

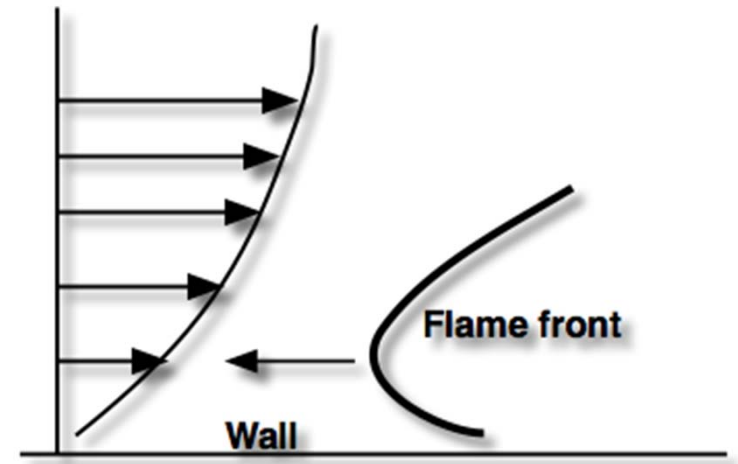
Modeling Flashback Propensity using LES and Experiments

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Boundary Layer Flashback

- Many different flashback modes possible
- Hydrogen-based combustion dominated by boundary layer flashback
- Flow near wall is slower than flame speed
 - ➔ Flame propagates upstream
 - ➔ Only wall quenching arrests flame
- Unique physics affects modeling
 - ➔ Turbulent boundary layer affecting flame physics



Understanding Flashback Fundamentals



- Previous project

- Flashback in swirling flow
- Looked at macrososcopic effects and flow physics
- LES modeling based on existing technology

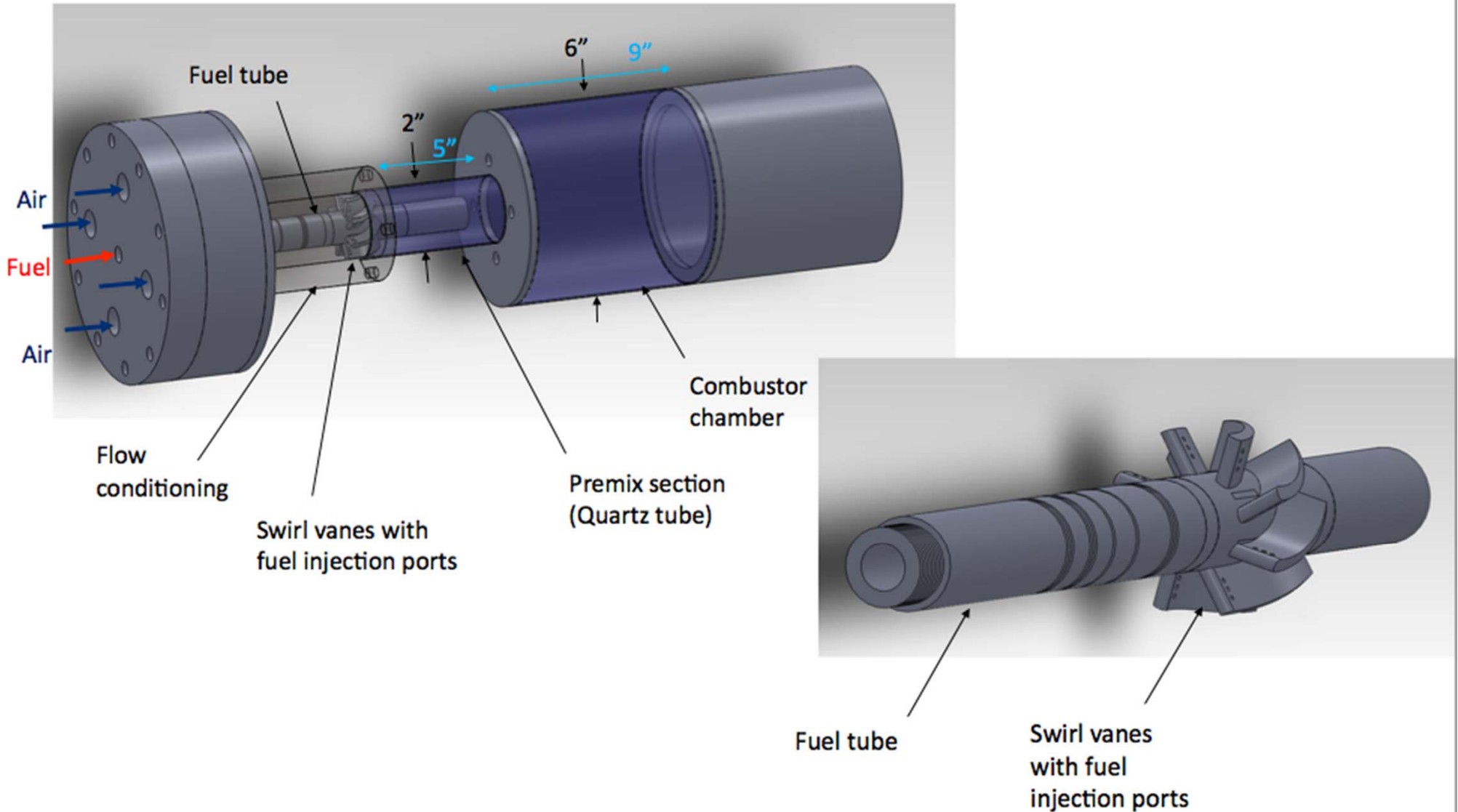
- Current project

- Oct. 2013-2016
- High pressure effects on flame propagation
- Fundamental aspects of LES modeling
 - Flame-wall interactions
- Predicting probabilities instead of average flashback

Experimental Program

UT Swirl Burner

- UT high-pressure swirl combustor



Flashback and Mitigation Strategies



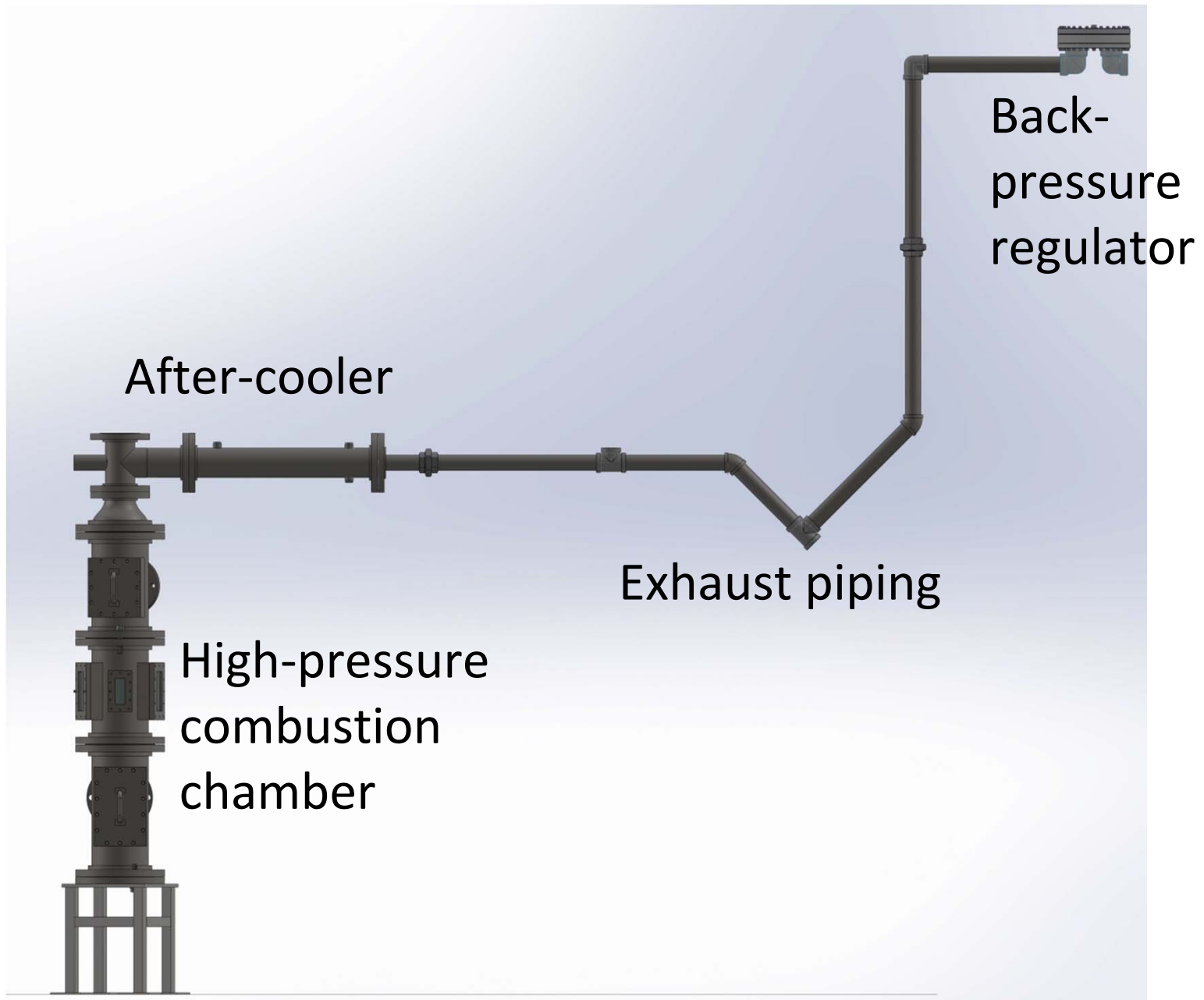
- Flashback at higher pressure
 - ➔ Effect of Reynolds number
- Stratification for flashback reduction
 - ➔ Fuel profiling
 - Different flow rates through different nozzle inlets
 - Less fuel near walls
 - Push inner boundary layer outside flammability limits
 - ➔ Prevent flame anchoring
 - Even with flashback, prevent flame from reaching inlet vanes

