

Computational Microstructural Optimization Design Tool for High Temperature Structural Materials

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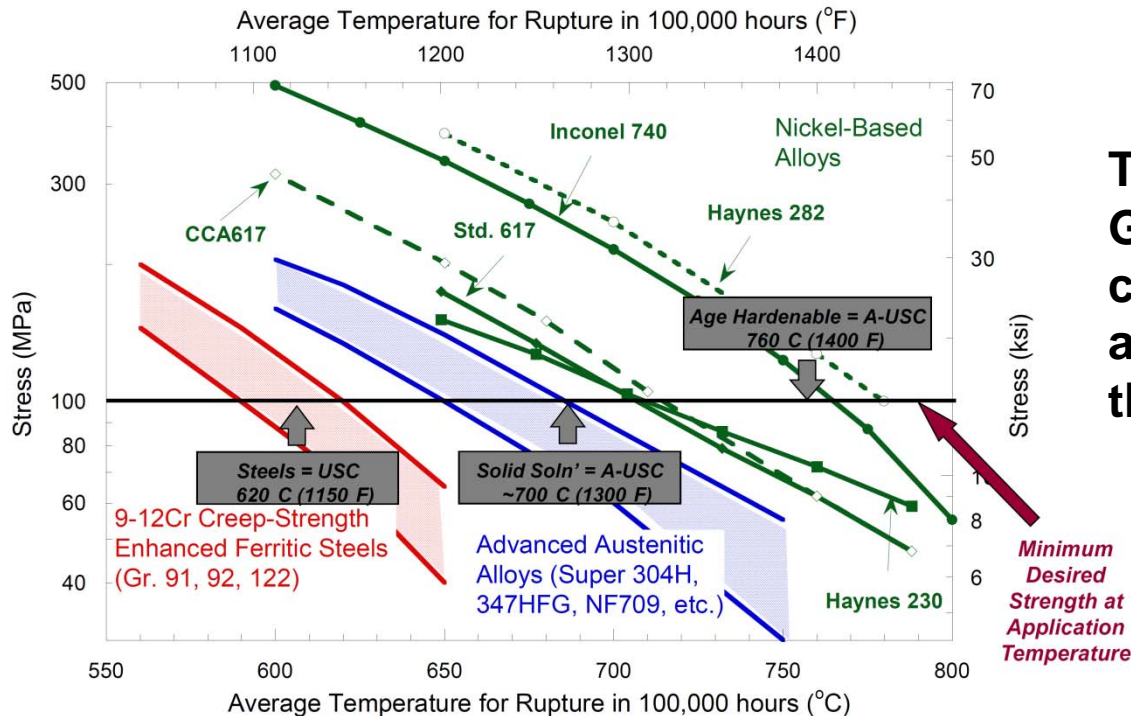
**2014 NETL Crosscutting Research Review Meeting
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Objectives

- Develop a methodology for microstructural optimization of alloys - genetic algorithm approach for alloy microstructural optimization using theoretical models based on fundamental micro-mechanisms, and
- Develop a new computationally designed Ni-Cr alloy for coal-fired power plant applications.

Robert R. Romanosky, National Energy Technology Laboratory, April 2012

Materials Limit the Current Technology



This translates into a **GOAL** of minimum creep rate $<2.7 \times 10^{-9} \text{ s}^{-1}$ at 100 MPa at 800 °C for this project

Timeline of dislocation-particle strengthening

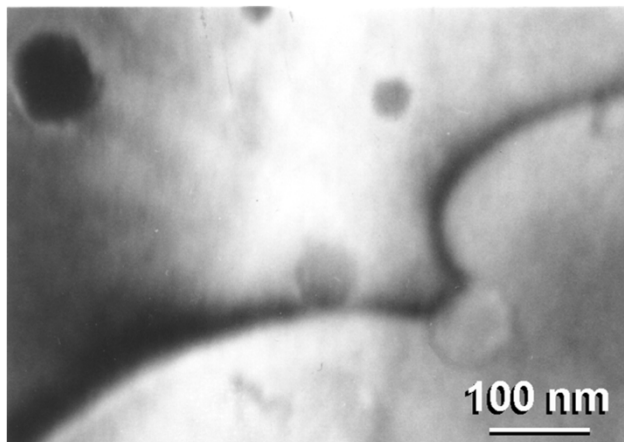
- Dispersion strengthening identified as a potent mechanism for enhancing elevated temperature strength in the early works of Ansell and Weertman in 1950s
 - CONCEPT- Elastically hard particle repels dislocation
- Srolovitz and co-workers in 1980s
 - FUNDAMENTAL SHIFT- dislocation-particle interaction undergoes repulsive→attractive transition at elevated temperatures $>0.35 T_m$

Background - RECAP

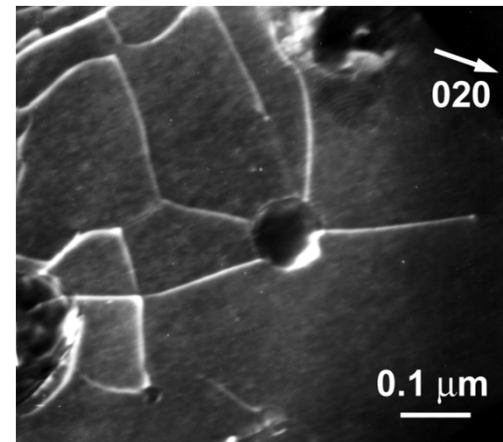
Summary of some of the key development made possible by TEM studies

Reference	Remarks
Nardone and Tien (1993)	First identification of departure side pinning.
Schroder and Arzt (1985)	Weak-beam micrographs showing clear dislocation contrast at the dispersoid.
Herrick et al. (1988)	First quantification of (a) percentage dislocation looped vs. attached, and (b) critical take-off angle as a function of temperature.
Liu and Cowley (1993)	Multiple dislocation-particle interaction; sharp kinks on the detached dislocations that straighten out.

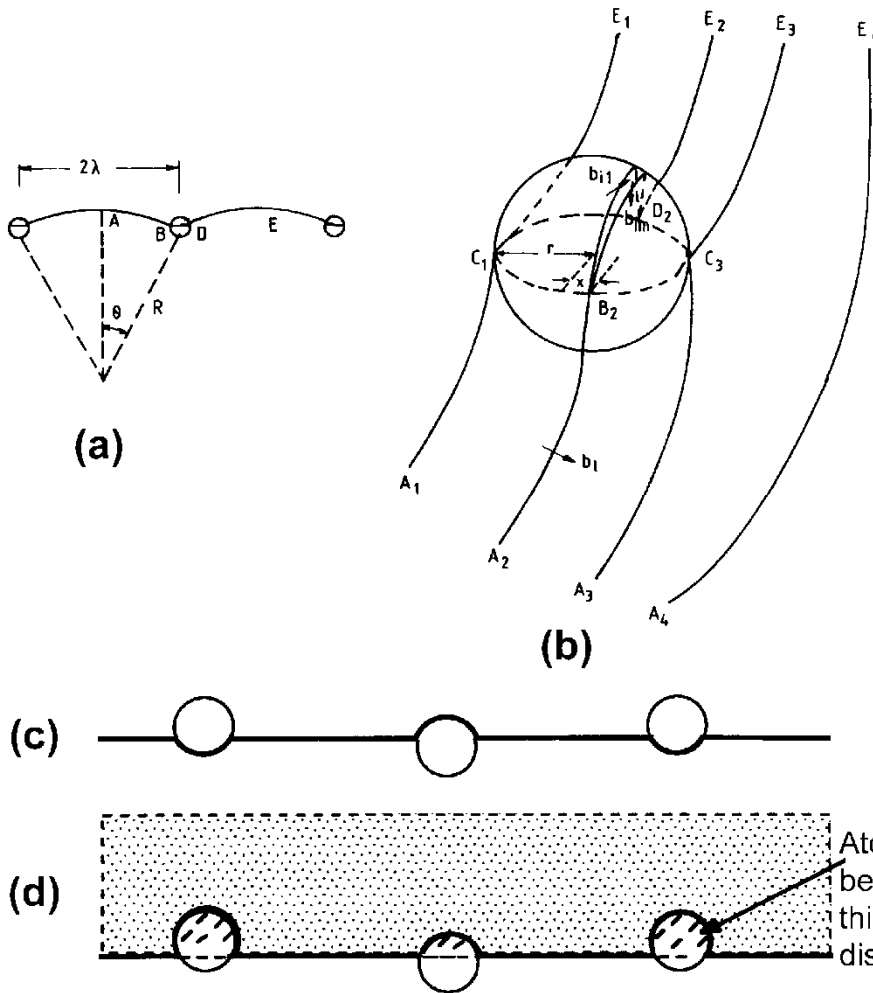
A dispersion strengthened platinum alloy
(Heilmaier *et al.* 1999)



Al-5 wt.% Ti alloy
(Mishra and Mukherjee 1995)



Development of dissociation and positive climb concepts



(a) and (b) A schematic illustration of dissociation of dislocation at matrix-particle interface that can result in an attractive dislocation-particle interaction (Mishra et al. 1994).

