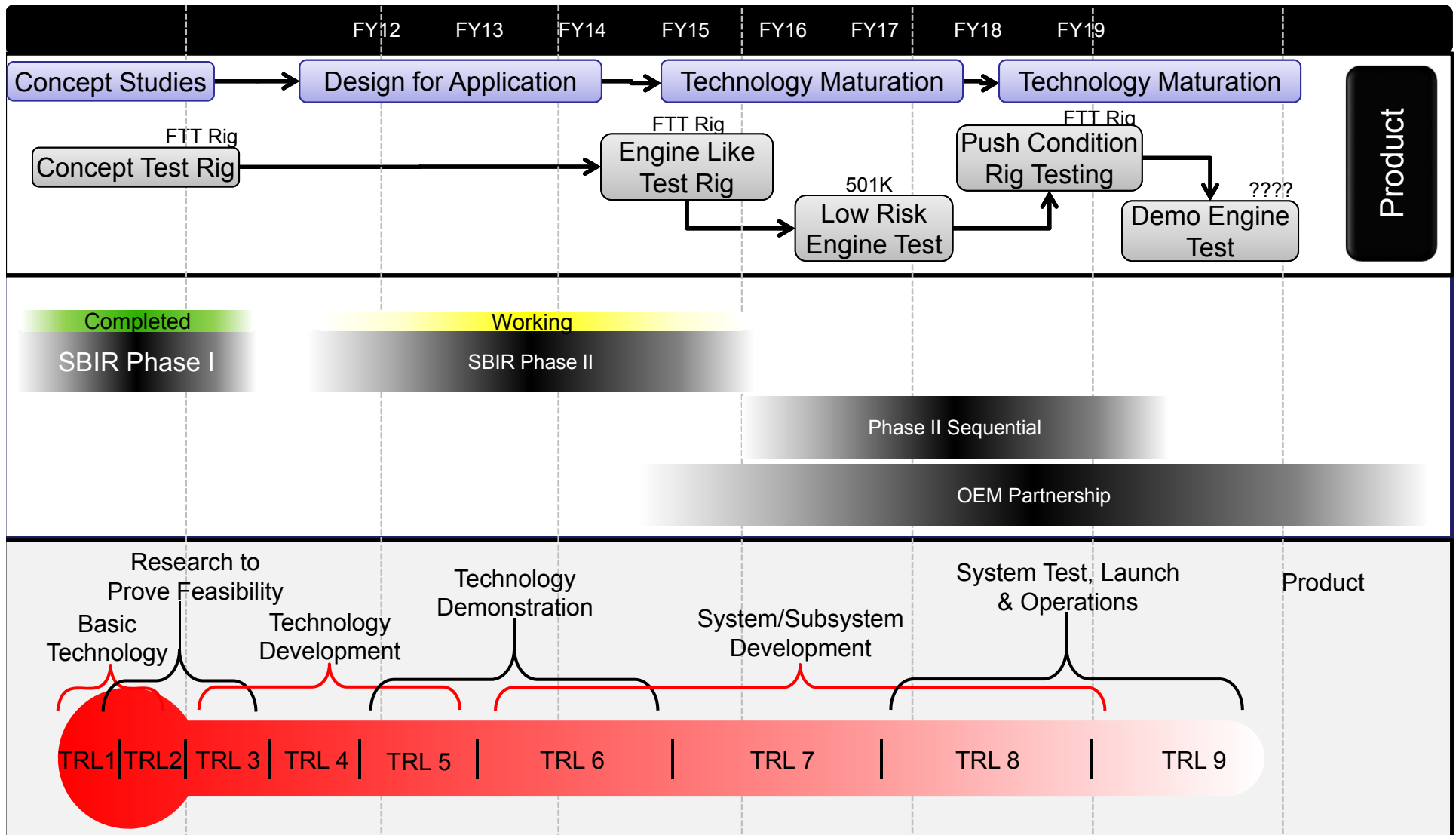
Rig

Rig

- Designed around engine hardware
- Design requirements and test objectives established from engine OEM
- Designed to establish a large design space centered around engine conditions



ARS Tech, Development Roadmap



TurboGT™ Conceptual Design Status



- Thermodynamic Performance Model for *TurboGT™* has been Created Using GateCycle Software
- High Fidelity Based On:
 - FTT's fast-iteration conceptual design system with:
 - Calibrated aerodynamic and turbine design codes
 - Compressor aerodynamics (Component efficiencies and diffuser pressure losses)
 - Turbine aerodynamics (Stage pressure ratios and efficiencies in HPT and LPT)
 - Fast FEM
 - Completed analysis: Rotor dynamics, Thrust balance, SAS
 - Incorporation of cooling and leakage flow rates and pressure losses based on latest mechanical arrangement
- Over past 12 months, design elements have been balanced to achieve a robust conceptual design which will be reviewed with the DOE in coming weeks