Experimental Program

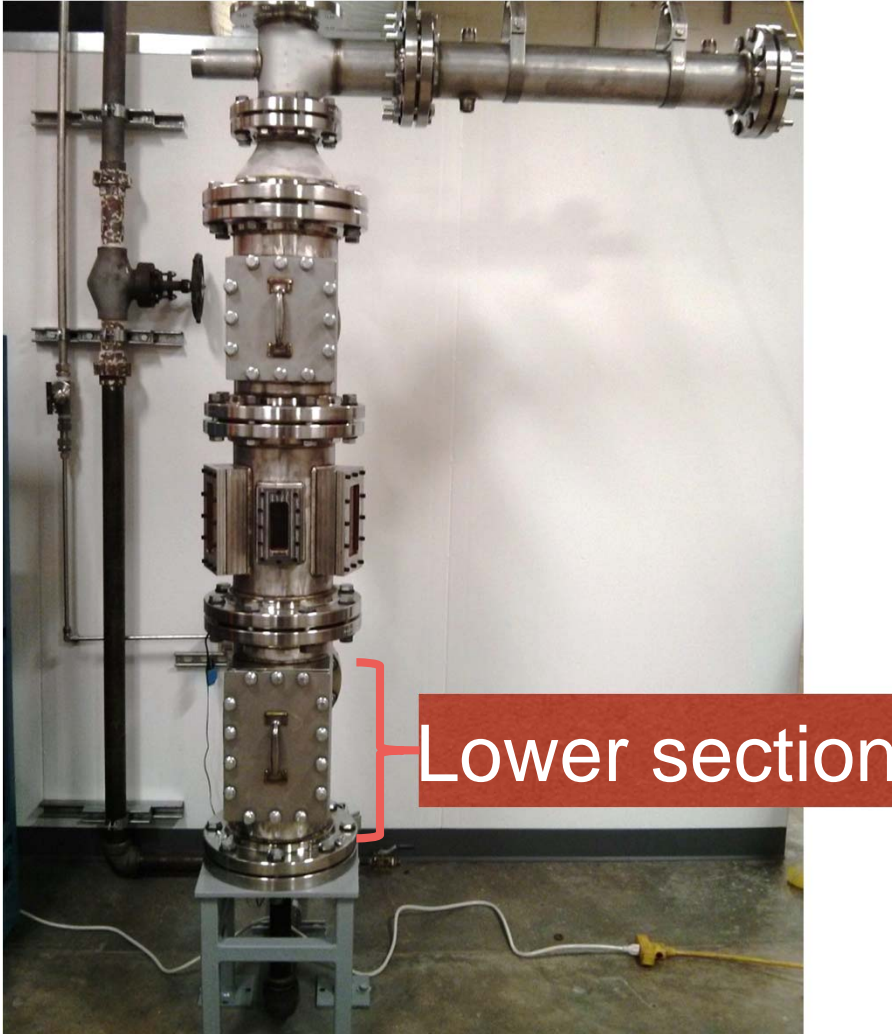
- Two main accomplishments
 - ➔ Complete the High-Pressure Combustion Facility
 - ➔ Develop Radially-Stratified Burner for use at 1 atm and in high-pressure combustor
- High Pressure Combustion Facility
 - ➔ Modular Structure
 - ➔ Stainless Steel
 - ➔ Designed for pressures up to 15 atm
 - ➔ Allows mounting of various combustors
 - Flashback
 - Stratified flames



High-Pressure Combustion Facility



High-Pressure Combustion Facility



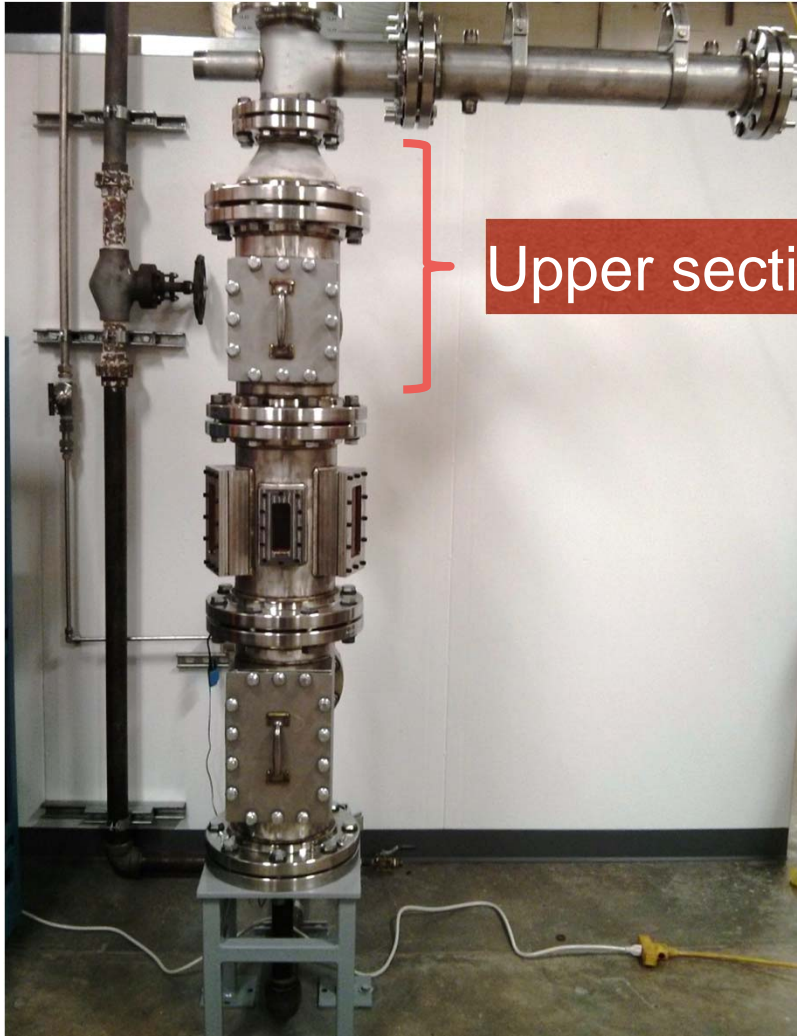
- Lower section
 - ➔ Access port for installation
 - ➔ Gas supply ports to the internal burner assembly

High-Pressure Combustion Facility



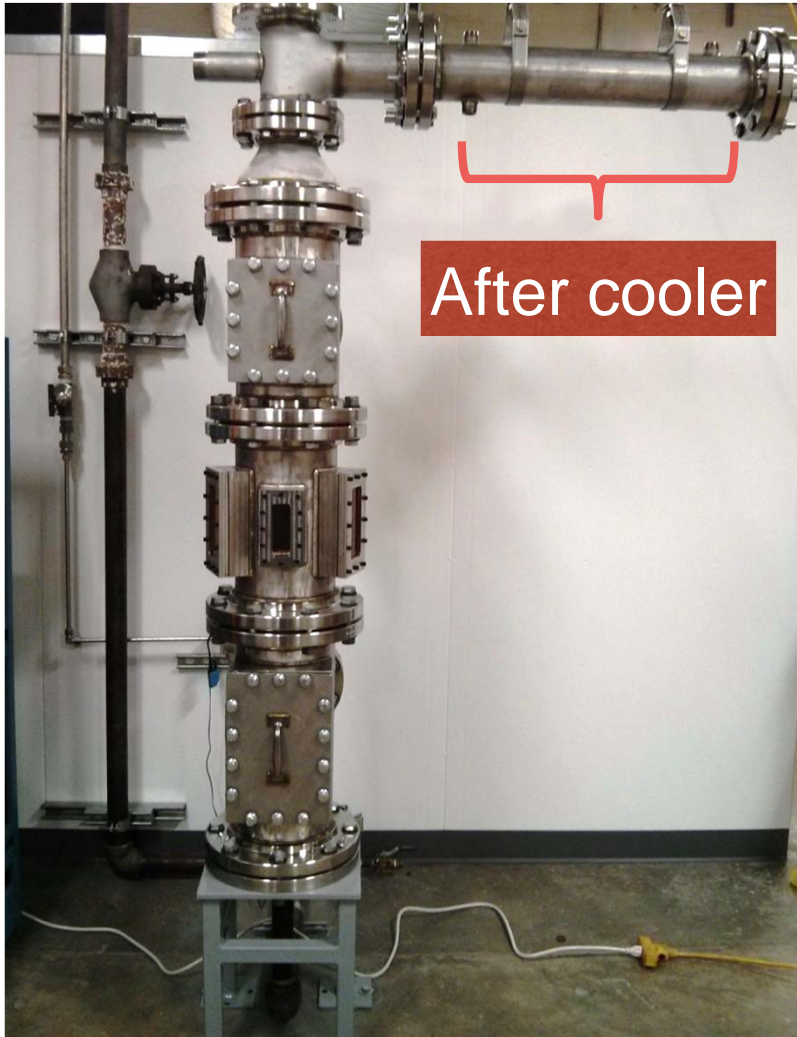
- Combustion Chamber
 - ➔ Contains three windows for laser diagnostics
 - ➔ High-speed stereo PIV
 - ➔ Chemi-luminescence
 - ➔ PLIF
- Uses shroud air-flow for cooling windows