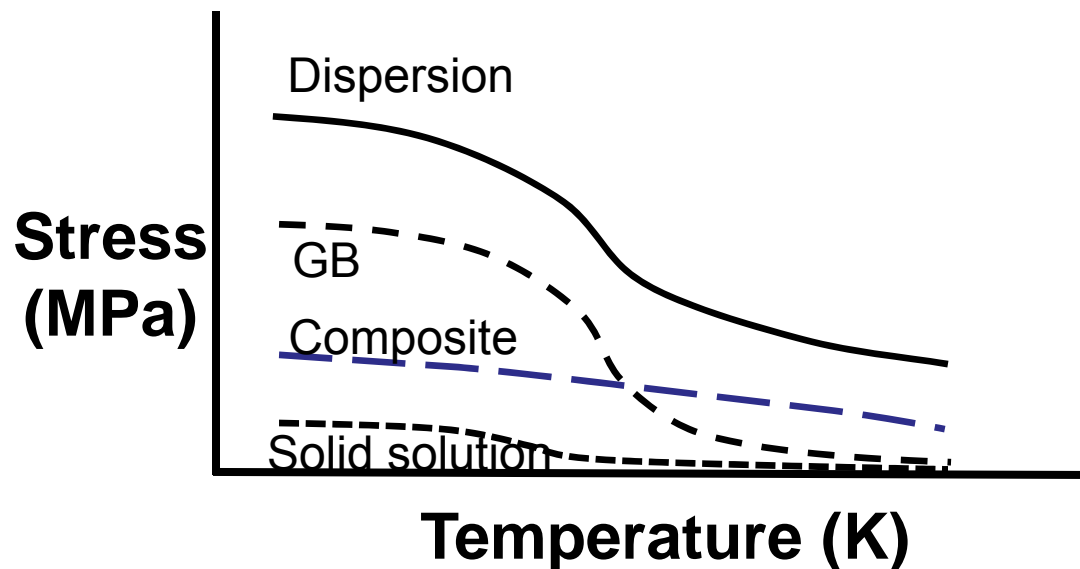
(c) Up and down climb concept of Shewfelt and Brown (1977) and Arzt and Ashby (1982).

(d) A modified concept of 'positive climb' (Mishra and Mukherjee 1995).

There are four major components to strengthening in the nanostructured nickel based alloys produced by mechanical alloying:

- grain boundary strengthening,
- solid solution strengthening
- dispersion strengthening, and
- composite strengthening.

Effect of temperature



What are the additivity rules?

$$\sigma_{alloy} = \sum \sigma_i$$

$$\sigma_{alloy} = \sqrt{\sum (\sigma_i)^2}$$

$$(\sigma_{alloy})^k = \sum \sigma_i^k$$

Proposed Microstructure

Develop dual-scale strengthened Ni-Cr-Al₂O₃ alloys

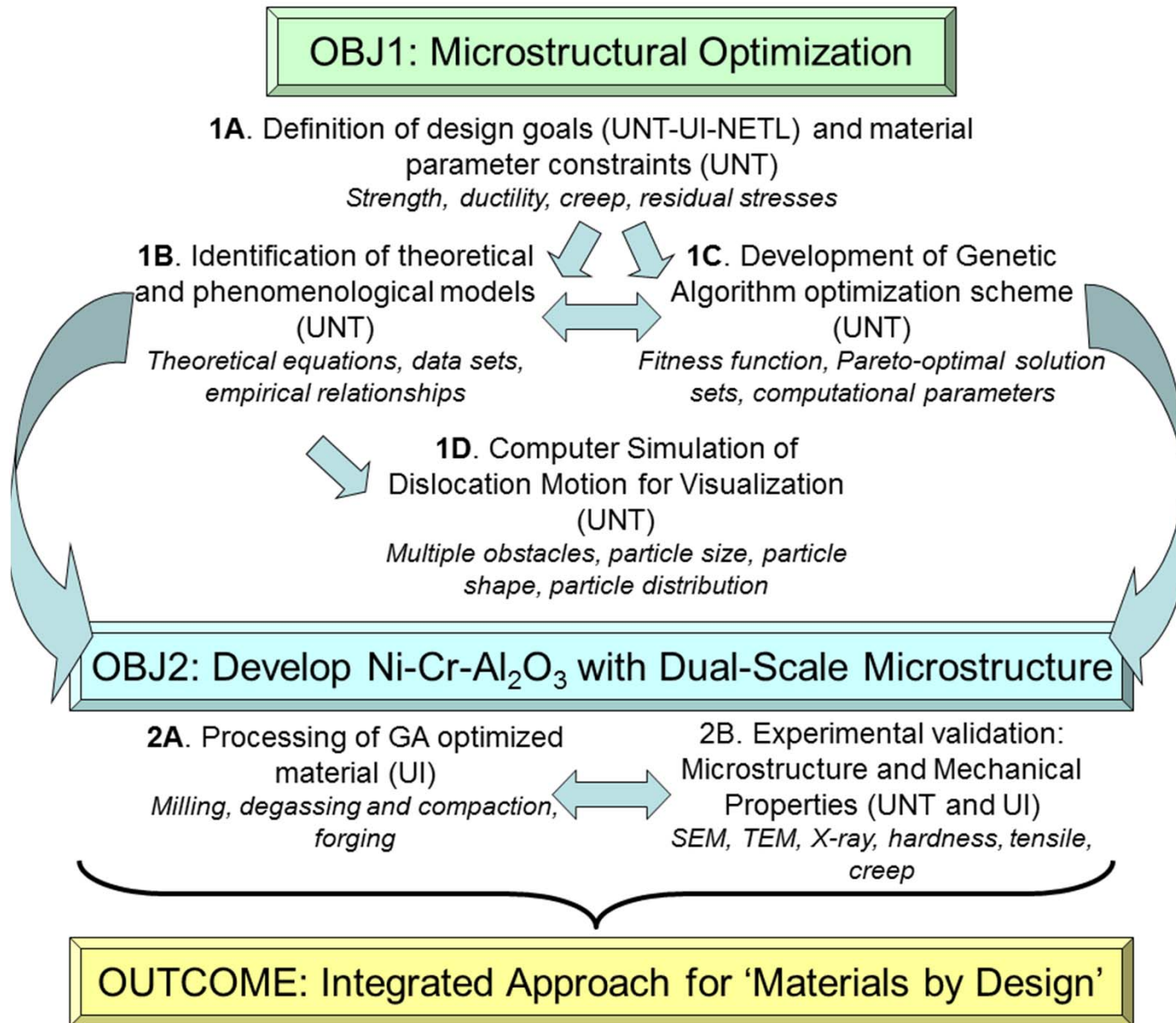
The chosen alloy system has:

- Cr for solid solution strengthening
- nano Cr₂O₃ and/or CrN particles of 2-3 nm diameter for dispersion (currently using nano-Y₂O₃) strengthening
- submicron Al₂O₃ of 0.5-1 micron diameter for composite strengthening through increase in modulus

What is the level of synergy?

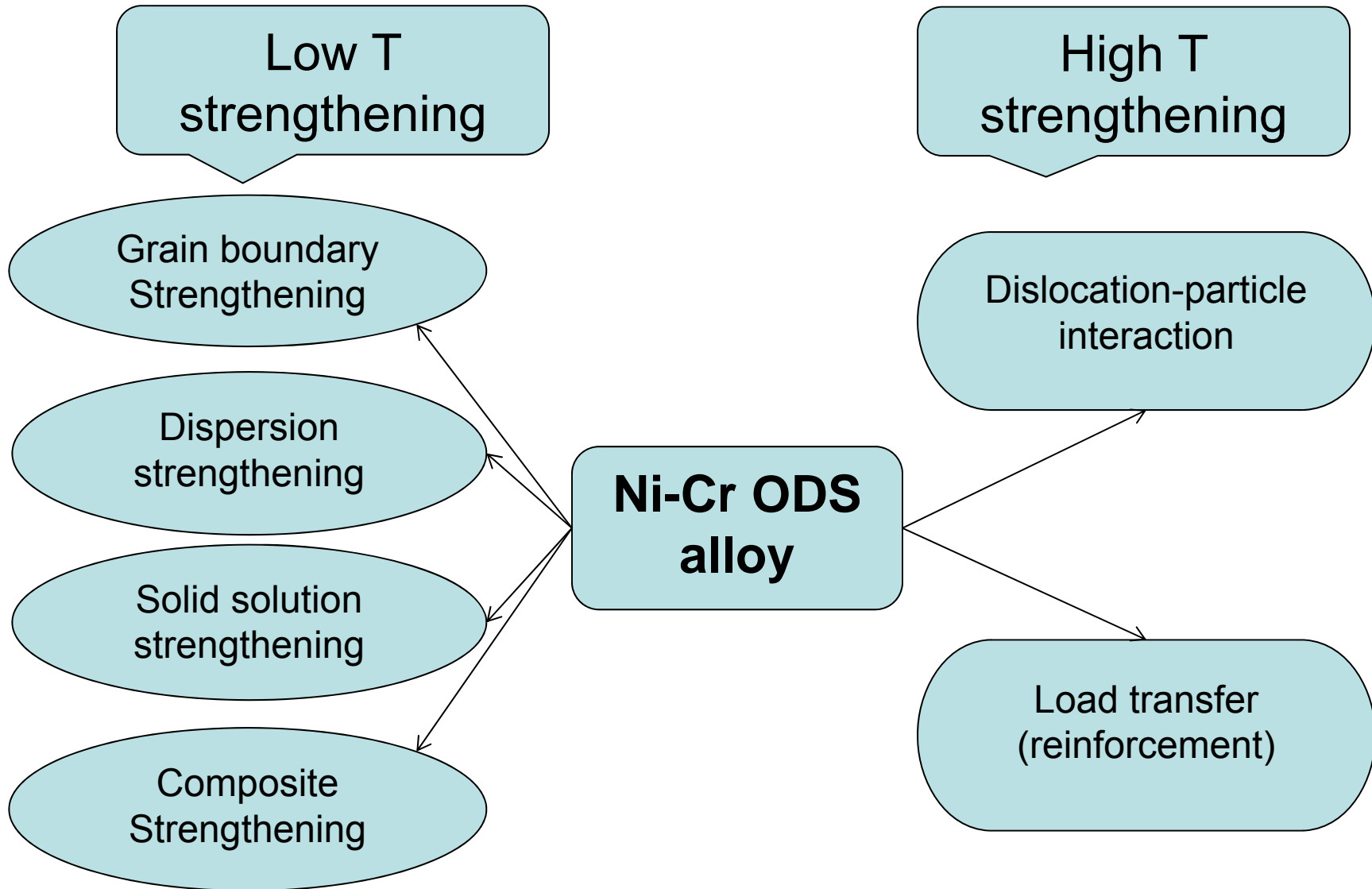
- Does the load transfer effectively enhance the creep life for equiaxed reinforcement?