TurboGT™ Cost Impact



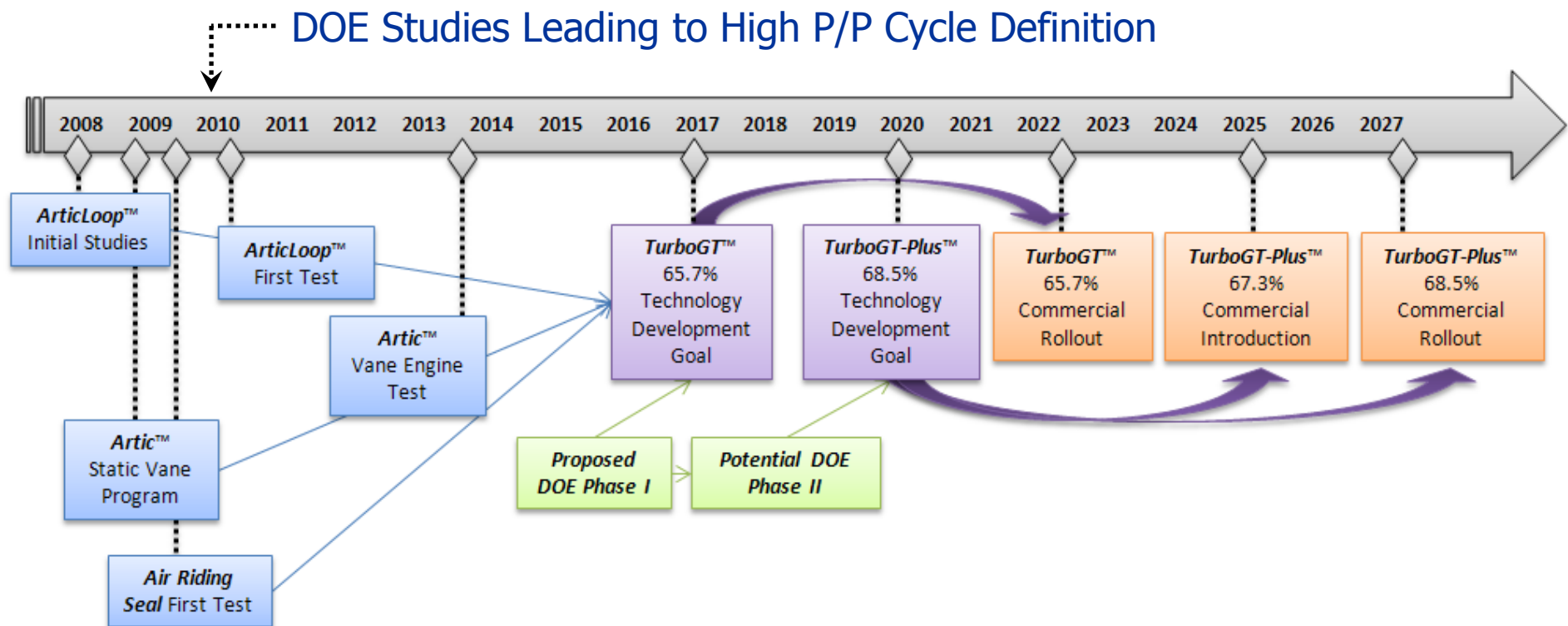
Multi-Pronged Approach to Reduce COE

- Low Equipment First Cost
 - More Steel, Less Nickel, per Megawatt
- Low Operating Cost
 - High Efficiency Maintained Over Wide Operating Range
 - Increased Utilization (Turn-Down to 10% Load)
- On Track For:
 - CC cost reduction > 25%
 - SC cost reduction > 30%

TurboGT™ Development Roadmap



Goal is for Technology Development of a 65% Combined Cycle to be Complete by 2017, with Commercial Rollout by 2022



Opportunities for UTSR Collaboration



- System-Level
 - Operability and controls validation
 - Controls logic development
- Component-Level
 - High pressure low NOx combustion using gaseous fuels
 - Emissions control over wide range of operating conditions
 - Coatings for high pressure environment
 - Cooling technology development
 - Seal durability test opportunities

Summary



- TurboGT™ Addresses All of the Advanced Turbines Program Goals
 - Affordability
 - Efficiency
 - Operability
 - Increased Temperatures
- Leverages Prior Successful FTT/DOE Component Development Experience
- Following Demonstration/Validation Test Phase, Commercial Rollout Could Be Realized by 2022
- Many Opportunities for Collaboration with UTSR Universities

Acknowledgements



Department of Energy
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Thank You & Questions?