High-Pressure Combustion Facility



- Upper Section
 - ➔ Access ports for installation and calibration

High-Pressure Combustion Facility



- After cooler
 - ➔ Shell and tube heat exchanger made using copper coils

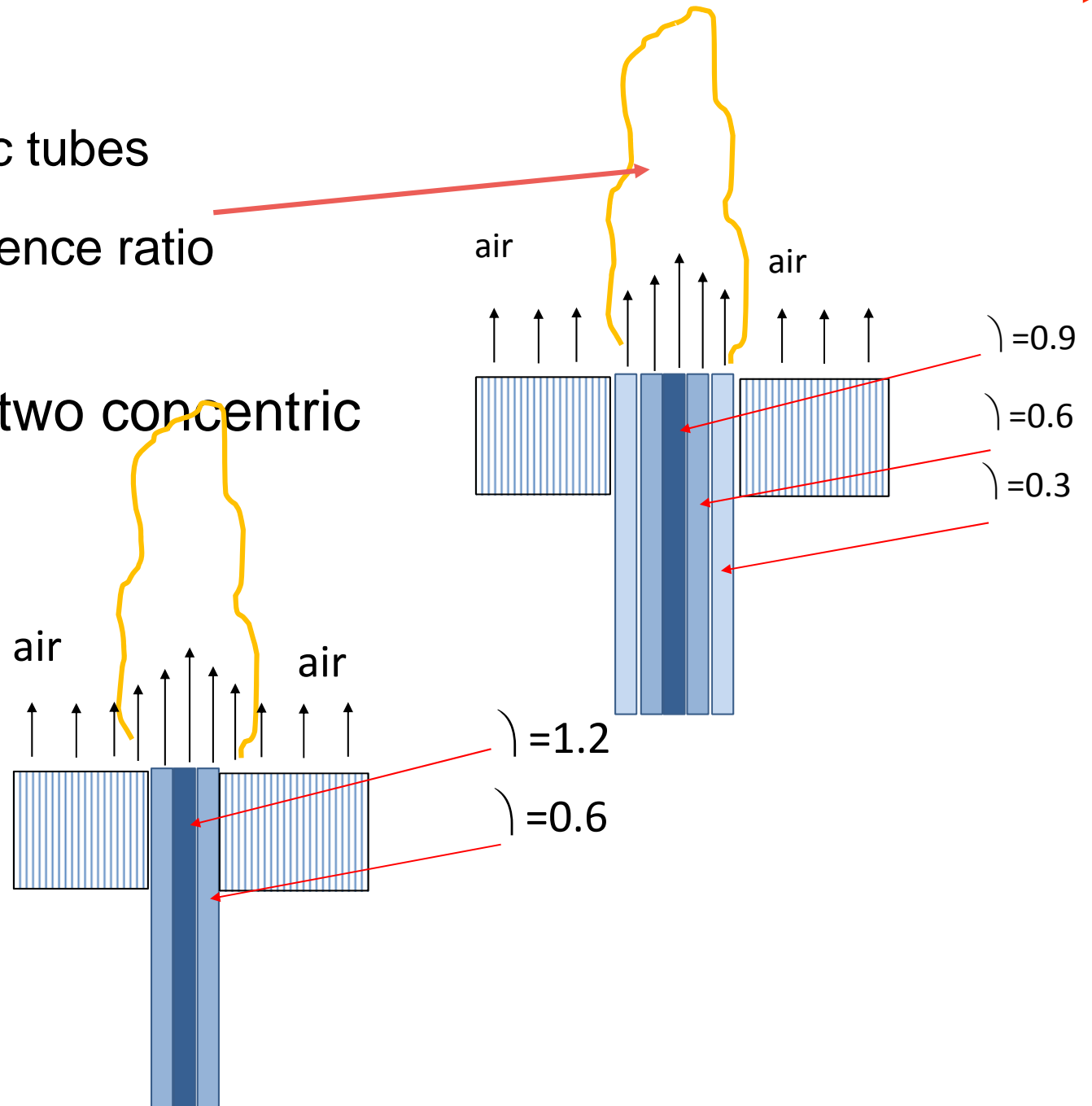
Radially-stratified Flame Burner

- Burner design

- Multiple concentric tubes

- Different equivalence ratio mixtures

- Initial design with two concentric tubes

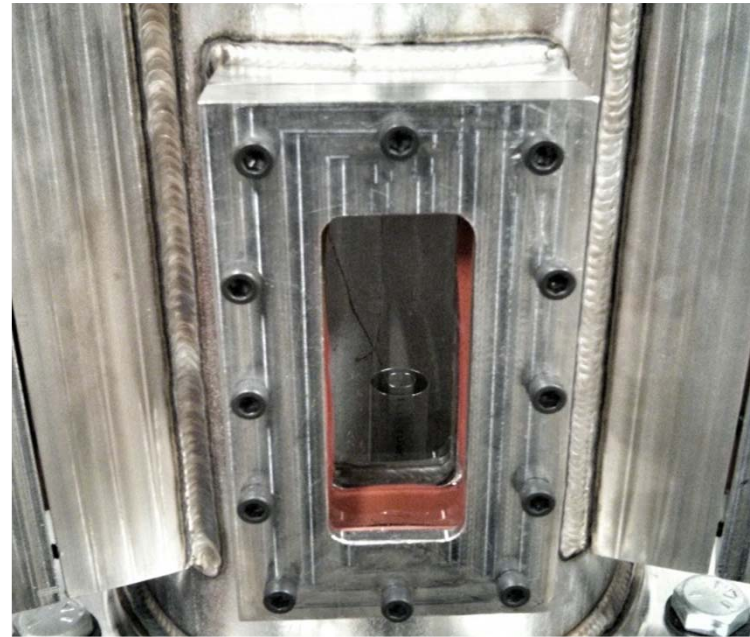


Radially-stratified flame burner

- Two concentric nozzles of dia. 0.5" and 1"
- Long nozzles ensure fully developed flow
- Concentric tubes will be surrounded by a co-flow section (under construction)



Stratified burner



Stratified burner mounted in chamber

Stratified Burner

- Stratified burner currently undergoing testing for flame stability with CH₄-air
- Rich mixtures in both nozzles is stable at high Reynolds numbers
- Lean outer flow and rich inner flow is lifted flame for Reynolds number > 3000
- Hydrogen addition should give wider stability limits



Methane-air
stratified

Methane-air stratified flames at 1

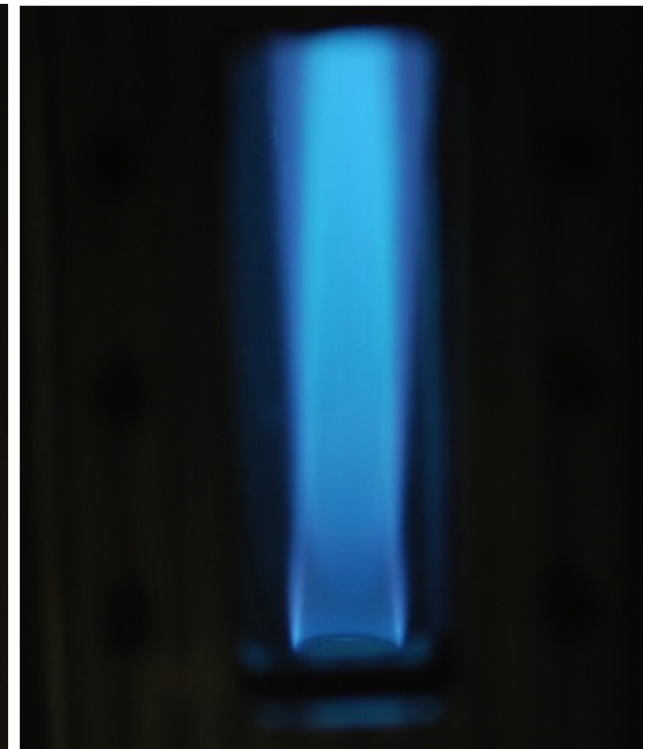
atm



Inner nozzle only
 $\text{Ø} = 2.72$, $\text{Re} = 4776$



Outer nozzle only
 $\text{Ø} = 2.12$, $\text{Re} = 3915$



Both nozzles
Inner $\text{Ø} = 4.08$
Outer $\text{Ø} = 1.9$

Planned work

- Use $\text{H}_2/\text{CH}_4/\text{N}_2/\text{air}$ pre-mixtures to widen stability limits
- Make extensive measurements at 1 atm
 - PIV
 - Temperature imaging (Rayleigh scattering using DLR fuel – $\text{H}_2/\text{CH}_4/\text{N}_2$)
 - OH/CH PLIF
- Make measurements at elevated-pressure conditions

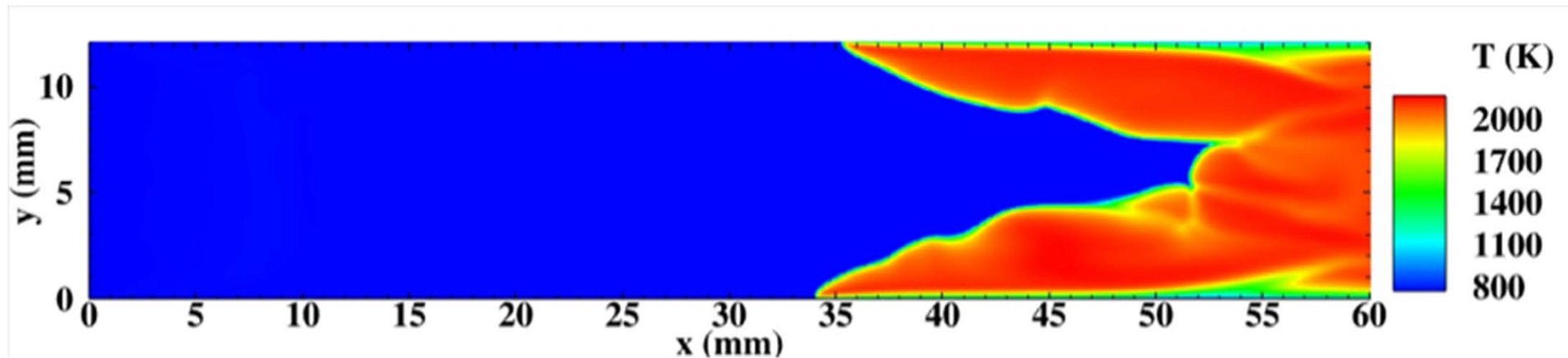
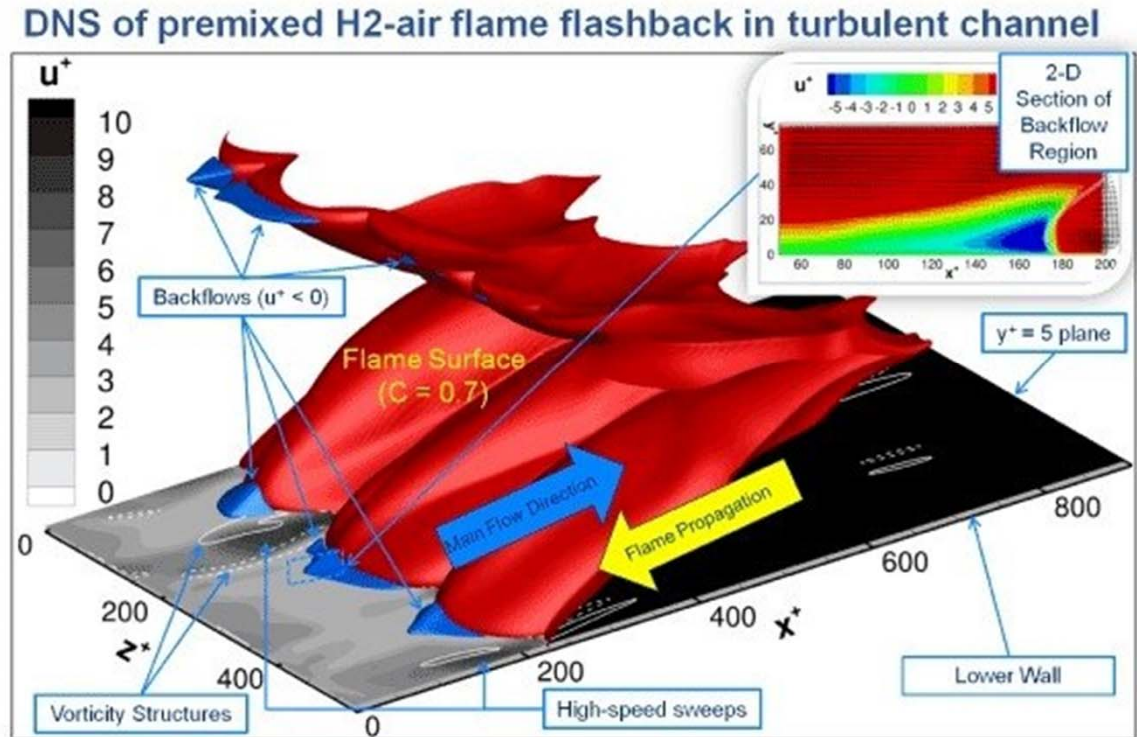


Methane-air
stratified

LES Modeling of Flashback

LES Modeling of Flame Flashback

- Experimental data is not refined enough to test model hypotheses
 - Use of DNS data
 - Sandia National Lab.
 - Chen and co-workers



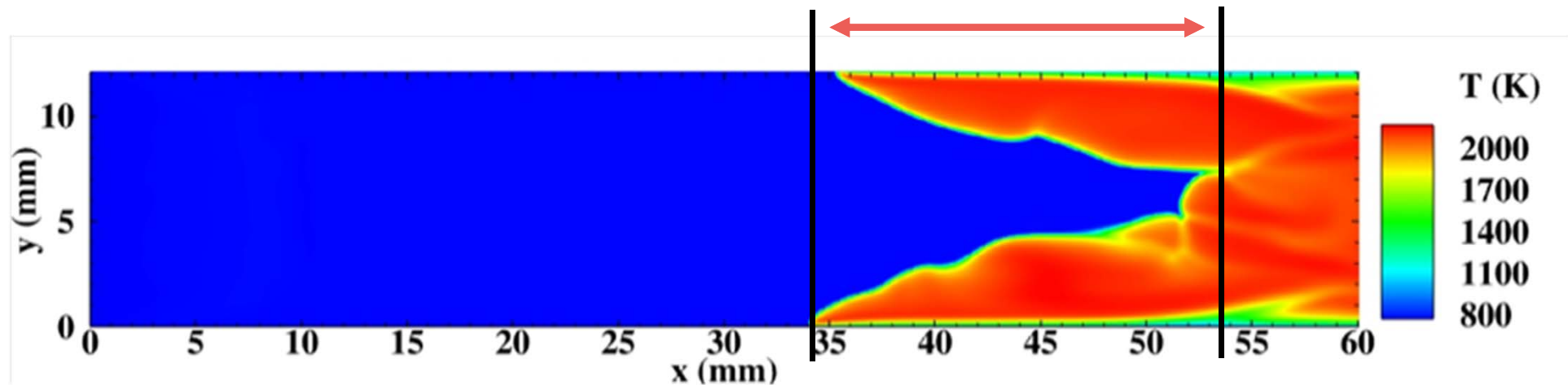
Modeling Approach



- Flame-front described using progress variable
 - ➔ Flame structure through flamelet model
 - This is strictly not necessary
 - ➔ Progress variable source term determined to predict the correct laminar flame speed
- Modeling issues
 - ➔ Near-wall heat loss effects
 - ➔ Small-scale flame wrinkling
 - ➔ Numerical solution of the progress variable equation