Overview of This Project



Computational part

Strengthening Mechanisms



Low temperature strength

Strengthening mechanism	Equation
Grain size strengthening	$\sigma_y = \sigma_0 + Kd^{-0.5}$
Solid solution strengthening	$\Delta\sigma_s = \left(\sum k_i^{\frac{1}{n}} c_i \right)^n$
Dispersion strengthening	$\Delta\sigma_p = \frac{Gb\sqrt{f_d}}{d_p}$
Composite strengthening Load transfer coefficient	$\sigma_c = V_p\sigma_p + V_m\sigma_m$ $\Lambda \approx 1 + 2 \left(2 + \frac{l}{R} \right) f_r^{\frac{3}{2}}$

Dislocation creep

Modified power law creep [1]

$$\dot{\epsilon} = 8.3 * 10^8 \frac{DGb}{k_B T} \left[\exp \left(-104 \sqrt{\frac{b}{\lambda}} \right) \right] \left(\frac{\sigma' - \sigma_0}{E} \right)^5$$

$$\sigma' = \sigma / \lambda$$

Threshold stress

Dissociation and positive climb model [2]

$$\sigma_0 = 0.002 G \left(\frac{b}{r} \right) \exp \left(20 \frac{r}{\lambda} \right)$$

1. R. S. Mishra and A. K. Mukherjee, Light weight alloys for aerospace application III, TMS, (1995), 319
2. R.S. Mishra *et al.*, Philosophical Magazine A, 1994, 69 (6), 1097-1109

Cost function

$$J = \frac{\left[\sum_{i=S,D,HTS} w_i \left| \left(\frac{P_i}{(P_i)_{desired}} \right) - 1 \right| \right]}{n}$$

Various considerations were taken in order to minimize the cost function:

- 100 Individuals were considered in each generation.
- Rank scales were used for the fitness scaling. The rank of the fittest individual was 1, the next fittest was 2 and so on.
- Different methods were used as a selection function to choose parents for the next generation.
- 10 best individuals survived to the next generation.
- Probability of crossover was chosen 0.85 and rest were produce via mutation.
- The optimization was running until 100 generations were completed or the cost function did not vary significant for 25 successive generations.

Notation used for variables:

- $[w_S w_D w_{HTS}]$ = Weight factors for low temperature strength, ductility and high temperature strength properties.
- r (nm) is the radius of dispersoids.
- $r1$ (nm) and $r2$ (nm) are radius of two different dispersoids.
- r_f (nm) is the radius of reinforced particles.
- f_r (%) is volume fraction of reinforcement.
- f_d (%) is volume fraction of dispersoids.
- f_d1 (%) and f_d2 (%) are the volume fraction of two different dispersoids.

Optimization conditions:

I: $15 \text{ nm} \leq r \leq 20 \text{ nm}$, $300 \text{ nm} \leq r_f \leq 400 \text{ nm}$, $f_r \leq 15 \%$, $T=1073 \text{ K}$, $\dot{\epsilon} = 10^{-9} \text{ s}^{-1}$

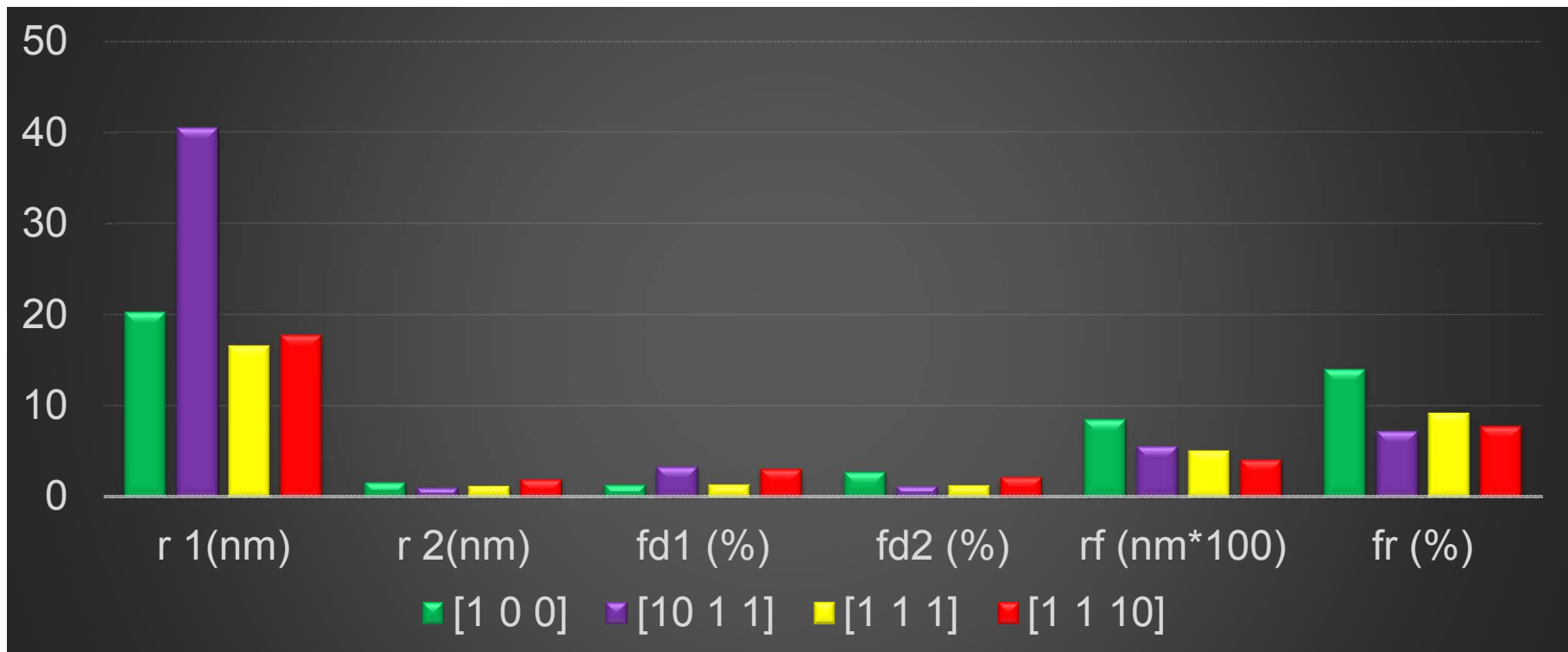
II: $10 \text{ nm} \leq r1 \leq 100 \text{ nm}$, $1 \text{ nm} \leq r2 \leq 3 \text{ nm}$, $400 \text{ nm} \leq r_f \leq 1000 \text{ nm}$, $f_r \leq 15 \%$,
 $T=1073 \text{ K}$, $\dot{\epsilon} = 10^{-9} \text{ s}^{-1}$

III: $1 \text{ nm} \leq r \leq 30 \text{ nm}$, $100 \text{ nm} \leq r_f \leq 1000 \text{ nm}$, $f_r \leq 15 \%$, $T=1073 \text{ K}$, $\dot{\epsilon} = 10^{-9} \text{ s}^{-1}$

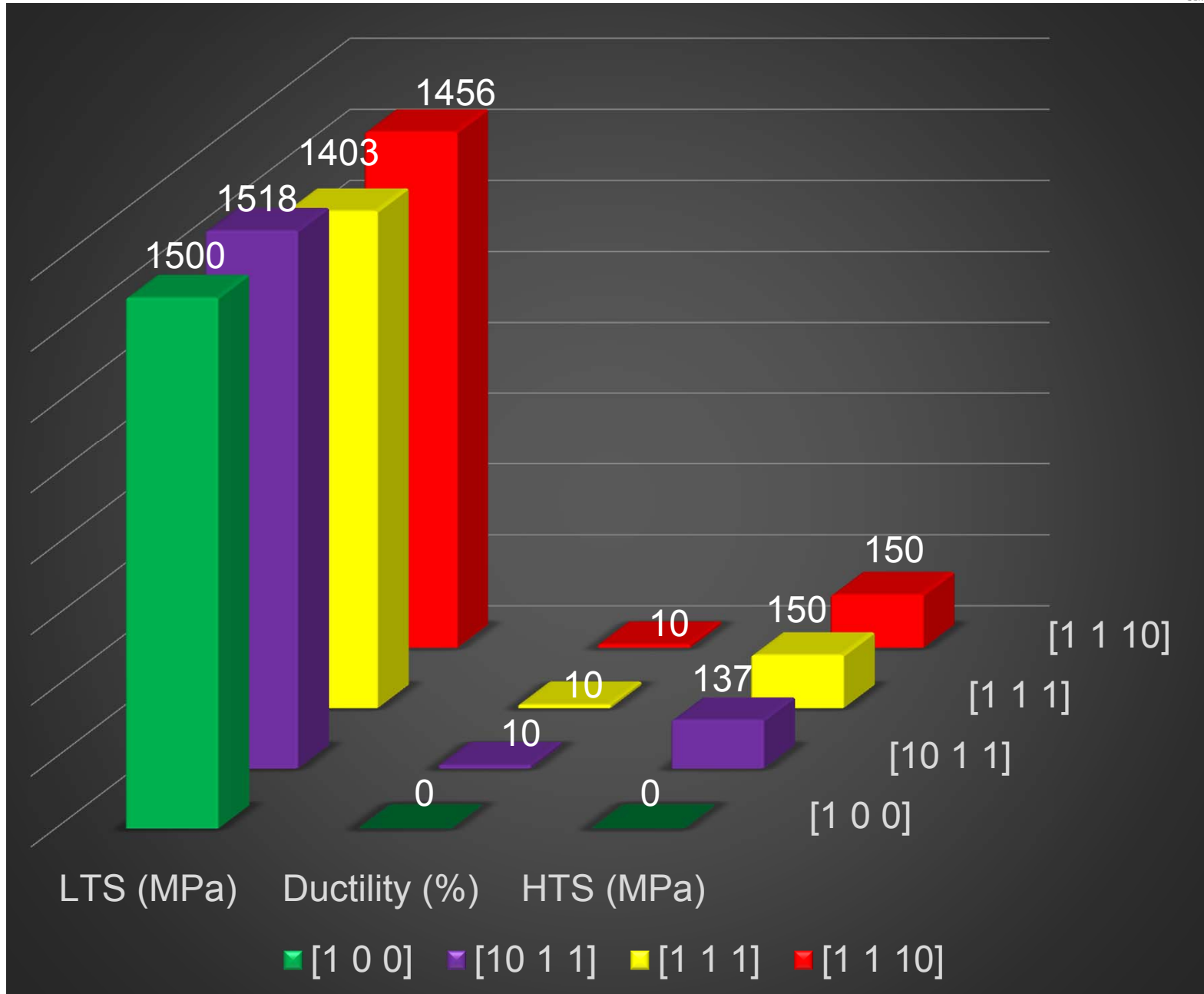
IV: $10 \text{ nm} \leq r1 \leq 100 \text{ nm}$, $1 \text{ nm} \leq r2 \leq 3 \text{ nm}$, $400 \text{ nm} \leq r_f \leq 1000 \text{ nm}$, $f_r \leq 15 \%$,
 $T=1073 \text{ K}$, $\dot{\epsilon} = 10^{-9} \text{ s}^{-1}$

GA results

Condition IV: $10 \text{ nm} \leq r_1 \leq 100 \text{ nm}$, $1 \text{ nm} \leq r_2 \leq 3 \text{ nm}$, $400 \text{ nm} \leq r_f \leq 1000 \text{ nm}$,
 $f_r \leq 15 \%$, $T=1073 \text{ K}$, $\dot{\epsilon} = 10^{-9} \text{ s}^{-1}$



GA operator	Selection	Crossover	Mutation
	Tournament	Arithmetic	Adaptive feasible



Summary for computational part

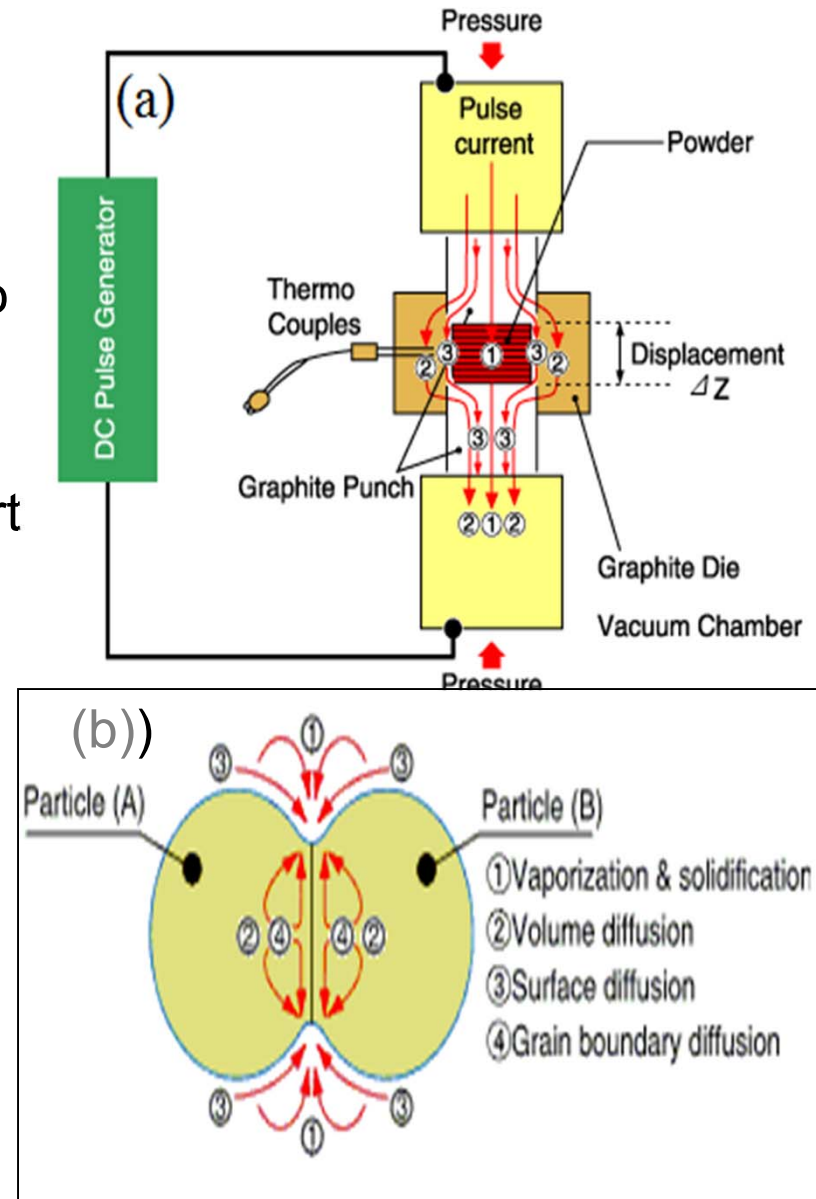
- The optimized results showed:

Condition I	Dispersoids radius (nm) ~ 15
	LTS (MPa) ~ 700
	HTS (MPa) ~ 40
Condition II	Dispersoids radii (nm) ~ 16, 2
	LTS (MPa) ~ 1200
	HTS (MPa) ~ 150
Condition III	Dispersoids radius (nm) ~ 2.5
	LTS (MPa) ~ 1373
	HTS (MPa) ~ 150
Condition IV	Dispersoids radii (nm) ~ 18, 2
	LTS (MPa) ~ 1456
	HTS (MPa) ~ 150

Experimental Part

- Develop fundamental understanding of microstructural characteristics and mechanical properties of the SPSed
 - Ni-20Cr,
 - Ni-20Cr-1.2Y₂O₃, and
 - Ni-20Cr-1.2Y₂O₃-5Al₂O₃ (wt%) alloys

- Hot Uniaxial Pressing with Joule heating by pulsed current
- Particle cleansing effect
- Metal or ceramic powder poured into dies (usually graphite)
- Rapid heating rates
- Near fully dense materials in as short as 5 min
- No texture or extrusion anisotropy
- Two dominant theories of SPS mechanisms
 - Plasma generation
 - Field theories



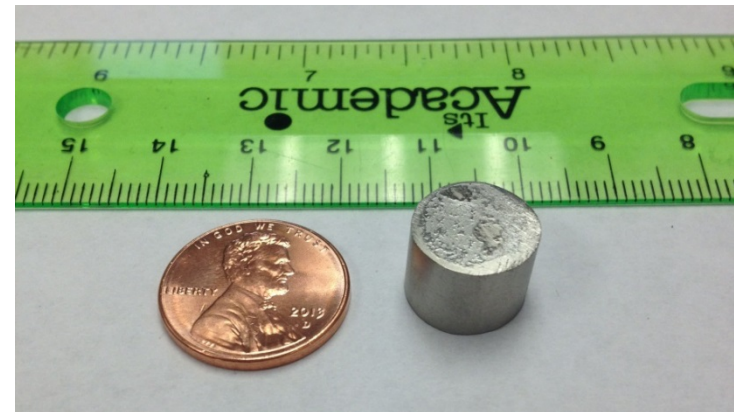
(M. Suárez et al., 2013)

Experimental – Spark Plasma Sintering

- Dr. Sinter 515S machine at the Center for Advanced Energy Studies (CAES), Idaho Falls
- Heating rate: 100 °C/min; applied pressure: ~80 MPa
- An intermediate 15 min dwell at 450 °C for 15 min (with 4.5 kN applied force) to remove the stearic acid
- Temperatures: 600 / 900 / 1000 / 1100 °C; dwell time: 5 and 30 min