DNS Statistics

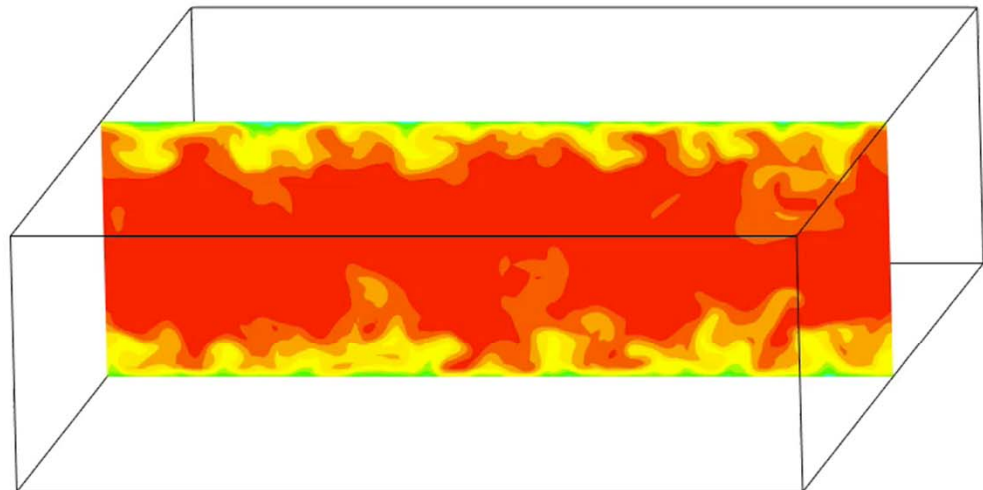
- DNS represents a single realization of flashback
 - No statistical information
- Derived statistical quantities
 - Flame depth
 - Spanwise averaged flame propagation velocity
 - Computed at leading edge



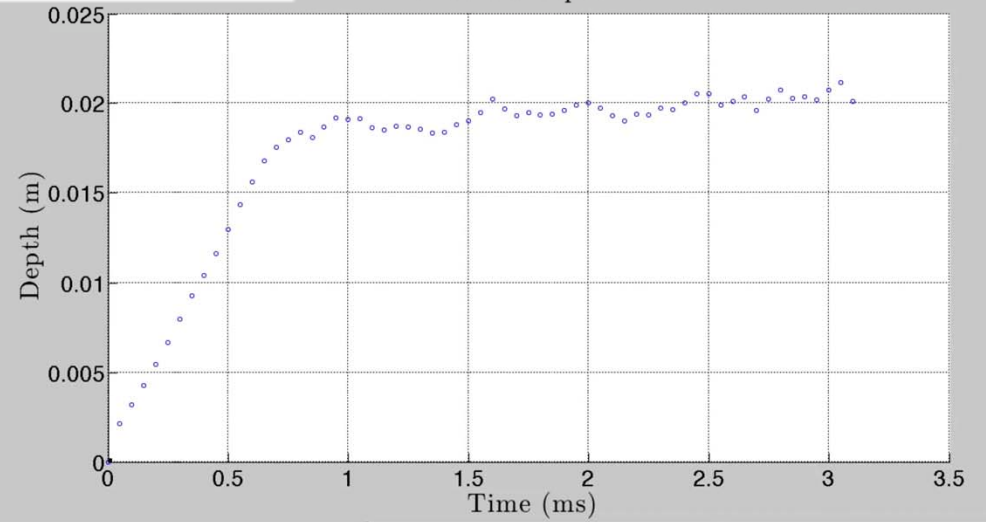
LES Statistics



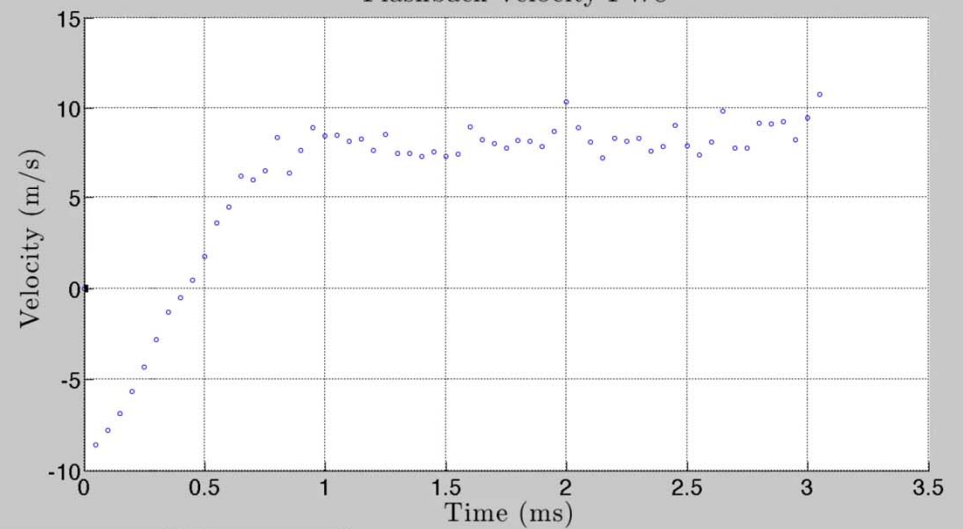
U (m/s)



Flashback Depth FW8



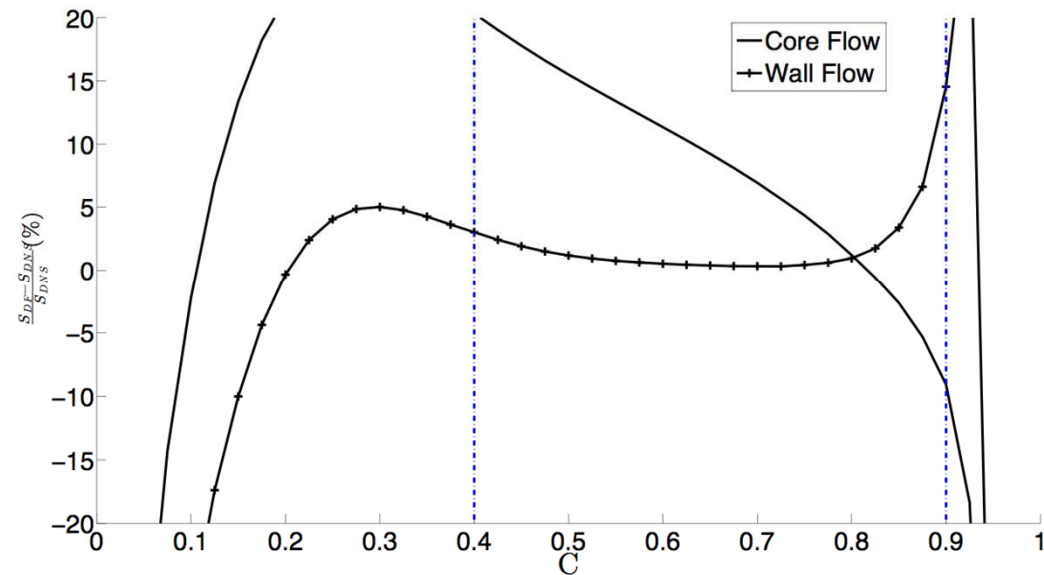
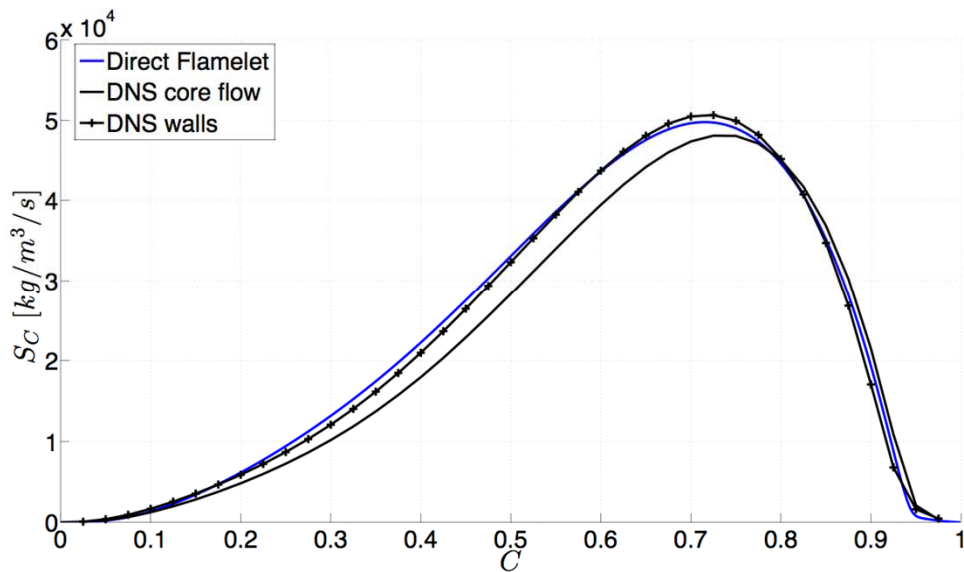
Flashback Velocity FW8



Flamelet Model Errors

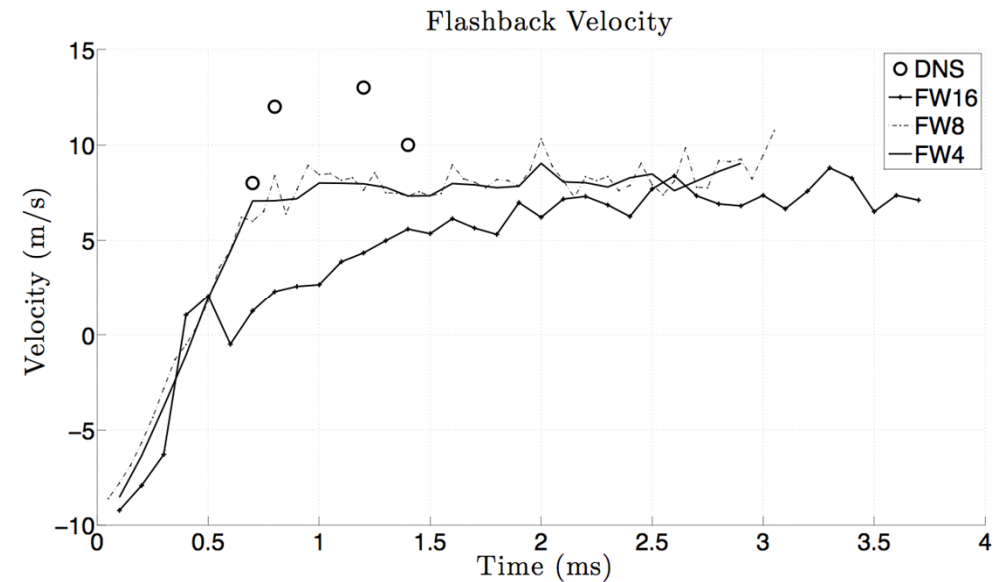
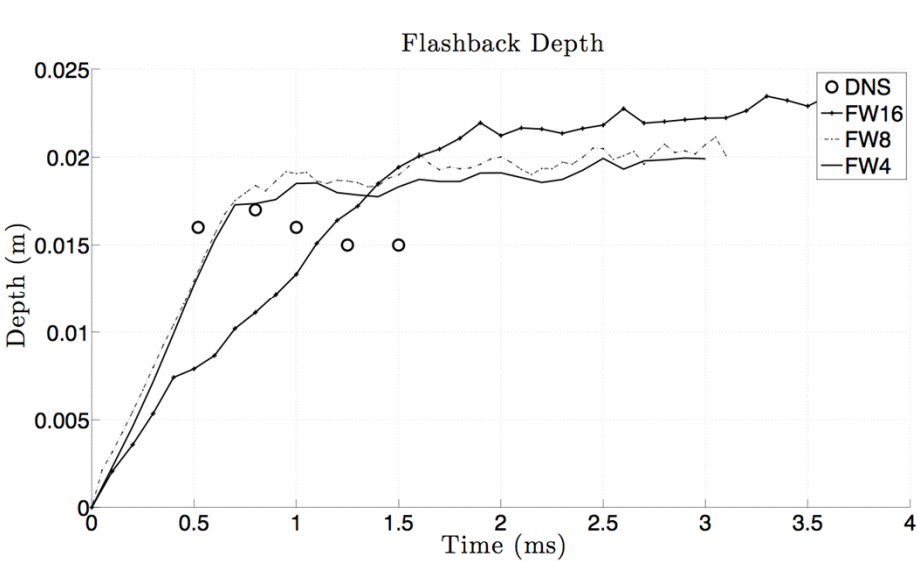
- Flamelet assumption used to obtain progress variable source term

➡ Evaluated also from DNS data



LES Results - Filter Width Effect

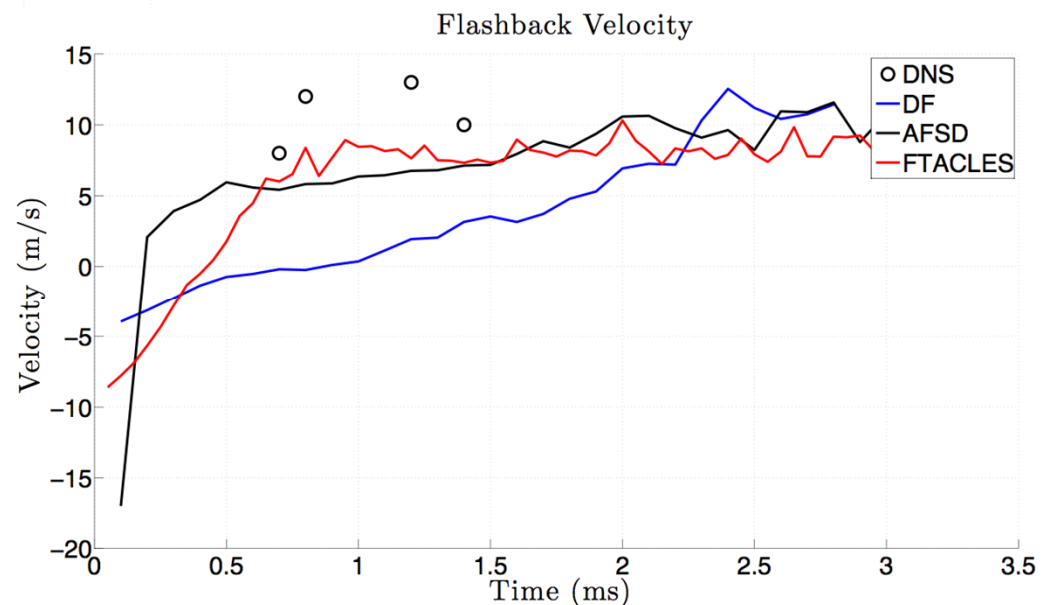
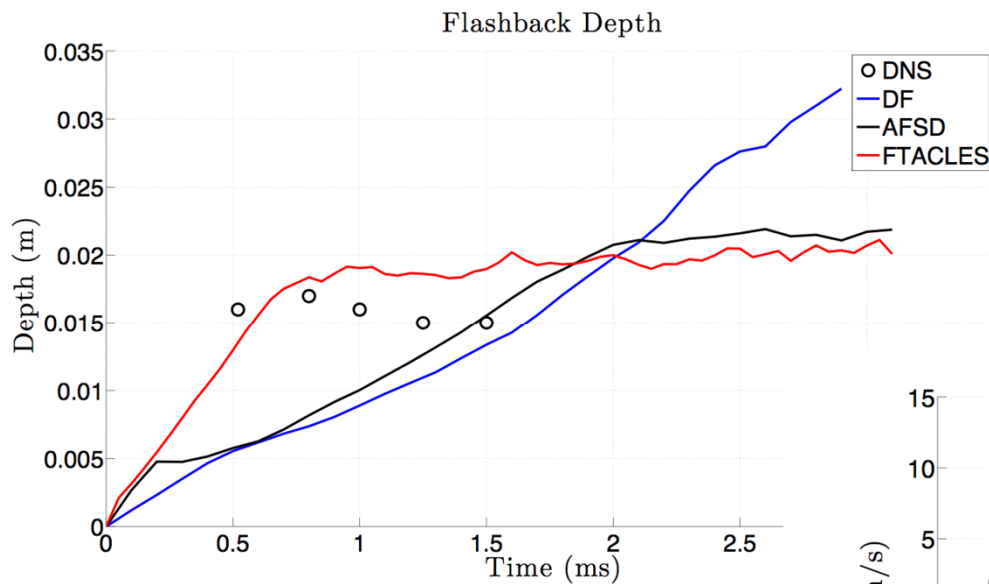
- LES conducted for different grid sizes
 - Filtered flame model used
- ➔ FTACLES approach of Fiorina and co-workers



FW = Filter Width, indicates ratio of LES to DNS grid size

LES Results - Model Performance

- Different source term approximations for progress variable tested
- FTACLES approach determined to be most suitable



Moving Beyond Averages

Computational Modeling



- Computational modeling a.k.a CFD targets statistical stationarity
 - ➔ Flow does not change with time
 - ➔ Flow is turbulent but the mean is constant
 - ➔ Why?
 - Allows for “Equilibrium Assumptions”
- What can CFD do?
 - ➔ Predict mean evolution of quantities
 - Average NOx at outlet
 - Mean and fluctuations of temperature
 - ➔ Cannot be trusted for transient problems

Fundamentals of CFD Modeling



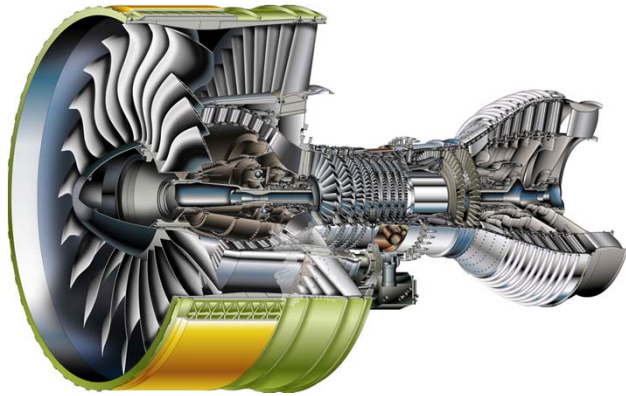
- At core of all CFD models lies Equilibrium Assumption (EA)
 - ➔ Not a single assumption but spans a suite of assumptions
- Examples of EA
 - ➔ At many different scales
 - Molecular thermodynamics (thermal equilibrium)
 - Spectral equilibrium (turbulence and scalar spectrum are similar)
 - Turbulence equilibrium (established spectrum)
- Why EA?
 - ➔ Makes modeling simpler (which is the goal of modeling)
 - ➔ Valid in many situations

EA with Averaging

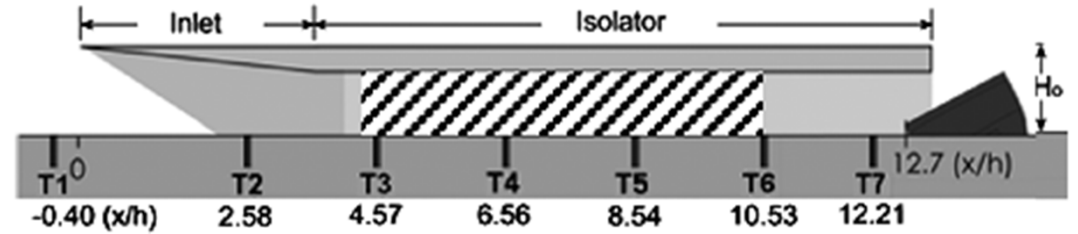
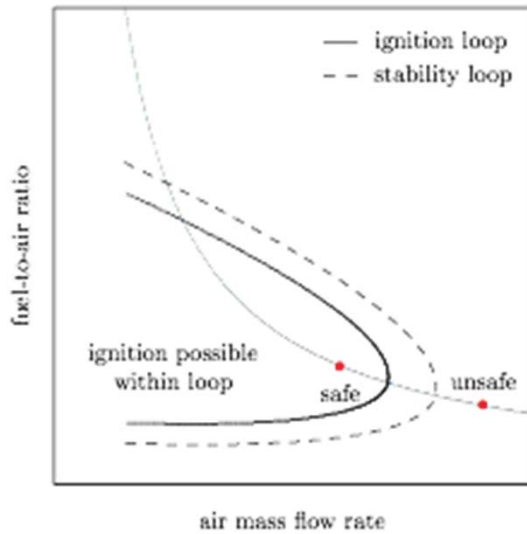


- All turbulence simulations use some form of averaging
 - ➔ RANS uses ensemble averaging
 - ➔ LES uses spatial averaging followed by ensemble averaging
 - The second part is not normally discussed
 - Important for transient flow problems
- Averaging further limits the utility
 - ➔ Turbulent flow is chaotic
 - ➔ Predicting average events is useful, but not critical
 - ➔ More importantly, experiments are ideally suited for this purpose
 - Simulations may not “predict” new information not already obtainable
 - Granted, experiments are expensive!

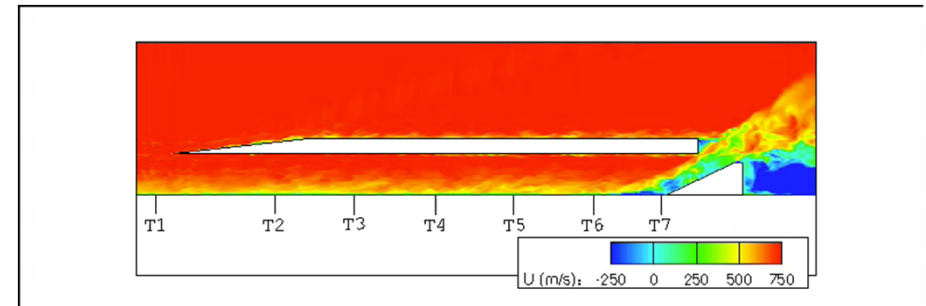
Rethinking CFD: Motivating Physics



High Altitude Relight



Supersonic Isolator Unstart



Flame Flashback in Confined Geometries

Changing the Simulation Target



- Simulations are designed to predict this:

- What is the average speed of flashback?