Spark Plasma Sintering Machine



*Spark Plasma Sintered
Ni-20Cr-1.2Y₂O₃ Alloy*

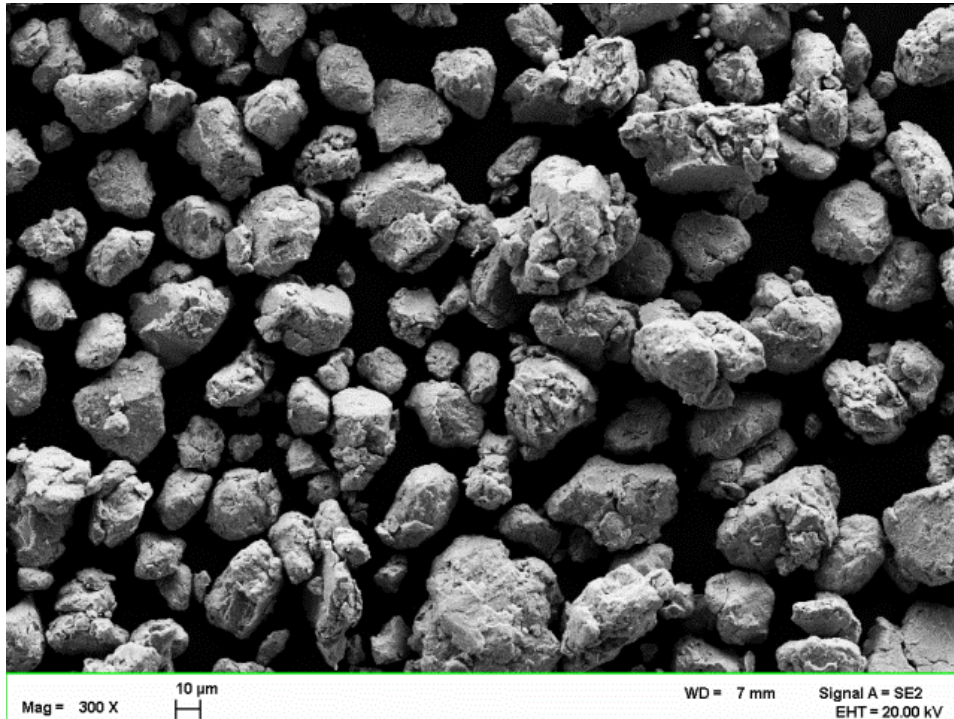
Microstructure – Ball Milled Powder

Ni-20Cr-1.2Y₂O₃ alloy

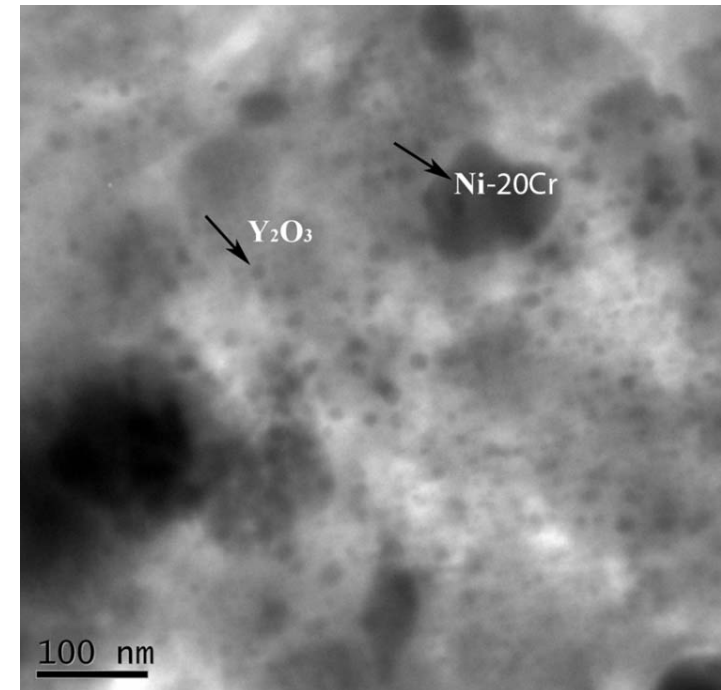
Milling Time (h)	Crystallite Size (nm)	Lattice Strain (%)	Lattice Constant (nm)	Mean Powder Size (μm)
0	44±12	0.03±0.001	0.3530±0.0002	23.6±1.1
1	17±9	0.03±0.001	0.3532±0.0003	39.2±2.2
2	14±7	0.03±0.001	0.3536±0.0003	33.6±1.5
4	4±2	0.15±0.003	0.3560±0.0004	39.4±3.1

Microstructural parameters quantified by XRD and SEM

Microstructure – Ball Milled Powder



A SEM micrograph of the ball milled (2) Ni-20Cr-1.2Y₂O₃ powder

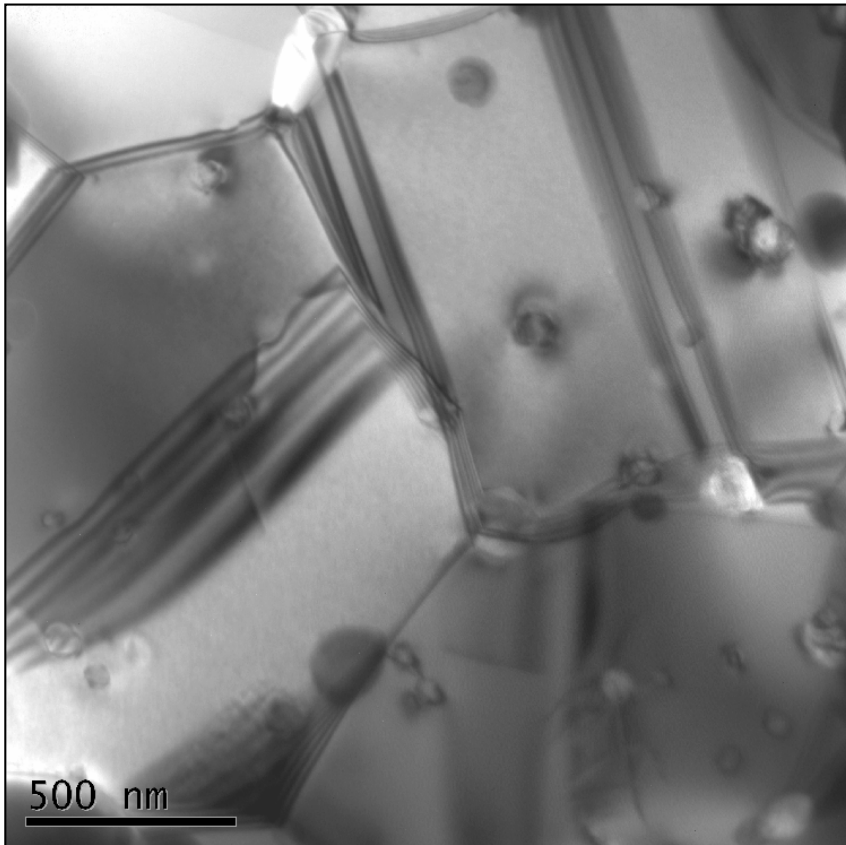


A TEM micrograph of ball milled (2 h) Ni-20Cr-1.2Y₂O₃ powder

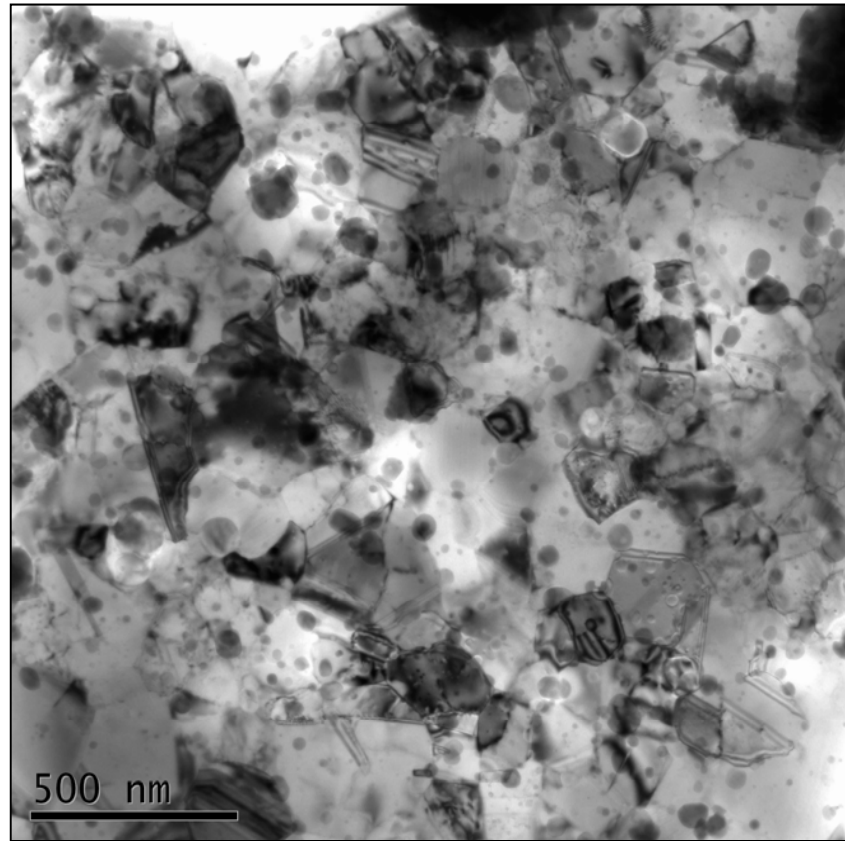
- Ni-20Cr-1.2Y₂O₃ : Avg. powder size - 34 μm; crystallite size - 14 nm
- Ni-20Cr: Avg. powder size - 40 μm ; crystallite size: 92 nm