- Simulations should predict this:

- What is the probability that the flashback speed $>$ some value?

Or

- What is the fastest propagation speed?

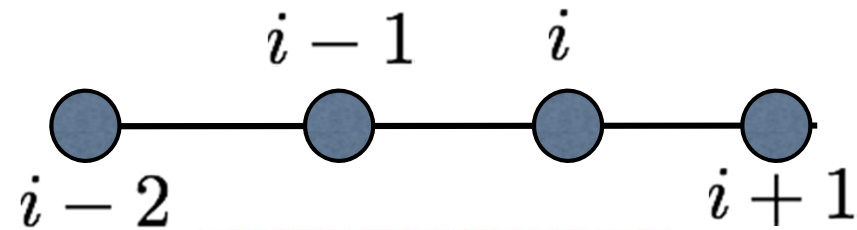
Three Approaches to Understanding LES



- Adrian (1977) provided one of the first studies on the implied meaning of filtering
 - ➔ Discussed this in terms of two different simulations approaches
 - ➔ Termed here as coarse DNS and filtered LES approaches
- Moser's ideal LES approach (1998)
 - ➔ Similar to Adrian's second approach
- Pope's self-conditional LES approach (2010)
 - ➔ Seeks to restate the CFD modeling problem
 - ➔ Unique model terms arise

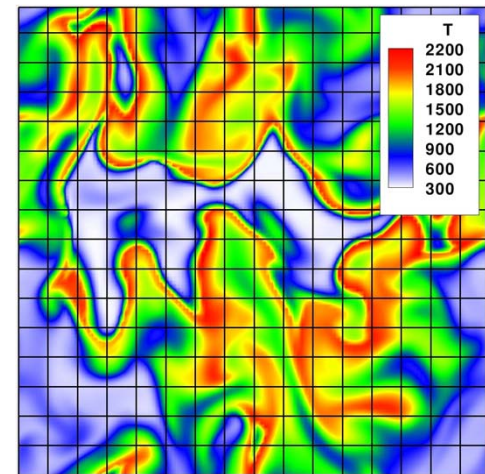
Statistical Definition of LES

- Consider a continuous velocity field $\mathbf{u}(\mathbf{x}, t)$



- Consider a computational grid of discrete points

→ Mesh points are spaced larger than the smallest flow scales



- Knowledge of \mathbf{w}_i alone does not determine the evolution of $\mathbf{u}(\mathbf{x}, t)$

→ For a given \mathbf{w}_i there are multiple possible transitions from $\mathbf{u}(\mathbf{x}, t)$ to $\mathbf{u}(\mathbf{x}, t + \Delta t)$

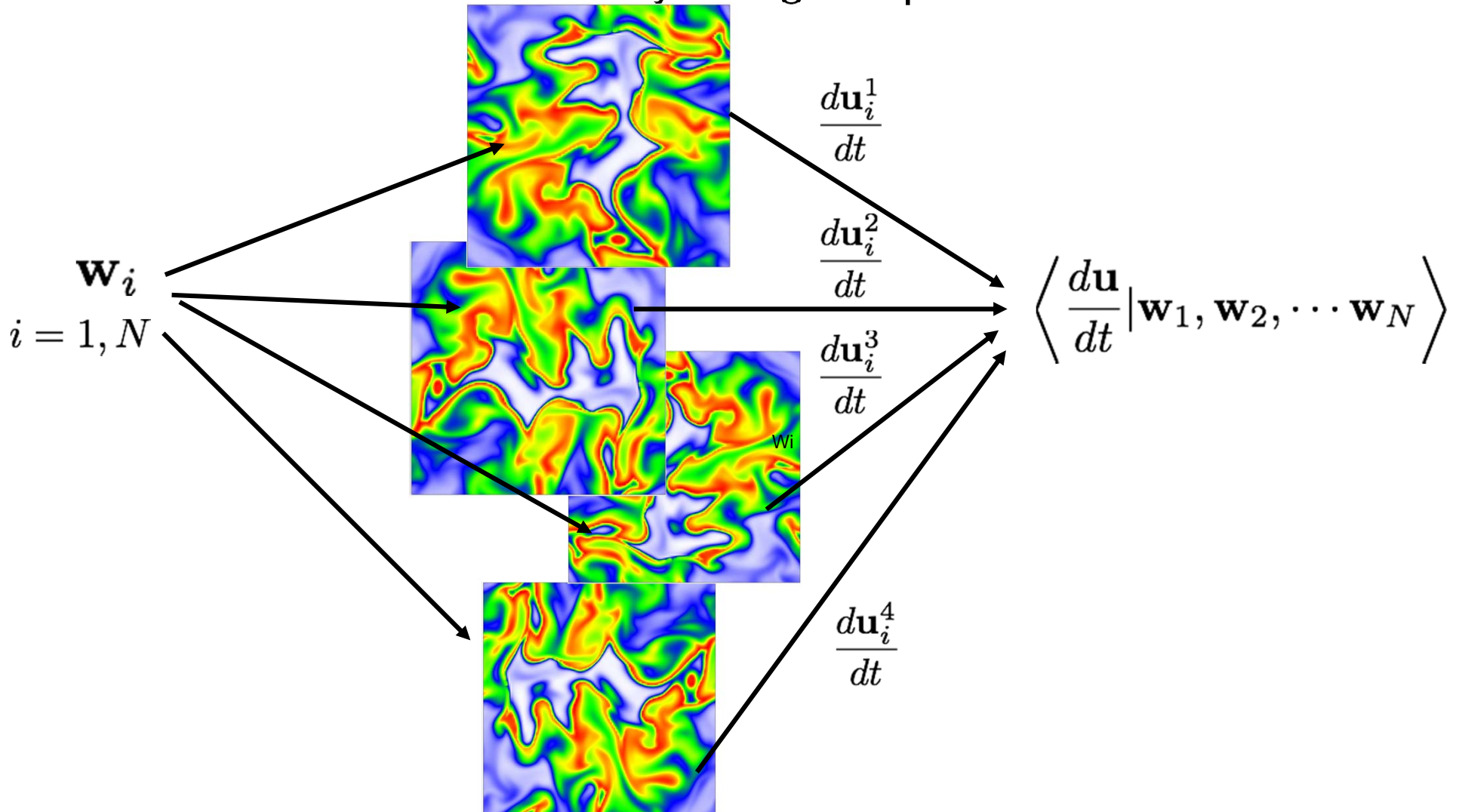
→ Transition should be described probabilistically

- How do we evolve \mathbf{w}_i ?

Understanding Filtered Evolution

- Consider representation of velocity on a mesh

→ \mathbf{W}_i is the vector of velocity at a given point



Multi-point Probability Density Function



- Consider the following event

$$\mathbf{E}_n = \{ \mathbf{v}_1 < \mathbf{u}(\mathbf{x}, t) < \mathbf{v}_1 + d\mathbf{v}_1, \mathbf{v}_2 < \mathbf{u}(\mathbf{x}, t) < \mathbf{v}_2 + d\mathbf{v}_2, \dots, \mathbf{v}_n < \mathbf{u}(\mathbf{x}, t) < \mathbf{v}_n + d\mathbf{v}_n \}$$

- ➔ n refers to the number of grid points in the computational grid
- ➔ The event refers to multi-point velocity information

- The joint-PDF evolves according to

$$\frac{\partial P}{\partial t} + \frac{\partial}{\partial \mathbf{v}_n} \left(\left\langle \frac{d\mathbf{u}}{dt} \middle| \mathbf{E}_n \right\rangle P \right) = 0.$$

- The best solution from the LES reproduces the multi-point PDF accurately

Conditional Evolution

Equations

● How do we evolve \mathbf{w}_i ?

→ The solution should capture the multi-point PDF correctly

$$\mathbf{w}_\alpha(t + \Delta t) = \mathbf{w}_\alpha(t) + \left\langle \frac{d\mathbf{u}}{dt} \middle| \mathbf{E}_n = \{\mathbf{w}_1, \mathbf{w}_2, \dots, \mathbf{w}_n\} \right\rangle \Delta t$$

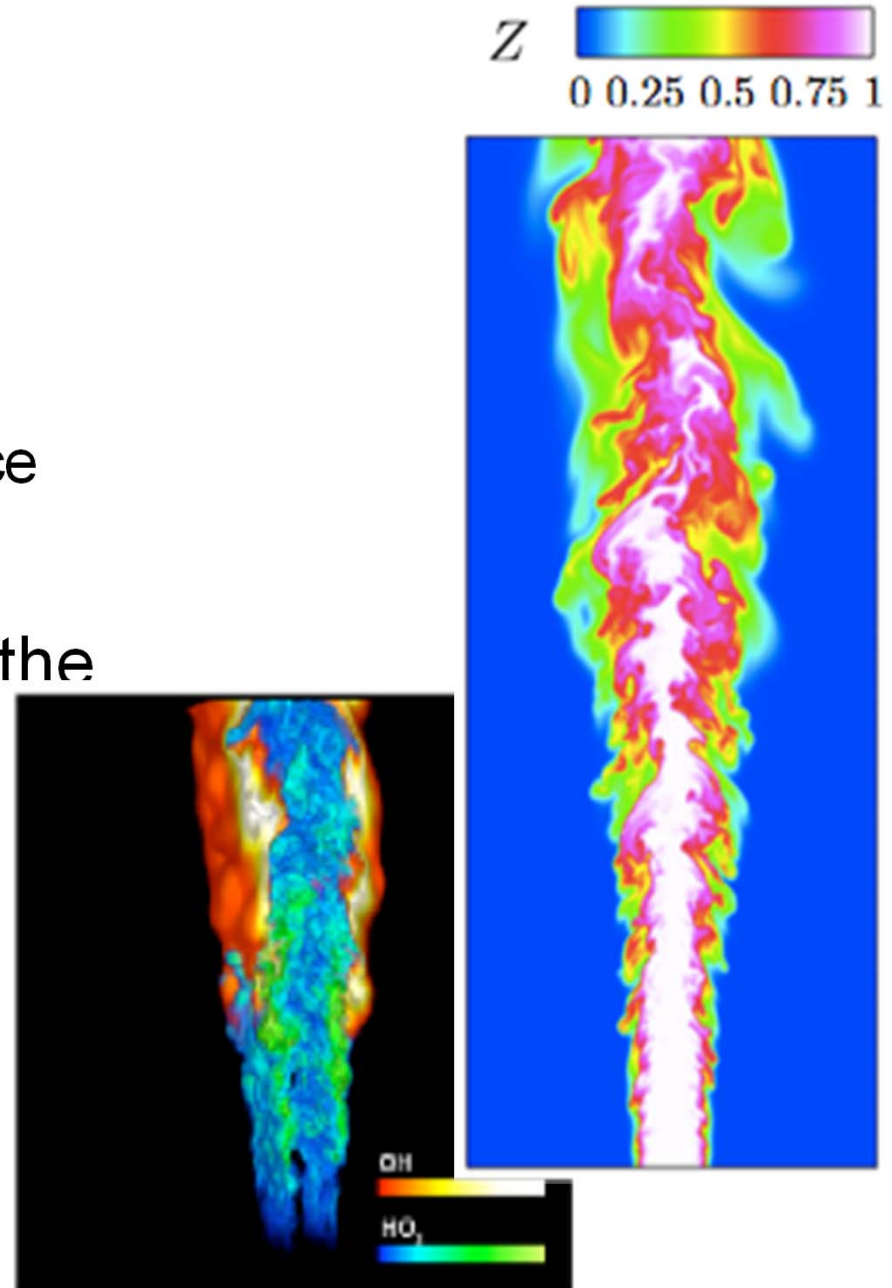
- The best solution evolves the conditional mean of all possible realizations