Microstructure of SPSed Alloys

SPS condition: 1100 °C / 30 min



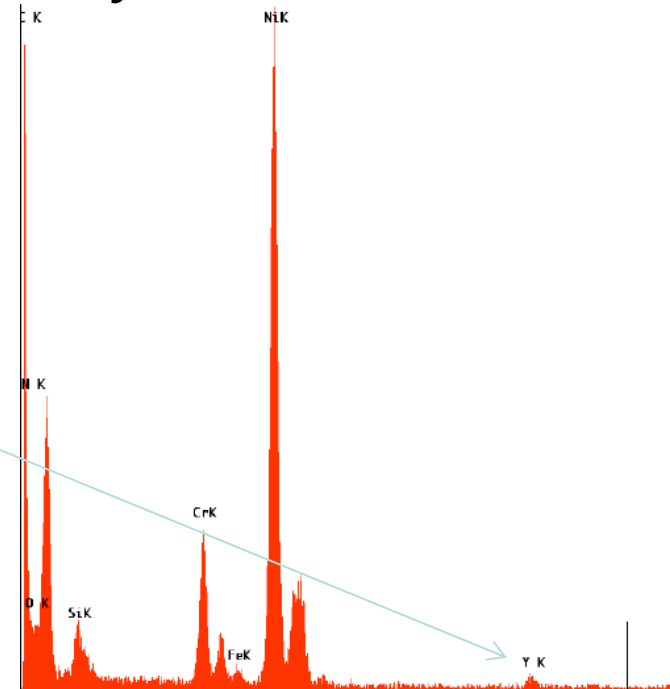
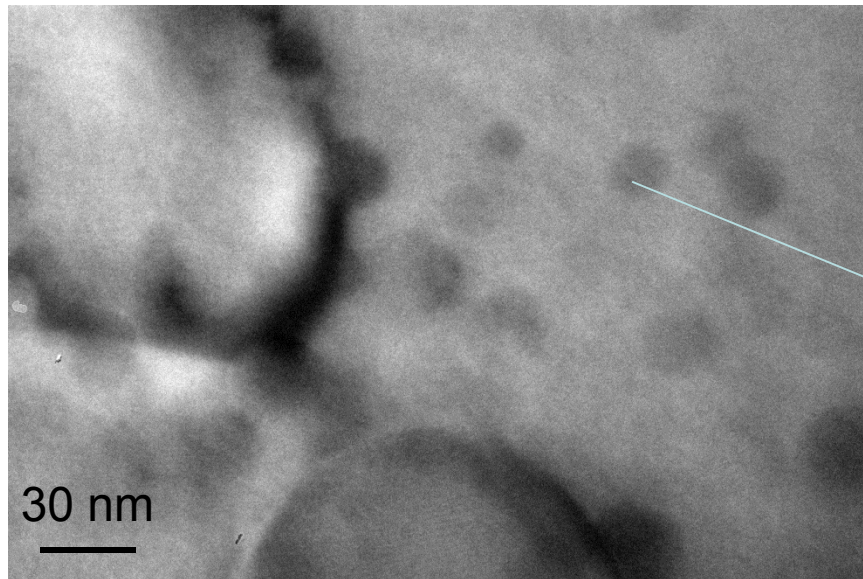
SPSed Ni-20Cr alloy
Avg. Grain size: 630 nm



SPSed Ni-20Cr-1.2Y₂O₃ alloy
Avg. Grain size: 130 nm

SPS condition: 1100 °C / 30 min

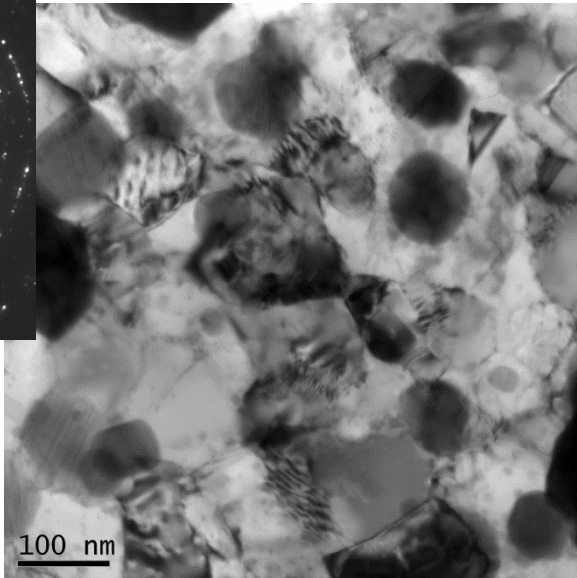
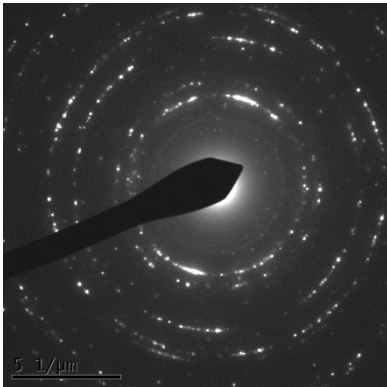
Ni-20Cr-1.2Y₂O₃ alloy



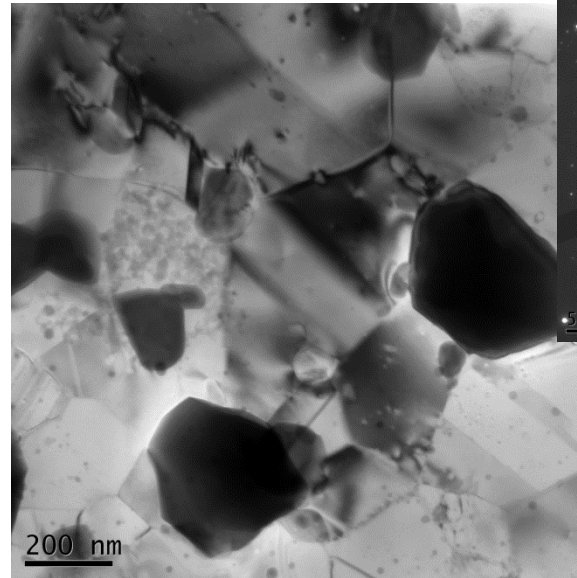
- Three main oxide particle categories in terms of their size:
 - Ni-based oxide in the range of 80-100 nm
 - Cr-based oxide in the range of 20-60 nm
 - Y-based oxide smaller than <15 nm

Microstructure of SPSeD Alloys

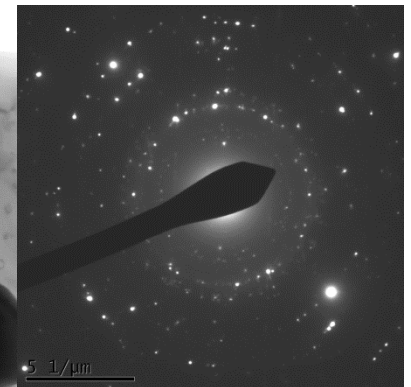
*Milled for 4 hours - SPS condition: 1100 °C / 30 min
Ni-20Cr-1.2Y₂O₃ alloy*



Smaller grain region



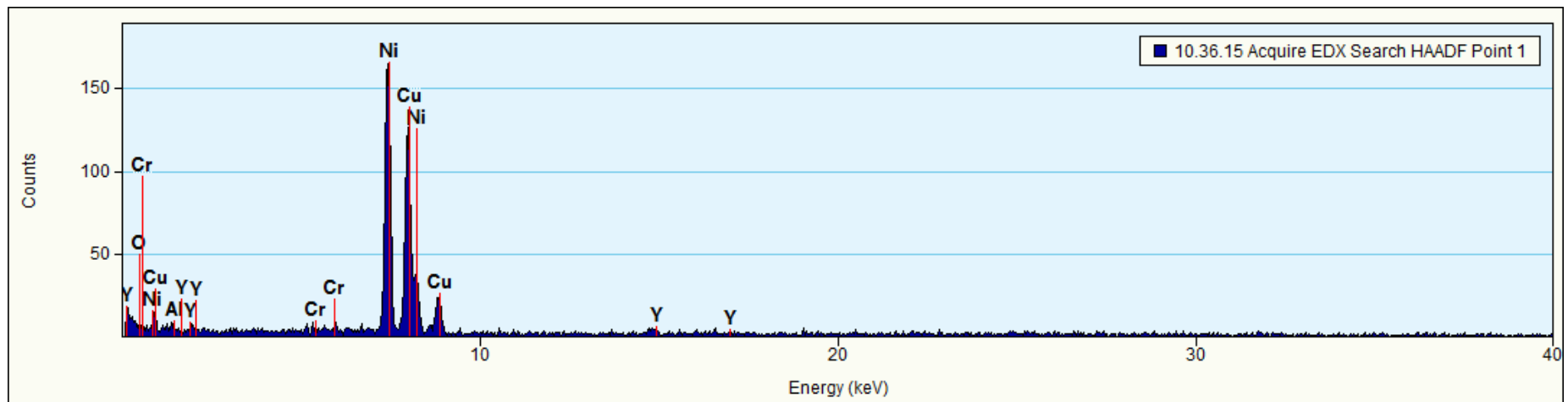
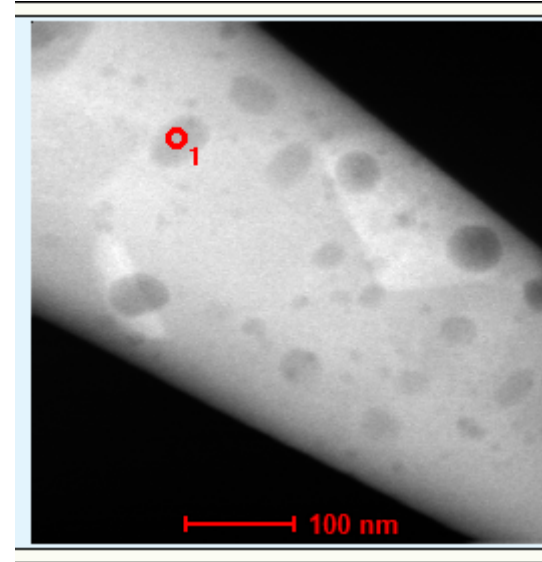
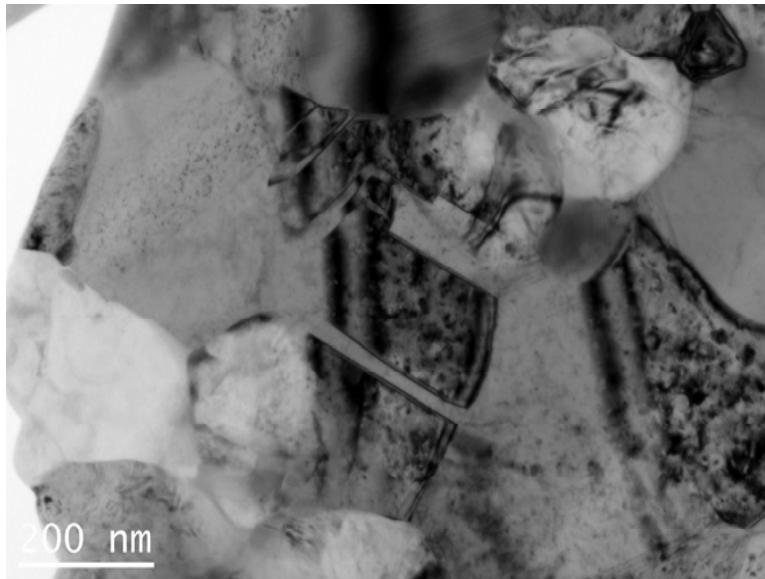
Larger grain region



- With increasing milling time from 2 to 4 hours, there is a tendency to develop bimodal grain size distribution.
- Possibly a higher amount of yttria dissolved in the Ni-Cr matrix.

Microstructure of SPSeD Alloys

*Milled for 2 hours / SPS condition: 1100 °C / 30 min
Ni-20Cr-1.2Y₂O₃-5Al₂O₃ alloy*



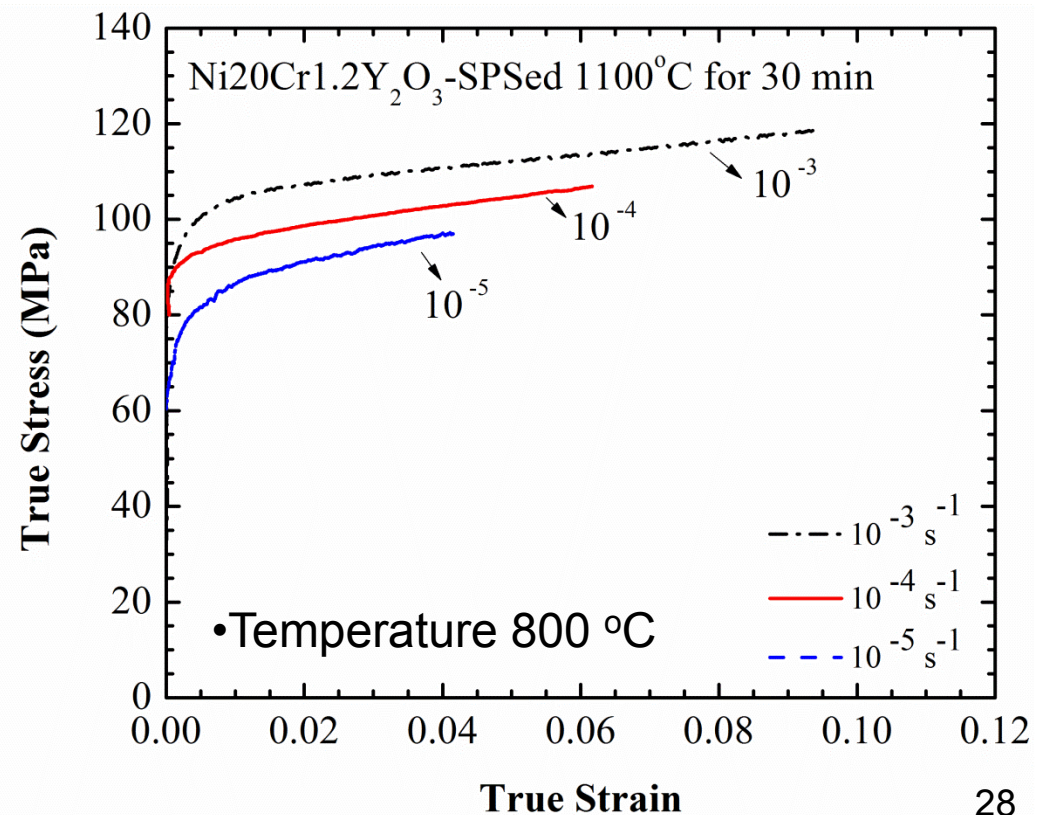
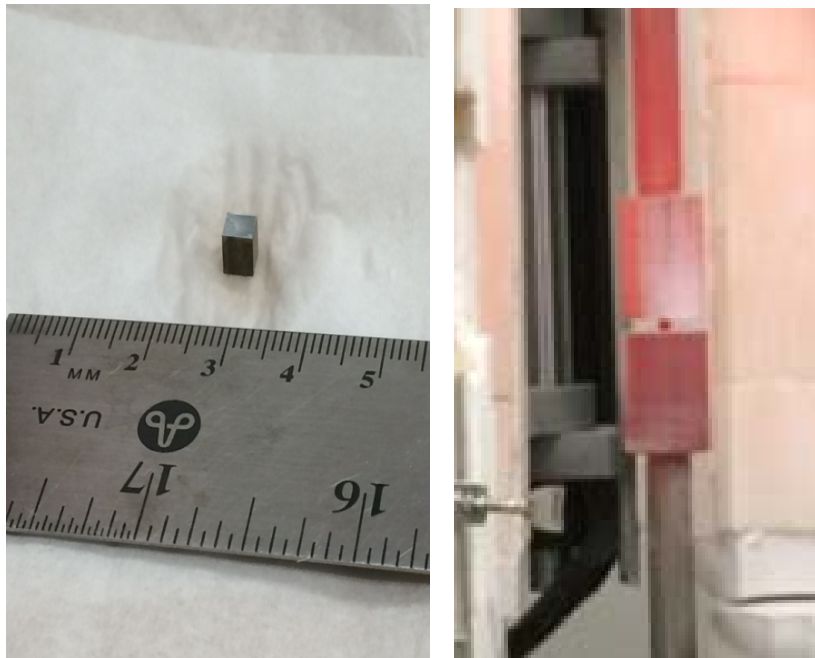
Powder milled for 2 h, BPR of 10 and ball diameter of 5 mm

Alloy Comp. (wt.%)	SPS parameters	Density (g/cm ³)	Relative Density (%)	Hardness (HV)
Ni-20Cr	1100 °C / 30 min	8.19	98.95±0.03	201.6±6.2
Ni-20Cr-1.2Y ₂ O ₃	900 °C / 5 min	7.71	93.93±0.03	395.1±11.4
Ni-20Cr-1.2Y ₂ O ₃	600 °C / 5 min	5.92	72.19±0.24	130.8±31.9
Ni-20Cr-1.2Y ₂ O ₃	1000 °C / 5 min	8.15	99.26±0.30	555.9±4.6
Ni-20Cr-1.2Y ₂ O ₃	1100 °C / 5 min	8.16	99.48±0.05	469.6±7.8
Ni-20Cr-1.2Y ₂ O ₃	1100 °C / 30 min	8.17	99.55±0.04	471.6±7.5
Ni-20Cr- 1.2Y₂O₃-5Al₂O₃	1100 °C / 30 min	7.7	99.18±0.02	505.5±10.3