→ Note that the best solution evolves the average path and is a statistically averaged result

- It is important to think of LES computations also in such average terms

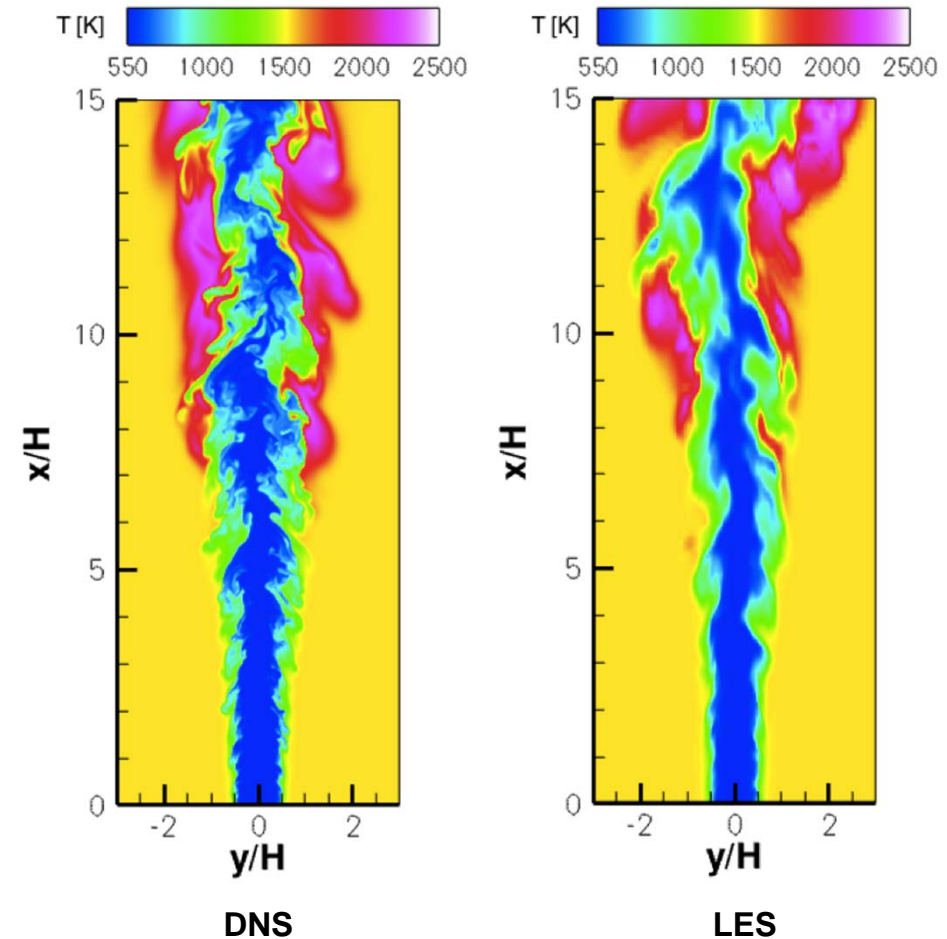
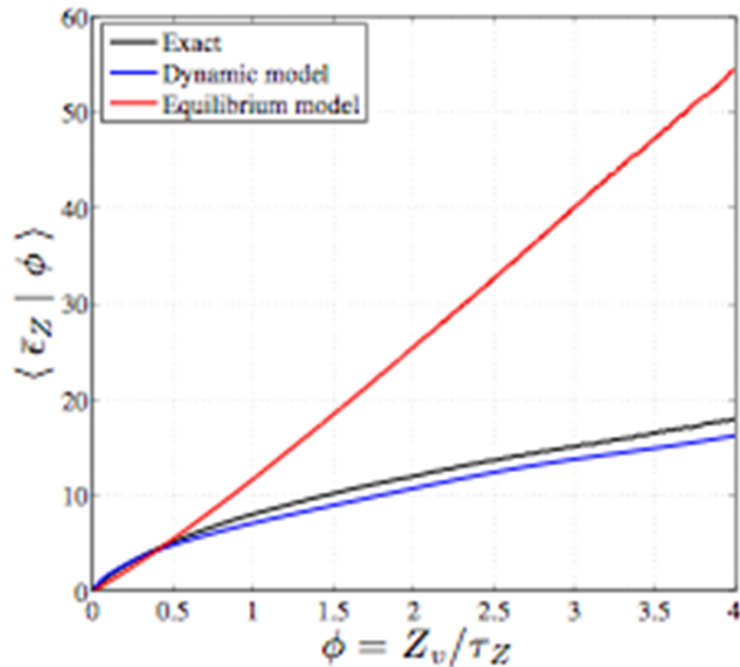
Developing Conditional Models in LES

- From conditional evolution equations, a set of model terms could be extracted
 - ➔ Similar to conventional LES
 - ➔ Unresolved stress, subfilter variance etc.
- Models in LES should also obey the conditional average formulation
- Consider scalar dissipation rate
 - ➔ Very important for combustion simulations
- Lifted flame configuration



Conditionally Averaged Combustion Models

- New optimal estimator based model selection
 - ➔ Provides an estimate of the least error that could be made with a given set of input variables
 - ➔ Model form chosen to be close to this error



From Kaul et al. (2013), Proc. Comb. Inst.

Multi-time Formulations



- CFD has to move beyond one-point one-time models

→ Multi-time models for transient flow behavior

$$\frac{d\mathbf{w}}{dt} = \left\langle \frac{d\mathbf{u}}{dt} \mid \mathbf{w}(t), \mathbf{w}(t - \Delta t), \dots \right\rangle$$

- Current project

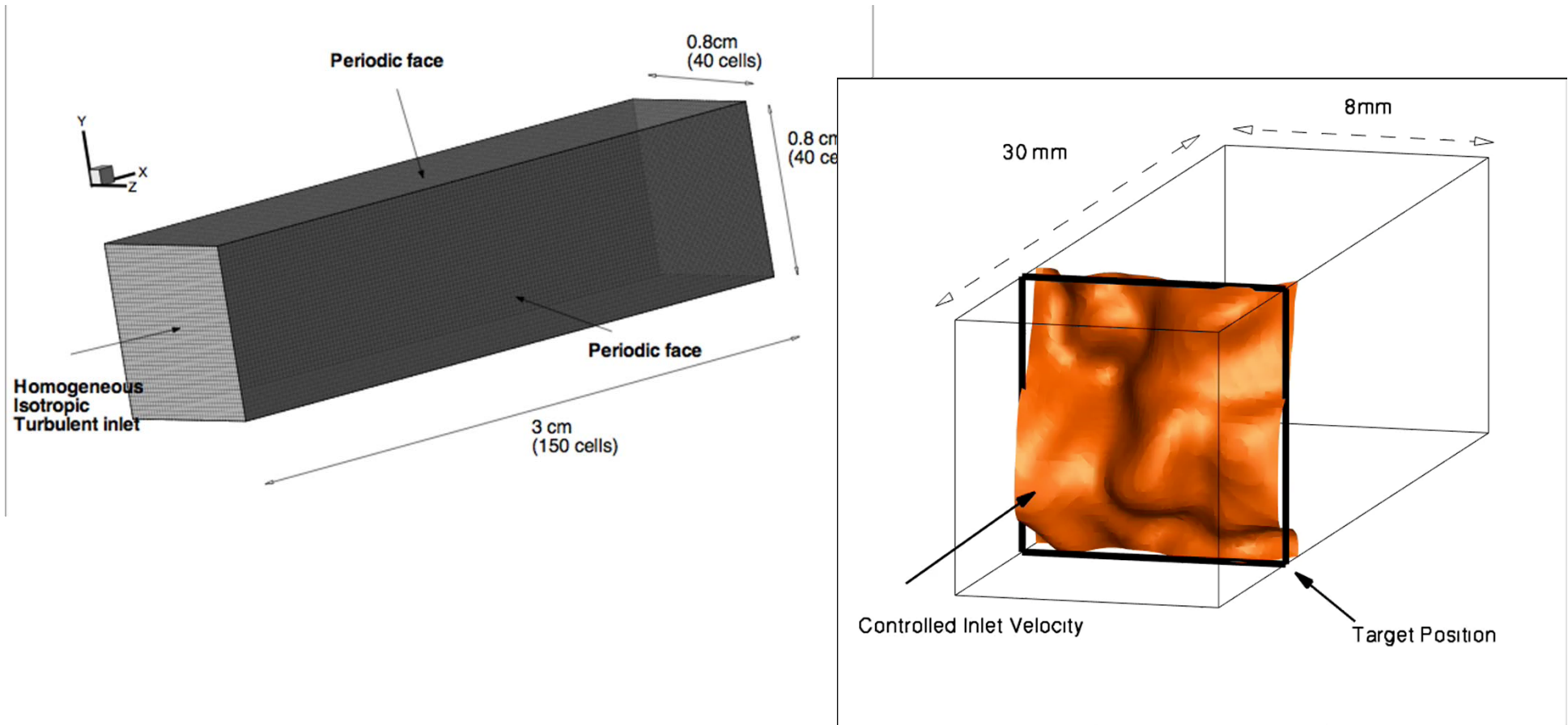
→ Understand variability in flashback

→ Devise modeling methods for predicting “extreme events”

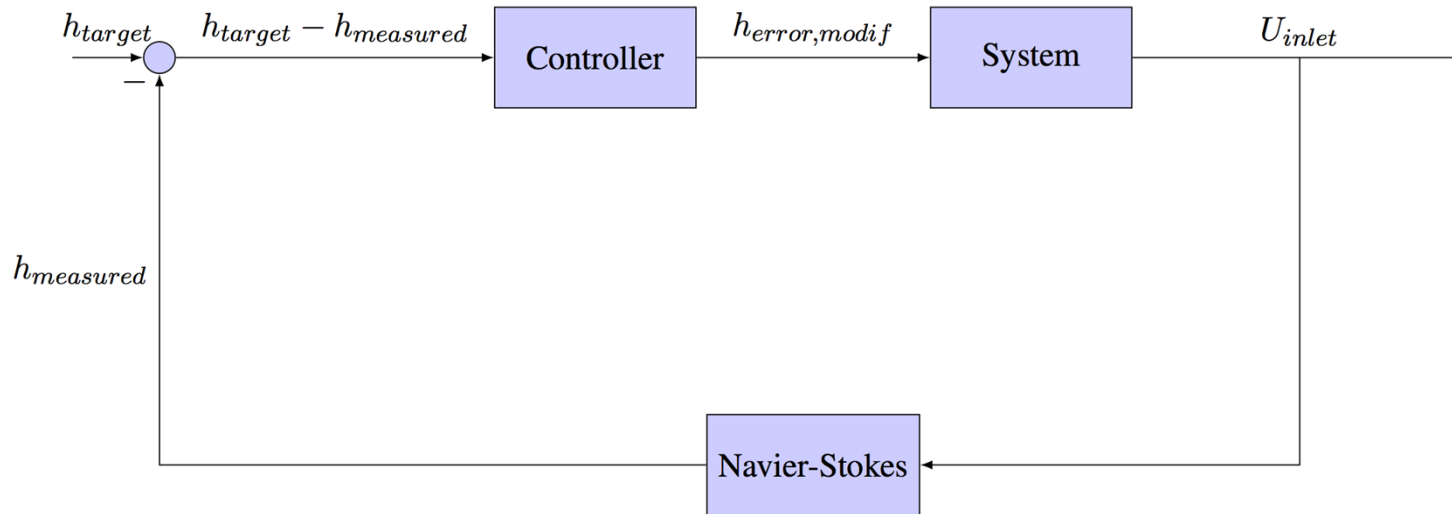
→ Map the limitations of LES in predicting such transient flows