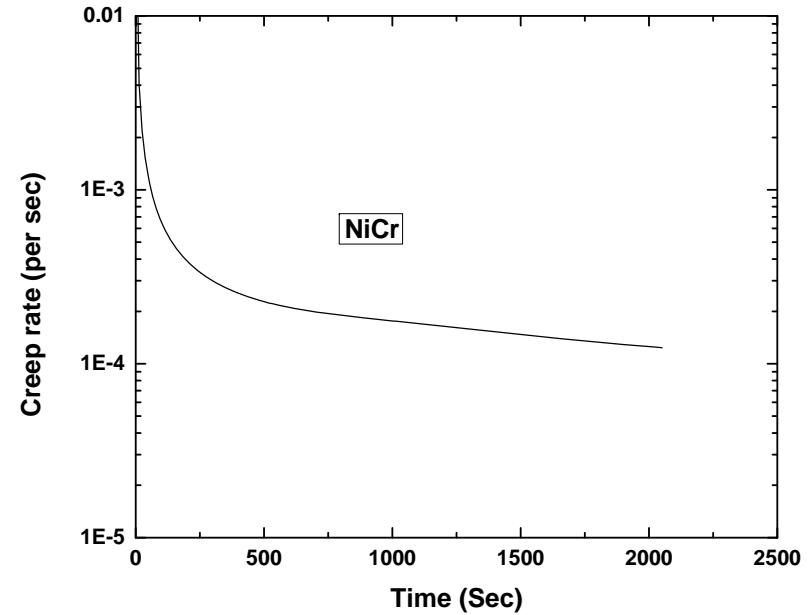
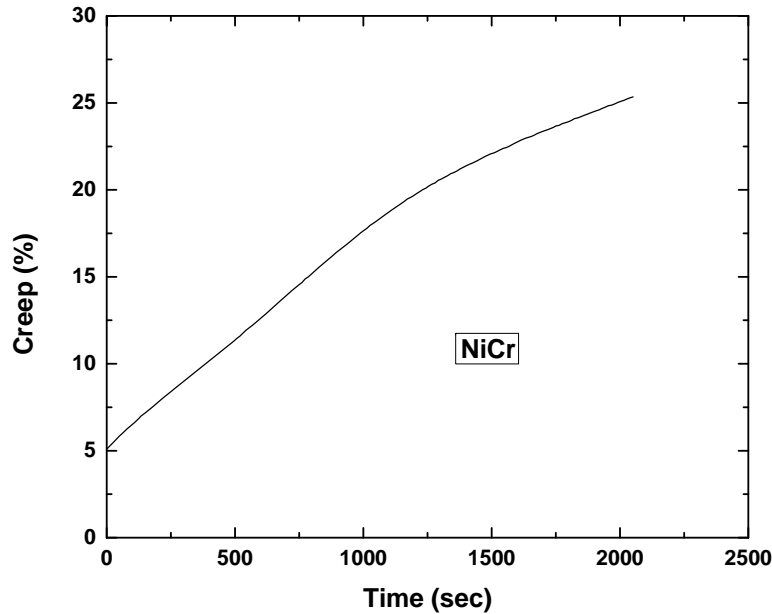
Nano-indentation of Ni-ODS alloys

Sample	NiCr	NiCr-Y ₂ O ₃ -Al ₂ O ₃
Elastic modulus (GPa)	170	249

Compression Test Results



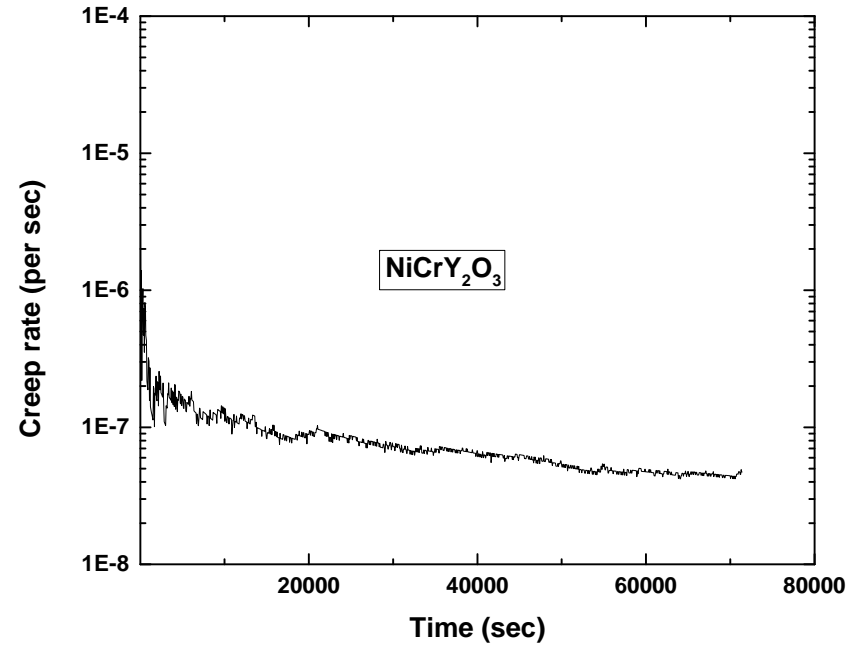
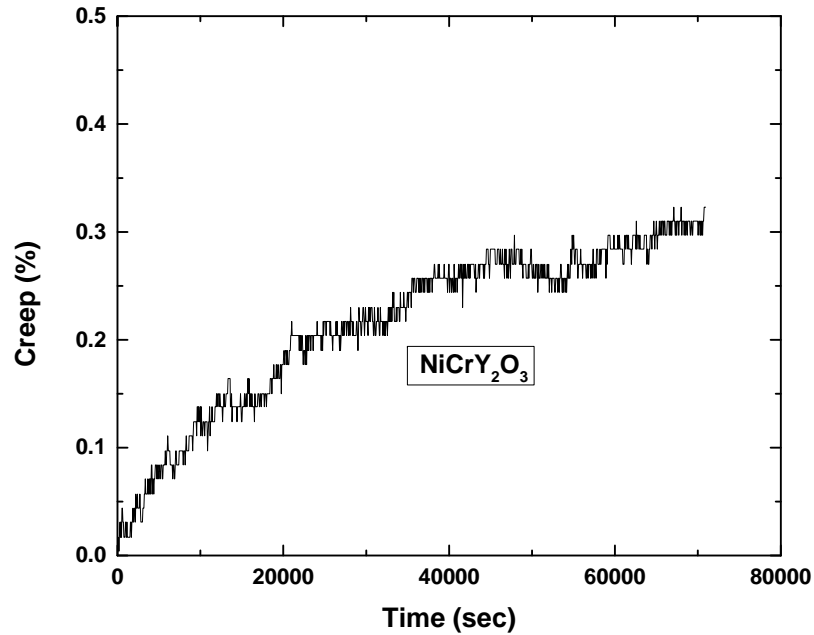
Creep test of NiCr



Specimen dimensions (mm*mm*mm)	Stress applied (MPa)	Testing temperature (°C)
(6.09) * (5.70) * (4.30)	100	800

Minimum creep rate (s^{-1}): 10^{-4}

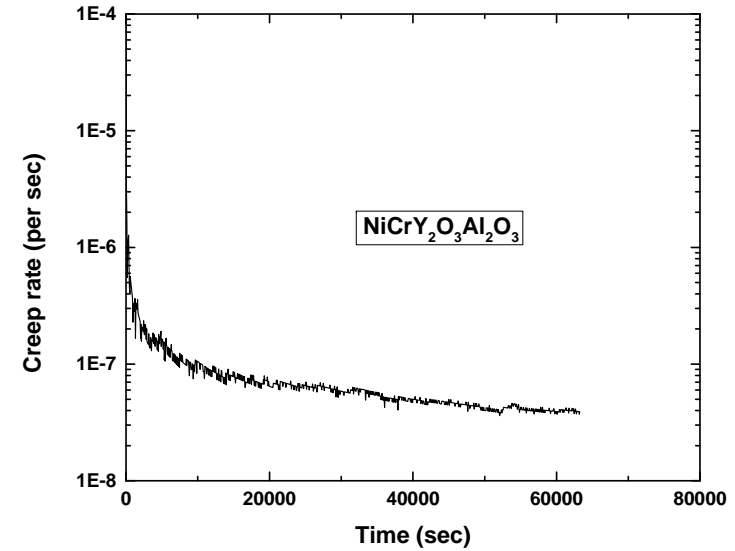
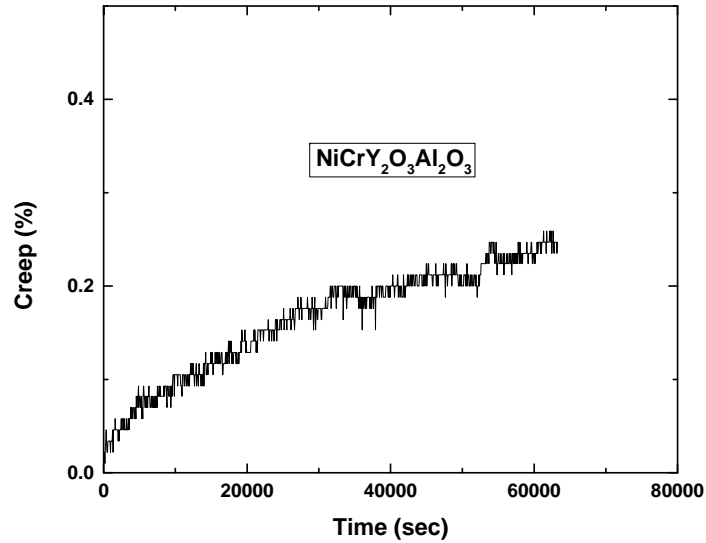
Creep test of NiCr-Y₂O₃



Specimen dimensions (mm*mm*mm)	Stress applied (MPa)	Testing temperature (°C)
(9.97) * (3.76) * (3.77)	100	800

Minimum creep rate (s⁻¹): 4.7*10⁻⁸

Creep test of NiCr-Y₂O₃-Al₂O₃



Specimen dimensions (mm*mm*mm)	Stress applied (MPa)	Testing temperature (°C)
(6.26) * (4.71) * (4.20)	100	800

Minimum creep rate (s⁻¹): 3.7*10⁻⁸

Future Work

- Continue dislocation simulation work
- Complete mechanical property evaluation
- Determine discrepancy between theoretical/computational predictions and experimental results
- Produce guidelines for high temperature microstructural design

Thank You