Constrained Premixed Flame

- Flame propagation in homogeneous isotropic turbulence
- Flame location fixed using a control loop

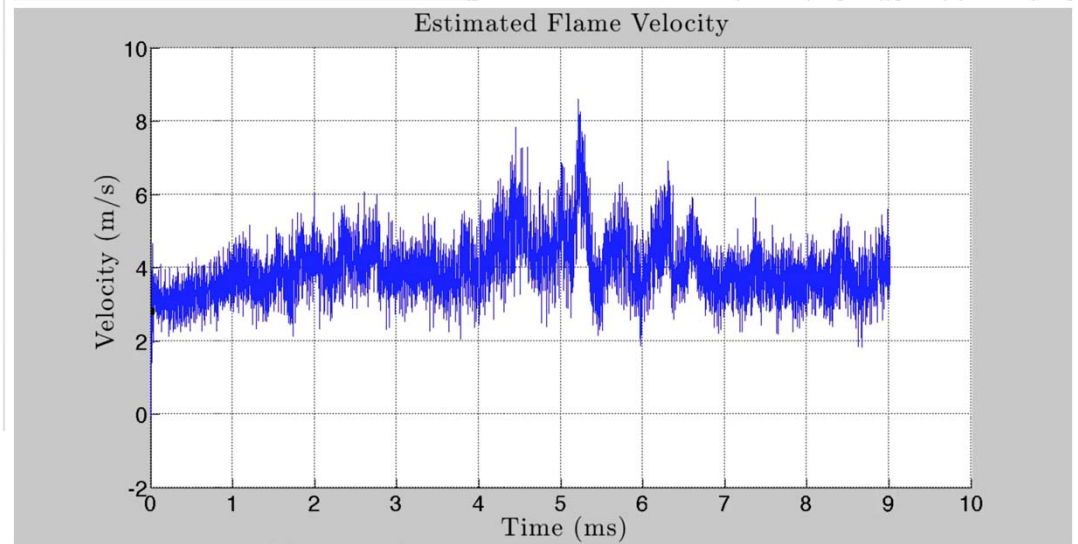
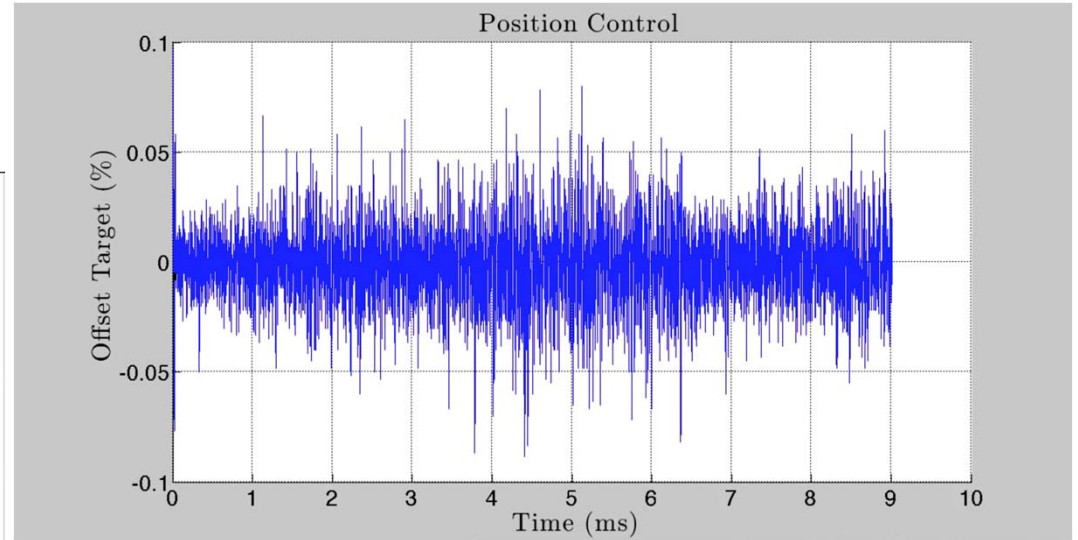
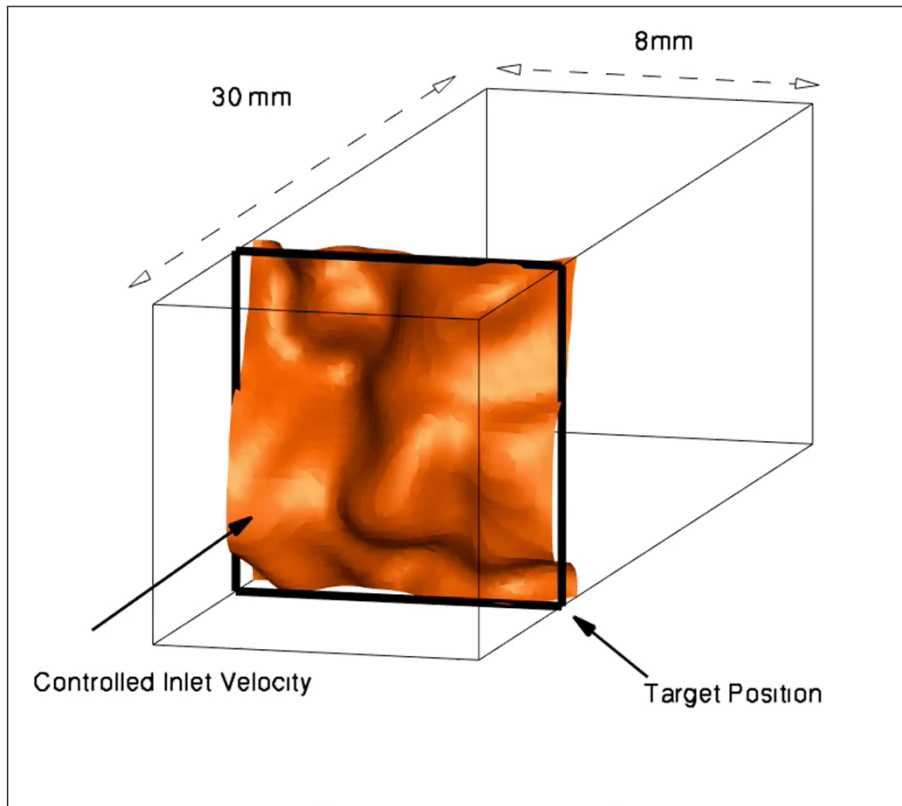


Position Control Algorithm



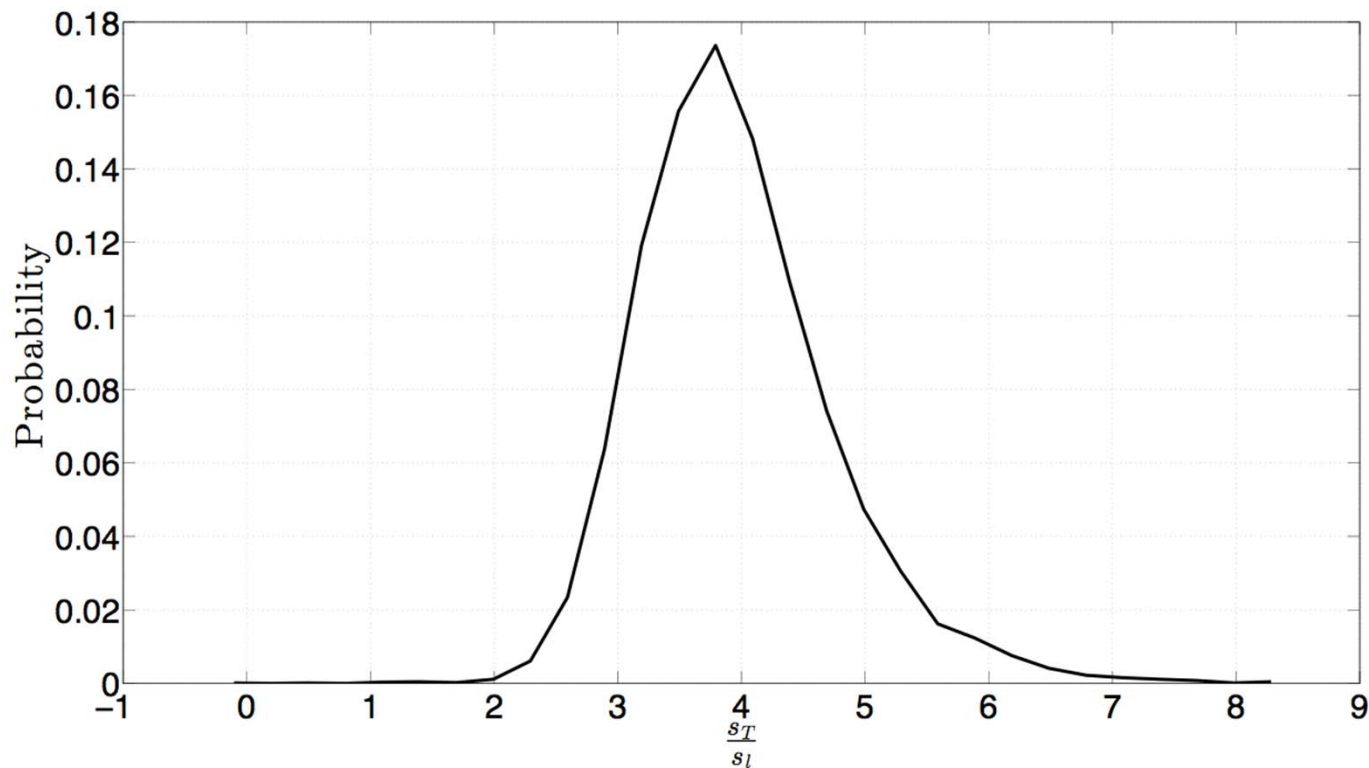
- Flame position adjusted by changing inflow velocity
 - ➔ Response time adjusted to ensure stability
 - ➔ Total adjustment a small fraction of the flame propagation velocity

Flame Evolution



PDF of Flame Propagation Velocity

- Strong asymmetry in flame propagation
 - ➔ Faster velocities are more common than slower velocities



Conclusions



- High-pressure setup constructed
 - ➔ Initial stratified flame studies underway
- LES of flashback
 - ➔ Progress variable approach predicts DNS statistics reasonably accurately
 - ➔ Flame wrinkling effects at larger filter widths need to be studied
- New modeling strategy for CFD
 - ➔ Towards probabilistic modeling of transient flows
 - ➔ Homogeneous cases used to understand time correlation of extreme events